Lie Group Variational Collision Integrators for a Class of Hybrid Systems*

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4 Abstract. The problem of 3-dimensional, convex rigid-body collision over a plane is fully investigated; this includes bodies with sharp corners that is resolved without the need for nonsmooth convex analysis 56 of tangent and normal cones. In particular, using nonsmooth Lagrangian mechanics, the equations 7 of motion and jump equations are derived, which are largely dependent on the collision detection 8 function. Following the variational approach, a Lie group variational collision integrator (LGVCI) 9 is systematically derived that is symplectic, momentum-preserving, and has excellent long-time, 10 near energy conservation. Furthermore, systems with corner impacts are resolved adeptly using ϵ -rounding on the sign distance function (SDF) of the body. Extensive numerical experiments are 11 12conducted to demonstrate the conservation properties of the LGVCI.

Key words. discrete variational mechanics, variational integrators, Lie group integrators, collisions, nonsmooth
 impacts, hybrid systems

15 **MSC codes.** 37M15, 65P10, 70F35, 70G65, 34A38, 49J52

1. Introduction. Hybrid systems are dynamical systems that exhibits both continuous 16 and discrete dynamics. The state of a hybrid system changes either continuously by the flow 17described by differential equations or discretely following some jump conditions. A canonical 18 example of a hybrid system is the bouncing ball, imagined as a point mass, over a horizontal 19 plane. The extension of this problem to 3-dimensions, wherein the bouncing body is rigid and 20convex, is rather complex, especially in the case of sharp corner impacts; in fact, these systems 21have unilateral constraints that describe the collision surface. We study such problems with 2223perfectly elastic collisions and the Lie group variational collision integrators (LGVCI) are derived following the approaches introduced in [10] and [22]. The advantage of these frameworks 24 is that they yield a global description of the system, in contrast to local representations such 25as Euler angles [30, 45]. Furthermore, in high-precision physics engine and graphics dynamics, 26the integrator becomes a foundation, and its extensions with inelastic collisions and friction 27can be derived to fully actualize the engine. This is also naturally applicable to problems in 28optimal control with similar nonlinear manifold constraints [27, 36, 38, 40, 41]. In particu-29 lar, these constraints and optimal control problems arise in robotics [9, 28, 39, 56, 57] and 30 31 multi-body dynamics [18, 26].

There is an extensive literature on various extensions of the bouncing ball example. In fact, it is a subset of the broader, classical field of rigid-body dynamics, which has a strong emphasis on collisions, contact, and friction. Due to its practical importance and various theoretical challenges, there have been extensive studies which have been summarized in textbooks [3,

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4, 13, 14, 19, 24, 43, 47, 53], some of which are considered classical references in the field. However, there is a notable absence in the literature on global descriptions of rigid-body dynamics with collisions, e.g., configurations of the bodies via Lie groups. This is due to the fact that systems with collisions are not continuous because there is an instantaneous jump in momenta after each collision. Consequently, research interests focus mostly on discrete impact solutions of systems with rigid-bodies, which date back to Brach's "Rigid Body Collisions" in 1989 [1, 2].

There are, of course, more general studies of impacts for rigid-bodies, for example, the case 43of extremely small deformations (also known as "hard" collisions) at the contacting points; the 44 research on this case stem from three main theories regarding the compression and restitution 45 phases brought forth by Newton in 1833 [35], Poisson in 1838 [44], and Stronge as early as 46 1990 [50, 51, 52, 54]. Actually, these "extremely small" deformations are one of the three 47 main categories of collision problems indicated by Chatterjee and Ruina in [5] and Najafabadi 48 et al. in [34]. The first other category is "small" deformation collision which can be resolved 49using compliant contact modeling such as Hertz's model [16] and non-linear damping model 50introduced by Hunt and Crossley [17]; the second is "large" deformation collisions which 51require tools from continuum mechanics, e.g., finite element methods [8, 21, 48, 55]. In short, 52the literature and history of rigid-body and contact dynamics modeling are extensive, so one 53 may refer to the following surveys and reviews in [10, 11, 12, 49] for a more complete picture 54of the field. 55

56 Our approach to the collision problem for a convex rigid-body is based on the variational methodology and integrators developed in Fetecau et al. [10]. They specifically develop 57 the theory for nonsmooth Lagrangian mechanics, which automatically gives a symplectic-58 momentum preserving integrator. Furthermore, near impacts, a collision point and time are determined to solve for the next configurations using the variational method as well. This 60 61 approach was extended to develop collision algorithms for dissipative systems [6, 29, 46] that take advantage of the near-energy preserving properties of the variational integrators in the 62 absence of dissipation in order to more reliably track the energy decay of dissipative systems. 63 64 The case of nonsmooth field theories was considered in [7], which is built on multisymplectic 65 field theories [15] and multisymplectic variational integrators [31].

This paper, however, extends the work of Fetecau et al. to the 3-dimensional case and 66 explicitly uses the special Euclidean group to give a complete description of the system away 67 and during impacts. In addition, it investigates the equations of motion and jump conditions 68 69 at impact for a class of hybrid systems, in which a convex rigid-body is bouncing elastically over a horizontal/tilted plane. The corresponding Lie group variational collision integrators 70(LGVCI) are derived, and extensions of the algorithm are developed for rigid-bodies with sharp 71corners, drawing from the tools of solid geometry. In particular, the signed distance functions 72 [37] will be utilized to cleverly regularize our hybrid systems at corner impacts; consequently, 73 nonsmooth analysis and differential inclusions may be avoided entirely. This work provides the 74 foundation for future directions involving dissipation, multi-body, and articulated rigid-body 75 collisions. 76

1.1. Contributions. We first investigate an ellipsoid bouncing elastically on the horizontal plane. The equations of motion and the jump conditions during impacts are derived,

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and they are expressed in terms of the signed distance function between the ellipsoid and 79 plane. Furthermore, the signed distance function allows us to detect collisions and it has the 80 necessary regularity for us to describe the jump conditions. For the collision response, jump 81 conditions give a unique, instantaneous configuration after each impact time based solely on 82 83 the instantaneous configuration before and the tangent space to the configuration at impact. The paper also develops the LGVCI for our hybrid system. The integrators adopt the usual 84 discrete flow for configurations away from the impact points, and the discrete equations are 85 modified to describe the discrete flow near and at the impacts. Since the integrators are based 86 on variational integrators, they are symplectic, momentum-preserving, and exhibit excellent 87 long-time, near energy conservation. In addition, they respect the Lie group structure of 88 the configuration space. Numerical simulations of the triaxial ellipsoid are presented, and we 89 discuss the Zeno phenomenon, wherein a hybrid system makes an infinite number of jumps – 90 collisions in this case – within a finite time. 91

We demonstrate how to extend the model problem, by considering tilted planes and the 92 unions and/or intersections of convex rigid-bodies. We further develop a sensible and practical 93 regularization for the collision response of convex rigid-bodies with sharp corners that avoids 94 the need for nonsmooth convex analysis, tangent, and normal cones. Since the tangent space of 95the configuration is not well-defined for corner impacts, the method introduces a regularization 96 by smoothing the boundary of the bodies by a small ϵ -parameter to handle such collisions. 97 We provide numerical results for the case of tilted planes, unions of two ellipsoids, and the 98 99 cube.

100 **1.2.** Organization. The paper is organized as follows: In Section 2, background material and a description of the model problem is given. The theory of nonsmooth Lagrangian 101 mechanics and the corresponding collision variational integrator discretization are presented 102 in Section 3. In Section 4, we derive the full equations of motion with jump conditions at 103 104 the point of collision, and then the full variational integrators and algorithms are derived in Section 5. We provide the extension to titled planes and more general rigid-bodies in Section 105 106 6. Finally, numerical experiments for four different hybrid systems are explored, and further discussions of the algorithm are given in Section 7. In Appendix A, we derive the detection 107collision for the ellipsoid; in Appendix B, we characterize the necessary inertia matrices for 108 the equations of motion and variational integrators; in Appendix C, we compute the tangent 109maps used to derive the variational integrators. 110

2. The Problem: Dynamics of a Bouncing Ellipsoid. We want to analyze the dynamics of an ellipsoid bouncing elastically on the plane under the effect of gravity. Hereafter, we will refer to this as the *dynamics of a bouncing ellipsoid*. Some relevant background will be introduced to describe our system, which will provide the necessary foundation to develop our proofs, computations, and generalizations of our theory.

116 **2.1. Notation.** The notation we adopt for linear algebra and the configuration space of 117 Special Euclidean group SE(3) is presented here.

118 **2.1.1. Skew Map, Trace, and Inner products.** Recall that the Lie algebra $\mathfrak{so}(3)$ of the 119 rotation group SO(3) is the set of skew-symmetric matrices. Consider the *skew map* $S : \mathbb{R}^3 \to$

 $\mathfrak{so}(3)$ defined by 120

$$S(\boldsymbol{x}) = \begin{bmatrix} 0 & -x_3 & x_2 \\ x_3 & 0 & -x_1 \\ -x_2 & x_1 & 0 \end{bmatrix}$$

where $S(\mathbf{x})\mathbf{y} = \mathbf{x} \times \mathbf{y}$ for any $\mathbf{x}, \mathbf{y} \in \mathbb{R}^3$. One can show the following properties hold: 122

123 (2.1a)
$$S(x)^T = -S(x),$$

124 (2.1b)
$$S(\boldsymbol{x})^2 = \boldsymbol{x}\boldsymbol{x}^T - \|\boldsymbol{x}\|^2 I_3,$$

124 (2.15)
125 (2.1c)

$$S(\boldsymbol{x}) = \boldsymbol{x}\boldsymbol{x} - \|\boldsymbol{x}\| \ \boldsymbol{1}_{3},$$

$$S(\boldsymbol{x} \times \boldsymbol{y}) = S(\boldsymbol{x})S(\boldsymbol{y}) - S(\boldsymbol{y})S(\boldsymbol{x})$$

$$= \boldsymbol{y}\boldsymbol{x}^{T} - \boldsymbol{x}\boldsymbol{y}^{T}$$

$$=yx^T - xy^T$$

126 (2.1d)
$$S(R\boldsymbol{x}) = RS(\boldsymbol{x})R^T,$$

for all $x, y \in \mathbb{R}^3$ and $R \in SO(3)$, where $I_3 \in \mathbb{R}^{3 \times 3}$ is the identity matrix and $\|\cdot\|$ is the 127Euclidean norm. Furthermore, S is an isomorphism with the inverse defined by 128

129 (2.2)
$$S^{-1}(\Omega)^T = (\Omega_{32}, \Omega_{13}, \Omega_{21}),$$

for any $\Omega \in \mathfrak{so}(3)$. 130

We introduce the following maps: Asym : $\mathbb{R}^{3\times3} \to \mathbb{R}^{3\times3}$ and Sym : $\mathbb{R}^{3\times3} \to \mathbb{R}^{3\times3}$, which 131 are defined respectively by 132

133 (2.3)
$$\operatorname{Asym}(A) = A - A^T,$$

134 (2.4)
$$\operatorname{Sym}(A) = A + A^T$$

for any $A \in \mathbb{R}^{3\times 3}$. The trace of a matrix is denoted by $tr[A] = \sum_{i=1}^{3} A_{ii}$, and satisfies the 135following property: 136

Proposition 2.1. For all skew-symmetric matrices $\Omega \in \mathbb{R}^{3 \times 3}$, 137

138 (2.5)
$$\operatorname{tr}[\Omega^T P] = 0,$$

if and only if $P \in \mathbb{R}^{3 \times 3}$ is symmetric, i.e., $\operatorname{Asym}(P) = 0$. 139

This fact provides an insight into the usual inner product of $\mathbb{R}^{3\times 3}$ where one of the matrices 140

is skew-symmetric. Recall that the inner product $\langle \cdot, \cdot \rangle : \mathbb{R}^{3 \times 3} \times \mathbb{R}^{3 \times 3} \to \mathbb{R}$ is defined by 141

142
$$\langle A, B \rangle = \operatorname{tr}[B^T A]$$

Suppose that Ω is skew-symmetric and $A \in \mathbb{R}^{3 \times 3}$. Since $A = \frac{1}{2} \operatorname{Sym}(A) + \frac{1}{2} \operatorname{Asym}(A)$, we have 143 that by Proposition 2.1, 144

145 (2.6)
$$\langle A, \Omega \rangle = \frac{1}{2} \operatorname{tr}[\Omega^T \operatorname{Asym}(A)].$$

Note that Asym(A) is also skew-symmetric, so this naturally leads us to the inner product on 146 $\mathfrak{so}(3)$. In fact, the Lie algebra is a linear space with the inner product $\langle \cdot, \cdot \rangle_S : \mathfrak{so}(3) \times \mathfrak{so}(3) \to \mathbb{R}$, 147which is induced by the standard inner product on \mathbb{R}^3 via the skew map, 148

149 (2.7)
$$\langle \Omega_1, \Omega_2 \rangle_S = \boldsymbol{\omega}_2^T \boldsymbol{\omega}_1,$$

121

150 where $\Omega_1, \Omega_2 \in \mathfrak{so}(3)$ and $\omega_1, \omega_2 \in \mathbb{R}^3$ such that $S(\omega_i) = \Omega_i$. It can be shown that $\langle \Omega_1, \Omega_2 \rangle_S =$ 151 $\frac{1}{2} \operatorname{tr}[\Omega_2^T \Omega_1]$, and by (2.6), the inner products on $\mathbb{R}^{3\times 3}$ and $\mathfrak{so}(3)$ are related by $\langle A, \Omega \rangle =$ 152 $\langle \operatorname{Asym}(A), \Omega \rangle_S$.

153 **2.1.2. Special Euclidean Group** SE(3). Given our main goal, it is natural to describe 154 the translation and rotation of the ellipsoid using the *Special Euclidean group* SE(3) as our 155 configuration space. Recall that SE(3) is the Lie group defined by

156
$$SE(3) = \{ (\boldsymbol{x}, R) \in \mathbb{R}^3 \times GL(3) \mid R^T R = RR^T = I \text{ and } \det(R) = 1 \}$$

$$= \mathbb{R}^3 \rtimes SO(3),$$

where \rtimes is the semidirect product. The semidirect product structure of SE(3) can be conveniently encoded in terms of homogeneous transformations,

160
$$G = \begin{bmatrix} R & \boldsymbol{x} \\ 0 & 1 \end{bmatrix},$$

where the group operation is defined by the usual matrix multiplication. This allows SE(3)to be viewed as an embedded submanifold in $GL_4(\mathbb{R})$. Furthermore, its action on \mathbb{R}^3 is given by matrix-vector product once we embed $\mathbb{R}^3 \hookrightarrow \mathbb{R}^3 \times \{1\} \subset \mathbb{R}^4$:

164 (2.8a)
$$\begin{bmatrix} R_2 & \boldsymbol{x_2} \\ 0 & 1 \end{bmatrix} \begin{bmatrix} R_1 & \boldsymbol{x_1} \\ 0 & 1 \end{bmatrix} = \begin{bmatrix} R_2 R_1 & R_2 \boldsymbol{x_1} + \boldsymbol{x_2} \\ 0 & 1 \end{bmatrix},$$

165 (2.8b)
$$\begin{bmatrix} R & \boldsymbol{x} \\ 0 & 1 \end{bmatrix} \begin{bmatrix} \boldsymbol{z} \\ 1 \end{bmatrix} = \begin{bmatrix} R\boldsymbol{z} + \boldsymbol{x} \\ 1 \end{bmatrix},$$

166 where
$$R, R_1, R_2 \in SO(3)$$
 and $\boldsymbol{z}, \boldsymbol{x}, \boldsymbol{x}_1, \boldsymbol{x}_2 \in \mathbb{R}^3$. As a result, $(\boldsymbol{x}, R) \in SE(3)$ represents a
167 configuration of the rigid-body where \boldsymbol{x} is the location of the origin of the body-fixed frame
168 relative to the inertial frame and R as the attitude of the body. In particular, if $\boldsymbol{\rho} \in \mathbb{R}^3$ is

169 a vector expressed in the body-fixed frame, then $\boldsymbol{x} + R\boldsymbol{\rho}$ is the same vector expressed in the 170 reference frame (see Figure 1).

171 Furthermore, the Lie algebra $\mathfrak{se}(3)$ is given by

172 (2.9)
$$\mathfrak{se}(3) = \{(\boldsymbol{y}, \Omega) \mid \boldsymbol{y} \in \mathbb{R}^3 \text{ and } \Omega \in \mathfrak{so}(3)\},\$$

173 where the elements of $\mathfrak{se}(3)$ have the following form in the homogeneous representation:

174
$$V = \begin{bmatrix} \Omega & \boldsymbol{y} \\ 0 & 0 \end{bmatrix}.$$

175 It has an induced inner product from the standard inner product of \mathbb{R}^3 , which is given by

176 (2.10)
$$(\boldsymbol{y}_1, \Omega_1) \cdot_S (\boldsymbol{y}_2, \Omega_2) = \boldsymbol{y}_2^T \boldsymbol{y}_1 + \langle \Omega_1, \Omega_2 \rangle_S = \boldsymbol{y}_2^T \boldsymbol{y}_1 + \frac{1}{2} \operatorname{tr}[\Omega_2^T \Omega_1],$$

177 where $(\boldsymbol{y}_1, \Omega_1), (\boldsymbol{y}_2, \Omega_2) \in \mathfrak{se}(3).$

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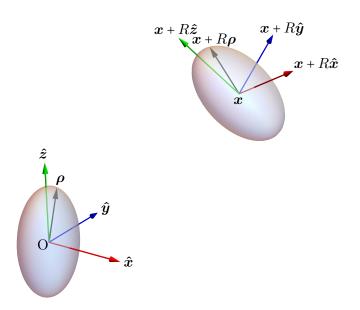


Figure 1: Illustration of the body-fixed frame (left) and inertial frame (right) of an ellipsoid for a given configuration $(\boldsymbol{x}, R) \in SE(3)$

2.2. Distance between an arbitrary ellipsoid and the plane. In order to simulate the dynamics of a bouncing ellipsoid, it is crucial to be able to perform *collision detection* between the ellipsoid and plane for each integration step. If the distance between the two is greater than zero, the next integration step is considered; if it is less than zero, we discard the current step and utilize a rootfinder to find the integration step that advances the solution to the impact point and time. Then, we use the variational collision integrator at the impact point to apply the discrete jump conditions, and then take the remainder of the integration step.

We briefly introduce the notation for the ellipsoid and plane, and then the formula for the collision detection function will be given. A complete description is given in Appendix A. For practicality, the plane is fixed as the horizontal plane defined by

188 (2.11)
$$\mathcal{P} = \{ \boldsymbol{z} \in \mathbb{R}^3 \mid \boldsymbol{n}^T \boldsymbol{z} + 0 = 0 \},\$$

189 where $\boldsymbol{n}^T = (0, 0, 1)$. In general, planes are represented as

190 (2.12)
$$\mathcal{P}(\tilde{\boldsymbol{n}}, D) = \{ \boldsymbol{z} \in \mathbb{R}^3 \mid \tilde{\boldsymbol{n}}^T \boldsymbol{z} + D = 0 \},\$$

where $\tilde{n} \in S^2$ and $D \in \mathbb{R}$. Now, suppose that a, b, c > 0 and consider $f_{\mathcal{E}} : \mathbb{R}^3 \to \mathbb{R}$ defined by 192

193 (2.13)
$$f_{\mathcal{E}}(\boldsymbol{z}) = \frac{z_1^2}{a^2} + \frac{z_2^2}{b^2} + \frac{z_3^2}{c^2} = \boldsymbol{z}^T (I_{\mathcal{E}}^{-1})^2 \boldsymbol{z}, \quad \text{where } I_{\mathcal{E}} = \text{diag}(a, b, c).$$

194 Let us write the standard ellipsoid centered at the origin as

195 (2.14)
$$\mathcal{E}(a,b,c) = \{ \boldsymbol{z} \in \mathbb{R}^3 \mid f_{\mathcal{E}}(\boldsymbol{z}) \le 1 \},\$$

with the shorthand notation \mathcal{E} when the lengths of the semiaxes are understood. Let $(\boldsymbol{x}, R) \in SE(3)$ denote the configuration of the ellipsoid, where \boldsymbol{x} is the center of mass and R is the attitude of the ellipsoid. Consider the map $T_{(\boldsymbol{x},R)} : \mathbb{R}^3 \to \mathbb{R}^3$ defined by

$$T_{(\boldsymbol{x},R)}(\boldsymbol{z}) = R\boldsymbol{z} + \boldsymbol{x},$$

which represents the action of SE(3) on \mathbb{R}^3 . Then, the arbitrary ellipsoid is simply the image of the map, denoted by

$$\mathcal{E}' = T_{(\boldsymbol{x},R)}(\mathcal{E}).$$

202

For a strictly positive distance between the plane and ellipsoid, their intersection is empty. Thus, $\mathcal{E}' \cap \mathcal{P} = \emptyset$ and $x_3 > 0$ since the center of mass of the ellipsoid is always above the horizontal plane for our system. In the case that the distance is zero, their intersection must be a singleton since \mathcal{P} is closed and convex and \mathcal{E}' is compact and strictly convex. Essentially, this is one of the key assumptions of our system to reduce complexity and ensure uniqueness of the ellipsoid's trajectory based on the variational approach. The empty intersection condition gives us the distance formula and a condition:

210 (2.15)
$$d_2(\mathcal{E}', \mathcal{P}) = \min\{|\boldsymbol{n}^T \boldsymbol{x}| \pm \|I_{\mathcal{E}} R^T \boldsymbol{n}\|\} \text{ and } \|I_{\mathcal{E}} R^T \boldsymbol{n}\| < |\boldsymbol{n}^T \boldsymbol{x}|,$$

which follows from Theorem A.11 in Appendix A.2. Note that the inequality $||I_{\mathcal{E}}R^T n|| < |n^T x||$ is equivalent to the condition $\mathcal{E}' \cap \mathcal{P} = \emptyset$. Furthermore, given $n^T x = x_3 > 0$ with the inequality, the minimum value is chosen with the minus sign. As a result, we have the following proposition for our hybrid system:

Proposition 2.2. Let $\Phi : SE(3) \to \mathbb{R}$ be the collision detection function, which is 216 defined by

217 (2.16)
$$\Phi(\boldsymbol{x}, R) = \boldsymbol{n}^T \boldsymbol{x} - \|I_{\mathcal{E}} R^T \boldsymbol{n}\|,$$

218 where n is the normal vector of the plane \mathcal{P} and $I_{\mathcal{E}} = diag(a, b, c)$. Let $(\boldsymbol{x}, R) \in SE(3)$, so 219 that $\mathcal{E}' = T_{(\boldsymbol{x},R)}(\mathcal{E})$ is the arbitrary ellipsoid. Assume that $\mathcal{E}' \cap \mathcal{P} = \emptyset$ and $x_3 > 0$, then

220 (2.17)
$$d_2(\mathcal{E}', \mathcal{P}) = \Phi(x, R).$$

The collision detection function Φ allows us to characterize the admissible set of configurations for the bouncing ellipsoid system. Namely, if the center of mass is below the plane $(x_3 < 0)$, we get $\Phi(\boldsymbol{x}, R) < 0$. In the case that the ellipsoid intersects the plane, the inequality becomes $\|I_{\mathcal{E}}R^T\boldsymbol{n}\| \ge x_3$, and so $\Phi(\boldsymbol{x}, R) \le 0$. Of course, equality here implies that the ellipsoid makes an impact on the plane without *interpenetration* – a description where no body element of the ellipsoid crosses the plane. Therefore, the signed distance function satisfies $\Phi(\boldsymbol{x}, R) \ge 0$ for all the "allowable" configurations $(\boldsymbol{x}, R) \in SE(3)$, which we will define more rigorously next.

3. Background. Before the variational collision integrators are derived for the bouncing ellipsoid, we will first give an overview of the main ideas and techniques in both the continuoustime and discrete-time setting following the approach developed in [10, 42]. Specific results for the ellipsoid dynamics will be stated in this section, and this will provide us with the tools to construct the necessary integrators in the subsequent sections. 8

3.1. Continuous-Time Model.. Let Q be the configuration manifold. Let the submani-233fold $C \subset Q$ be the *admissible set*, which consists of all the possible configurations where no 234contact occurs. The contact set ∂C consists of all the points at which contact occurs without 235any interpenetration. 236

237Consider a Lagrangian $L: TQ \to \mathbb{R}$. In the nonautonomous approach, we introduce a parameterized variable $\tau \in [0,1]$ for the time and trajectories in Q with mappings $c_t(\tau)$ 238and $c_q(\tau)$, respectively. Assume that there is one contact point occurring at $\tau_i \in (0,1)$ for 239simplicity, but the theory can be easily extended for multiple contacts. Now, let us consider 240the smooth manifold *path space* in [10], 241

242 (3.1)
$$\mathcal{M} = \mathcal{T} \times \mathcal{Q}([0,1], \tau_i, \partial C, Q),$$

where 243

 $\mathcal{T} = \{ c_t \in C^{\infty}([0,1], \mathbb{R}) \mid c_t' > 0 \text{ in } [0,1] \},\$ 244

245
$$\mathcal{Q}([0,1],\tau_i,\partial C,Q) = \{c_q: [0,1] \to Q \mid c_q \in C^0, \text{ piecewise } C^2, \}$$

 $[c_q: [0,1] \to Q \mid c_q \in C^{\circ}, \text{ piecewise } C^{\circ}, \text{ has one singularity at } \tau_i, \text{ and } c_q(\tau_i) \in \partial C \}.$ 246

Then $c \equiv (c_t, c_q) \in \mathcal{M}$, and define the associated curve $q : [c_t(0), c_t(1)] \to Q$ by 247

248
$$q(t) = c_q(c_t^{-1}(t))$$

As a result, we are essentially considering paths on an extended configuration manifold $Q_e =$ 249 $\mathbb{R} \times Q$. Now, the extended action map $\mathfrak{S} : \mathcal{M} \to \mathbb{R}$ is given by 250

251 (3.2)
$$\tilde{\mathfrak{S}}(c) = \int_0^1 \tilde{L}(c(\tau), c'(\tau)) d\tau,$$

where $\tilde{L}: TQ_e \to \mathbb{R}$ is defined as 252

253 (3.3)
$$\tilde{L}(c(\tau), c'(\tau)) = L\left(c_q(\tau), \frac{c'_q(\tau)}{c'_t(\tau)}\right) c'_t(\tau).$$

The factor of c'_t appears due to the Jacobian from the change of coordinates $s = c_t(\tau)$ for the 254usual action of the associated curve 255

256 (3.4)
$$\mathfrak{S}(q) = \int_{c_t(0)}^{c_t(1)} L(q(s), \dot{q}(s)) \, ds.$$

Lastly, we introduce the extended configuration manifold of Q_e , 257

258 (3.5)
$$\ddot{Q}_e = \left\{ \frac{d^2c}{d\tau^2}(0) \in T(TQ_e) \mid c \in C^2([0,1],Q_e) \right\},$$

which helps us derive the equations of motion and jump conditions after taking variations of 259260 the action:

Theorem 3.1. Given a C^k Lagrangian $L, k \ge 2$, there exist a unique C^{k-2} Euler-Lagrange derivative $EL : \ddot{Q}_e \to T^*Q_e$ and a unique C^{k-1} Lagrangian one-form Θ_L on TQ_e such that for all variations $\delta c \in T_c \mathcal{M}$, the variation of the extended action is given by

264 (3.6)
$$d\tilde{\mathfrak{S}} \cdot \delta c = \int_0^{\tau_i} EL(c'') \cdot \delta c \, d\tau + \int_{\tau_i}^1 EL(c'') \cdot \delta c \, d\tau + \Theta_L(c') \cdot \hat{\delta} c \Big|_0^{\tau_i^-} + \Theta_L(c') \cdot \hat{\delta} c \Big|_{\tau_i^+}^1,$$

265 where, in coordinates,

266 (3.7a)
$$EL(c'') = \left[\frac{\partial L}{\partial q}c'_t - \frac{d}{d\tau}\left(\frac{\partial L}{\partial \dot{q}}\right)\right]dc_q + \left[\frac{d}{d\tau}\left(\frac{\partial L}{\partial \dot{q}}\frac{c'_q}{c'_t} - L\right)\right]dc_t,$$

267 (3.7b)
$$\Theta_L(c') = \left[\frac{\partial L}{\partial \dot{q}}\right] dc_q - \left[\frac{\partial L}{\partial \dot{q}}\frac{c'_q}{c'_t} - L\right] dc_t$$

268 (3.7c)
$$\hat{\delta}c(\tau) = \left(\left(c(\tau), \frac{\partial c}{\partial \tau}(\tau) \right), \left(\delta c(\tau), \frac{\partial \delta c}{\partial \tau}(\tau) \right) \right).$$

Hamilton's principle states that the possible trajectories of the system are the critical points of the action map. Therefore, all of the solution paths $c \in \mathcal{M}$ must satisfies $d\tilde{\mathfrak{S}} \cdot \delta c = 0$ for all $\delta c \in T_c \mathcal{M}$, which vanish at the boundary points $\tau = 0$ and $\tau = 1$. Then, if c is a solution, it satisfies

273 (3.8)
$$d\tilde{\mathfrak{S}} \cdot \delta c = \int_0^{\tau_i} EL(c'') \cdot \delta c \, d\tau + \int_{\tau_i}^1 EL(c'') \cdot \delta c \, d\tau + \Theta_L(c') \cdot \hat{\delta} c \Big|_{\tau_i^-}^{\tau_i^+} = 0.$$

The above equation holds if and only if the Euler-Lagrange derivative is zero on the smooth portions $[0, \tau_i) \cup (\tau_i, 1]$, and the Lagrangian one-form has a zero jump at τ_i . The first gives us the *extended Euler-Lagrange equations* expressed in the time domain as

277 (3.9a)
278 (3.9b)

$$\begin{pmatrix} \frac{\partial L}{\partial q} - \frac{d}{dt} \left(\frac{\partial L}{\partial \dot{q}} \right) \end{pmatrix} \cdot \delta q = 0,$$

$$\frac{d}{dt} \left(\frac{\partial L}{\partial \dot{q}} \dot{q} - L \right) = 0,$$

for all $\delta q \in T_{q(t)}C$ and $t \in [t_0, t_i) \cup (t_i, t_1]$ where $t_0 = c_t(0), t_1 = c_t(1)$, and $t_i = c_t(\tau_i)$. There is some redundancy in this set of equations because (3.9b) is actually a consequence of (3.9a):

281 Note that $\frac{d}{dt}\left(\frac{\partial L}{\partial \dot{q}}\right) = \frac{\partial L}{\partial q}$ for all $\delta q \in T_{q(t)}C$, so

282
$$\frac{d}{dt}\left(\frac{\partial L}{\partial \dot{q}}\dot{q} - L\right) = \frac{d}{dt}\left(\frac{\partial L}{\partial \dot{q}}\right)\dot{q} + \frac{\partial L}{\partial \dot{q}}\ddot{q} - \frac{dL}{dt}$$

283
$$= \left[\frac{\partial L}{\partial q}\dot{q} + \frac{\partial L}{\partial \dot{q}}\ddot{q}\right] - \frac{dL}{dt}$$

284 = 0,

since the time derivative of the Lagrangian is the expression in the square brackets. In addition, (3.9b) implies energy conservation for the autonomous system where $E = \frac{\partial L}{\partial \dot{a}} \dot{q} - L$, which is unsurprising, since the standard Euler–Lagrange equation already preserves energy. Hence, it
suffices to consider the standard Euler–Lagrange equation for the equations of motion.

For the Lagrangian one-form, one write it compactly as $\Theta_L = \frac{\partial L}{\partial \dot{q}} dq - E dt$ in the time domain. Then, having a zero jump at τ_i implies that

291 (3.10a)
$$\left(\frac{\partial L}{\partial \dot{q}}\Big|_{t=t_i^+} - \frac{\partial L}{\partial \dot{q}}\Big|_{t=t_i^-}\right) \cdot \delta q = 0,$$

292 (3.10b)
$$E(q, \dot{q})\Big|_{t=t_i^+} - E(q, \dot{q})\Big|_{t=t_i^-} = 0.$$

for all $\delta q \in T_{q(t_i)}\partial C$. This set of equations is known as the *Weierstrass-Erdmann* type conditions for impact. While (3.10a) indicates that the momentum is conserved in the tangential direction of ∂C , (3.10b) indicates the conservation of energy during an elastic impact.

Together, (3.10) gives the solution to $q(t_i^+)$, and it is clear that an obvious solution would be $q(t_i^+) = q(t_i^-)$. However, this is not allowed as the trajectory will no longer stay within the admissible set C. For a contact set ∂C that is a codimension-one smooth submanifold, the solution to (3.10) exists and is locally unique [10].

300 **3.1.1. Legendre Transform.** Although the proof to Theorem 3.1 is omitted, its results, 301 namely the Euler-Lagrange derivative, were obtained by taking the variation with respect to 302 the tangent bundle TQ. Hence, it is also natural to derive Hamilton's equations by taking 303 variations in terms of momenta $\mathbb{F}L(q, \dot{q}) \in T^*Q$ where,

304 (3.11)
$$\mathbb{F}L(q,\dot{q})\cdot\delta\dot{q} = \frac{d}{d\epsilon}\bigg|_{\epsilon=0}L(q,q+\epsilon\delta\dot{q}).$$

The map $\mathbb{F}L$ is the Legendre Transform or fibre derivative, and T^*Q denotes the cotangent bundle. From this transformation, it can be shown that Hamilton's equations are equivalent to the Euler-Lagrange equations and they provide an alternative description of our system [32].

309 **3.2.** Discrete-Time Model. Ideally, we would like to introduce the discrete-time model 310 analogously to the continuous case using the nonautonomous approach. However, there is 311 no guarantee that there exists a varying timestep solution to the extended discrete Euler-312 Lagrange equations as discussed in [20] and [25]. Therefore, we shall only introduce the 313 discrete-time equations using the autonomous approach with a fixed timestep $h \in \mathbb{R}$ instead. 314 In the discrete setting, consider a discrete Lagrangian $L_d: Q \times Q \to \mathbb{R}$ that approximates 315 a segment of the autonomous action integral

316
$$L_d(q_k, q_{k+1}, h) \approx \int_{t_k}^{t_{k+1}} L(q(t), \dot{q}(t)) dt,$$

where $q_k = q(t_k)$, $q_{k+1} = q(t_{k+1})$, and $h = t_{k+1} - t_k$. In general, the action integral is approximated using discrete fixed timesteps,

319
$$t_k = kh, \text{ for } k = 0, 1, \dots, N.$$

Let $\tilde{\alpha} \in [0,1]$ such that $\tilde{\tau} = t_{i-1} + \tilde{\alpha}h$ is the fixed impact time corresponding to the param-320 eterized variable. Denote the actual impact time by $t = t_{i-1} + \alpha h$, where $\alpha = t_d(\tilde{\alpha})$ with t_d 321 being a strictly increasing function from the closed unit interval onto the closed unit interval. 322 Now, define the *discrete path space* to be 323

324 (3.12)
$$\mathcal{M}_d = \mathcal{T}_d \times \mathcal{Q}_d(\tilde{\alpha}, \partial C, Q),$$

where 325

326
$$\mathcal{T}_d = \{ t_d(\tilde{\alpha}) = \alpha \mid t_d \in C^{\infty}([0,1], [0,1]), t_d \text{ onto}, t'_d > 0 \},$$

$$\mathcal{Q}_d(\tilde{\alpha}, \partial C, Q) = \{ q_d : \{ t_0, \dots, t_{i-1}, \tilde{\tau}, t_i, \dots, t_N \} \to Q \mid q_d(\tilde{\tau}) \in \partial C \}.$$

Moreover, we remark that \mathcal{T}_d is actually the closed unit interval [0, 1] given all the possible 328 strictly increasing, surjective functions t_d . 329

For a more convenient notation, identify the discrete trajectory with its image 330

$$(\alpha, q_d) \leftrightarrow (\alpha, \{q_0, \dots, q_{i-1}, \tilde{q}, q_i, \dots, q_N\}),$$

where $q_k = q_d(t_k)$, $\tilde{q} = q_d(\tilde{\tau})$, and $\alpha = t_d(\tilde{\alpha})$. In fact, \mathcal{M}_d is isomorphic to $[0,1] \times Q \times \cdots \times$ 332 $\partial C \times \cdots \times Q$, so it can be viewed as a smooth manifold. Then, for $q_d \in \mathcal{Q}_d(\tilde{\alpha}, \partial C, Q)$, we have 333 the tangent space 334

335
$$T_{q_d}\mathcal{Q}_d(\tilde{\alpha},\partial C,Q) = \{v_{q_d}: \{t_0,\ldots,t_{i-1},\tilde{\tau},t_i,\ldots,t_N\} \to Q \mid v_{q_d}(\tilde{\tau}) \in T_{\tilde{q}}\partial C\}.$$

Now, for $(\alpha, q_d) \in \mathcal{M}$, also identify $(\delta \alpha, v_{q_d}) \in T_{(t_d, q_d)} \mathcal{M}_d$ with its image 336

337
$$(\delta\alpha, v_{q_d}) \leftrightarrow (\delta\alpha, \delta q_d) = (\delta\alpha, \{\delta q_0, \dots, \delta q_{i-1}, \delta\tilde{q}, \delta q_i, \dots, \delta q_N\}),$$

where $\delta q_k = v_{q_d}(t_k)$, $\delta \tilde{q} = v_{q_d}(\tilde{\tau})$, and $\delta \alpha = v_{t_d}(\tilde{\alpha})$. 338 The discrete action map $\mathfrak{S}_d : \mathcal{M}_d \to \mathbb{R}$ is defined by 339

340 (3.13)
$$\mathfrak{S}_d(\alpha, q_d) = \sum_{\substack{k=0\\k\neq i-1}}^{N-1} L_d(q_k, q_{k+1}, h) + L_d(q_{i-1}, \tilde{q}, \alpha h) + L_d(\tilde{q}, q_i, (1-\alpha)h),$$

on which we will take the variations with respect to the discrete path and parameter α . 341 342

Lastly, we define the *discrete second-order manifold* to be

$$\ddot{Q}_d = Q \times Q \times Q,$$

which is locally isomorphic to \ddot{Q}_e . 344

Theorem 3.2. Given a C^k discrete Lagrangian $L_d: Q \times Q \times \mathbb{R} \to \mathbb{R}, k \geq 1$, there exist 345a unique C^{k-1} mapping EL_d : $\ddot{Q}_d \to T^*Q$ and one-forms $\Theta_{L_d}^-$ and $\Theta_{L_d}^+$ on the discrete Lagrangian phase space $Q \times Q$ such that for all variation $(\delta \alpha, \delta q_d) \in T_{(t_d, q_d)} \mathcal{M}_d$ of (t_d, q_d) , the 346 347

348 variation of the discrete action is given by

$$d\mathfrak{S}_{d}(\alpha, q_{d}) \cdot (\delta\alpha, \delta q_{d}) = \sum_{k=1}^{i-2} EL_{d}(q_{k-1}, q_{k}, q_{k+1}) \cdot \delta q_{k} + \sum_{k=i+1}^{N-1} EL_{d}(q_{k-1}, q_{k}, q_{k+1}) \cdot \delta q_{k} + \Theta_{L_{d}}^{+}(q_{N-1}, q_{N}) \cdot (\delta q_{N-1}, \delta q_{N}) - \Theta_{L_{d}}^{-}(q_{0}, q_{1}) \cdot (\delta q_{0}, \delta q_{1}) + [D_{2}L_{d}(q_{i-2}, q_{i-1}, h) + D_{1}L_{d}(q_{i-1}, \tilde{q}, \alpha h)] \cdot \delta q_{i-1} + h[D_{3}L_{d}(q_{i-1}, \tilde{q}, \alpha h) - D_{3}L_{d}(\tilde{q}, q_{i}, (1-\alpha)h)] \cdot \delta \alpha + i^{*}(D_{2}L_{d}(q_{i-1}, \tilde{q}, \alpha h) + D_{1}L_{d}(\tilde{q}, q_{i}, (1-\alpha)h)) \cdot \delta \tilde{q} + [D_{2}L_{d}(\tilde{q}, q_{i}, (1-\alpha)h) + D_{1}L_{d}(q_{i}, q_{i+1}, h)] \cdot \delta q_{i}$$

where $i^*: T^*Q \to T^*\partial C$ is the **cotangent lift** of the embedding $i: \partial C \to Q$.

351 The map EL_d is called the **discrete Euler–Lagrange derivative** and the one-forms $\Theta_{L_d}^-$

352 and $\Theta_{L_d}^+$ are the **discrete Lagrangian** one-forms. These are the coordinate expressions,

353 (3.15)
$$EL_d(q_{k-1}, q_k, q_{k+1}) = [D_2L_d(q_{k-1}, q_k, h) + D_1L_d(q_k, q_{k+1}, h)] dq_k,$$

354 for $k \in \{1, \dots, i-2, i, \dots, N-1\}$ and

355 (3.16a)
$$\Theta_{L_d}^+(q_k, q_{k+1}) = D_2 L_d(q_k, q_{k+1}, h) \, dq_{k+1},$$

356 (3.16b)
$$\Theta_{L_d}^-(q_k, q_{k+1}) = -D_1 L_d(q_k, q_{k+1}, h) \, dq_k.$$

Note that D_i denotes the partial derivative with respect to the *i*-th argument of the discrete Lagrangian. Now, by the discrete Hamilton's principle, the possible discrete, stationary paths (t_d, q_d) are critical points of the discrete action map. Therefore, the solution,

$$(\alpha, \{q_0, \ldots, q_{i-1}, \tilde{q}, q_i, \ldots, q_N\}),$$

satisfies $d\mathfrak{S}_d(\alpha, q_d) \cdot (\delta\alpha, \delta q_d) = 0$ for all variations $(\delta\alpha, \delta q_d) \in T_{(\alpha, q_d)}\mathcal{M}_d$ that vanish at index 0 and N.

363 Using equation (3.14), we conclude that the discrete Euler–Lagrange derivative vanishes,

364 (3.17)
$$[D_2L_d(q_{k-1}, q_k, h) + D_1L_d(q_k, q_{k+1}, h)] \cdot \delta q_k = 0,$$

for all $\delta q_k \in T_{q_k}C$ and all $k \in \{1, 2, \dots, i-2, i+1, \dots, N-1\}$. This is know as the *discrete Euler-Lagrange equations*, which describes the system away from the impact point.

367 Then prior to the impact, we have this system of equations,

368 (3.18a)
$$[D_2L_d(q_{i-2}, q_{i-1}, h) + D_1L_d(q_{i-1}, \tilde{q}, \alpha h)] \cdot \delta q_{i-1} = 0,$$

369 (3.18b) $\tilde{q} \in \partial C,$

370 which can be used to solve for α and \tilde{q} , the impact point. Next, we have the discrete impact 371 condition,

372 (3.19a)
$$D_3L_d(q_{i-1}, \tilde{q}, \alpha h) - D_3L_d(\tilde{q}, q_i, (1-\alpha)h) = 0,$$

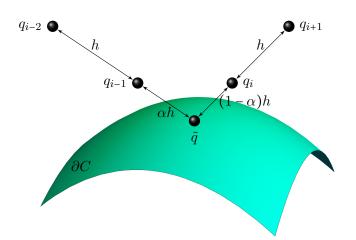


Figure 2: The discrete variational principle for collisions

373 (3.19b)
$$i^*(D_2L_d(q_{i-1}, \tilde{q}, \alpha h) + D_1L_d(\tilde{q}, q_i, (1-\alpha)h)) \cdot \delta \tilde{q} = 0,$$

where we solve for q_i . To provide an interpretation for the discrete equations above, define the *discrete energy* $E_d: Q \times Q \to \mathbb{R}$ by

376 (3.20)
$$E_d(q_k, q_{k+1}, h) = -D_3 L_d(q_k, q_{k+1}, h).$$

This definition is motivated by the fact that it yields the exact Hamiltonian if we apply it to the exact discrete Lagrangian L_d^E . The exact discrete Lagrangian L_d^E is equal to action integral of the exact solution of the Euler-Lagrange equations that satisfy the prescribed boundary conditions, and is related to Jacobi's solution of the Hamilton-Jacobi equation. Hence, equation (3.19a) represents the conservation of discrete energy while equation (3.19b) expresses the conservation of the discrete momentum projected onto $T^*\partial C$. Lastly, we obtain q_{i+1} by solving the last system of equations,

384 (3.21)
$$[D_2L_d(\tilde{q}, q_i, (1-\alpha)h) + D_1L_d(q_i, q_{i+1}, h)] \cdot \delta q_i = 0.$$

Equations (3.17)-(3.19) and (3.21) describe the complete trajectory of the discrete path including the impact point and time. We will rely on these equations to construct our algorithm to simulate the dynamics of the bouncing ellipsoid.

388 **3.2.1. Discrete Legendre Transforms.** In the discrete-time model, we may also write the 389 variational collision integrator in Hamiltonian form via the discrete analogue of the Legendre 390 transform, known as the *discrete Legendre transforms* or *discrete fiber derivatives* $\mathbb{F}^{\pm}L_d$: 391 $Q \times Q \to T^*Q$. They are defined by

392 (3.22a)
$$\mathbb{F}^{-}L_d(q_k, q_{k+1}) \cdot \delta q_k = -D_1 L_d(q_k, q_{k+1}) \cdot \delta q_k,$$

393 (3.22b)
$$\mathbb{F}^+ L_d(q_k, q_{k+1}) \cdot \delta q_{k+1} = D_2 L_d(q_k, q_{k+1}) \cdot \delta q_{k+1}.$$

394 This allows us to introduce momenta as images of the discrete Legendre transform,

395 (3.23a)
$$p_{k,k+1}^+ = p^+(q_k, q_{k+1}, h) = \mathbb{F}^+ L_d(q_k, q_{k+1}),$$

396 (3.23b)
$$p_{k,k+1}^- = p^-(q_k, q_{k+1}, h) = \mathbb{F}^- L_d(q_k, q_{k+1}).$$

³⁹⁷ The discrete Euler–Lagrange equations (3.17) may be rewritten as

398
$$\mathbb{F}^+L_d(q_{k-1}, q_k) \cdot \delta q_k = \mathbb{F}^-L_d(q_k, q_{k+1}) \cdot \delta q_k.$$

This translates to $p_{k-1,k}^+ = p_{k,k+1}^-$, which shows that the discrete momentum expressed on the time interval $[t_{k-1}, t_k]$ is the same as on the time interval $[t_k, t_{k+1}]$. This allows us to interpret (3.19b) as the conservation of discrete momentum projected onto $T^*\partial C$ during the impact.

402 **4. Dynamics of the Bouncing Ellipsoid.** In this section, we derive the continuous equa-403 tions of motion and the jump conditions for the bouncing ellipsoid. However, we first construct 404 the Lagrangian based on the approach described in [22].

Consider the configuration space Q = SE(3). Denote the set of body elements of a rigidbody by \mathcal{B} , namely the ellipsoid, and let $(\boldsymbol{x}, R) \in SE(3)$ describe its configuration. Then, the inertial position of a body element of \mathcal{B} is $\boldsymbol{x} + R\boldsymbol{\rho}$, where $\boldsymbol{\rho} \in \mathbb{R}^3$ is the position of the body element relative to the origin of the body-fixed frame. Thus, the kinetic energy is written as

409
$$T = \frac{1}{2} \int_{\mathcal{B}} \|\dot{\boldsymbol{x}} + \dot{R}\boldsymbol{\rho}\|^2 \, dm = \frac{1}{2} \int_{\mathcal{B}} \|\dot{\boldsymbol{x}}\|^2 \, dm + \int_{\mathcal{B}} \dot{\boldsymbol{x}}^T \dot{R}\boldsymbol{\rho} \, dm + \frac{1}{2} \int_{\mathcal{B}} \|\dot{R}\boldsymbol{\rho}\|^2 \, dm$$

410 Recall that $\int_{\mathcal{B}} \rho \, dm = 0$ since the origin of the body-fixed frame is the center of mass of the

411 rigid-body. Expanding $\|\dot{R}\boldsymbol{\rho}\|^2 = \operatorname{tr}[\dot{R}\boldsymbol{\rho}\boldsymbol{\rho}^T\dot{R}^T]$, we obtain,

412
$$T = \frac{1}{2}m\|\dot{\boldsymbol{x}}\|^2 + \frac{1}{2}\operatorname{tr}[\dot{R}J_d\dot{R}^T],$$

413 where *m* is the total mass and $J_d = \int_{\mathcal{B}} \boldsymbol{\rho} \boldsymbol{\rho}^T dm$ is a nonstandard moment of inertia matrix. 414 By property (2.1b) of the skew map, J_d is related to the standard moment of inertia matrix 415 $J = \int_{\mathcal{B}} S(\boldsymbol{\rho})^T S(\boldsymbol{\rho}) dm$ by

416 (4.1)
$$J = \operatorname{tr}[J_d]I_3 - J_d,$$
$$J_d = \frac{1}{2}\operatorname{tr}[J]I_3 - J,$$

417 and

418 (4.2)
$$S(J\mathbf{\Omega}) = S(\mathbf{\Omega})J_d + J_d S(\mathbf{\Omega}),$$

for any $\Omega \in \mathbb{R}^3$ (see Proposition B.1 in Appendix B). Since the ellipsoid is under the influence of a uniform, constant gravitational field in the z-component, the potential energy is written as

422
$$U = mge_3^T x$$

423 where g is the gravitational acceleration. Hence the Lagrangian $L: TSE(3) \to \mathbb{R}$ is defined 424 by

425 (4.3)
$$L(\boldsymbol{x}, R, \dot{\boldsymbol{x}}, \dot{R}) = \frac{1}{2}m\|\dot{\boldsymbol{x}}\|^2 + \frac{1}{2}\operatorname{tr}[\dot{R}J_d\dot{R}^T] - mg\boldsymbol{e}_{\boldsymbol{3}}^T\boldsymbol{x}.$$

This form of the Lagrangian is useful for computing the equations of motion and the jump 426 conditions directly from Theorem 3.1. However, a modified Lagrangian expressed in terms of 427 identifications with the Lie algebra, which is a linear space, is useful for computing variations. 428 We recall that when $R \in SO(3)$, $\dot{R} = RS(\Omega)$ for some $\Omega \in \mathbb{R}^3$ by left-trivialization. Since 429 $SE(3) = \mathbb{R}^3 \rtimes SO(3)$ is defined by a semidirect product, we are careful with the identification 430 on the tangent bundle $TSE(3) = SE(3) \times \mathfrak{se}(3)$ where SE(3) is associated with $\mathbb{R}^3 \times SO(3)$ and 431 $\mathfrak{se}(3)$ with $\mathbb{R}^3 \times \mathfrak{so}(3)$. Furthermore, $\mathfrak{so}(3) \simeq \mathbb{R}^3$ by the skew map. With these identifications, 432 we can introduce a modified Lagrangian $\tilde{L}: \mathbb{R}^3 \times SO(3) \times \mathbb{R}^3 \times \mathbb{R}^3 \to \mathbb{R}$. Denote the position, 433 attitude, linear velocity, and angular velocity of the ellipsoid by $(\boldsymbol{x}, \boldsymbol{R}, \dot{\boldsymbol{x}}, \boldsymbol{\Omega})$, respectively. 434 Then, the modified Lagrangian is defined by 435

436 (4.4a)
$$\tilde{L}(\boldsymbol{x}, R, \dot{\boldsymbol{x}}, \boldsymbol{\Omega}) = \frac{1}{2}m\|\dot{\boldsymbol{x}}\|^2 + \frac{1}{2}\operatorname{tr}[S(\boldsymbol{\Omega})J_dS(\boldsymbol{\Omega})^T] - mg\boldsymbol{e}_{\boldsymbol{3}}^T\boldsymbol{x}$$

437 (4.4b)
$$= \frac{1}{2}m\|\dot{\boldsymbol{x}}\|^2 + \frac{1}{2}\boldsymbol{\Omega}^T J\boldsymbol{\Omega} - mg\boldsymbol{e}_{\boldsymbol{3}}^T \boldsymbol{x}$$

438 which follows from Proposition B.2 in Appendix B.

439 Here, the admissible set is

440 (4.5)
$$C = \{ (\boldsymbol{x}, R) \in SE(3) \mid \Phi(\boldsymbol{x}, R) > 0 \} \subset Q,$$

441 where Φ is the collision detection function discussed in Section 2.2. Furthermore, this means 442 that the boundary is the zero level set of the function,

443 (4.6)
$$\partial C = \{(\boldsymbol{x}, R) \in SE(3) \mid \Phi(\boldsymbol{x}, R) = 0\} = \Phi^{-1}(0),$$

444 where 0 is a regular value.

445 4.1. Equations of Motion. We shall compute the equations of motion in two different 446 ways. The first involves careful choices of variations of the modified Lagrangian that respect 447 the geometry of the configuration space, and the second directly computes the Euler–Lagrange 448 equations.

449 **4.1.1. Equations of Motion: First Variation.** Consider the action integral in terms of 450 the modified Lagrangian,

451
$$\mathfrak{S}(\boldsymbol{x},R) = \int_{t_0}^{t_f} \tilde{L}(\boldsymbol{x},R,\dot{\boldsymbol{x}},\boldsymbol{\Omega}) dt,$$

where $(\boldsymbol{x}(t), R(t)) \in C$ for $t \in [t_0, t_f]$. In order to compute the first variation, we introduce the variations for each variable. First, the variation $\delta \boldsymbol{x} : [t_0, t_f] \to \mathbb{R}^3$ vanishes at the initial time t_0 and final time t_f , and this induces the variation $\delta \dot{\boldsymbol{x}} : [t_0, t_f] \to \mathbb{R}^3$. Now, vary the ⁴⁵⁵ rotation matrix in SO(3) by expressing it as $R^{\epsilon} = Re^{\epsilon \eta}$ where $\epsilon \in \mathbb{R}$ and $\eta \in \mathfrak{so}(3)$. Then, the ⁴⁵⁶ variation is given by

457 (4.7)
$$\delta R = \frac{d}{d\epsilon} \bigg|_{\epsilon=0} R^{\epsilon} = R\eta.$$

Lastly, the variation of Ω can be derived from the relationship $\dot{R} = RS(\Omega)$ and (4.7), giving us

460 (4.8)
$$S(\delta \mathbf{\Omega}) = \delta (R^T \dot{R})$$
$$= \delta R^T \dot{R} + R^T \delta \dot{R}$$
$$= -\eta S(\Omega) + S(\Omega)\eta + \dot{\eta},$$

461 where $\delta \dot{R} = \dot{R}\eta + R\dot{\eta}$.

Taking the variation of the action integral will require the variation of the Lagrangian. We compute the variation of the kinetic energy first:

464
$$\delta \tilde{T} = \frac{d}{d\epsilon} \bigg|_{\epsilon=0} \tilde{T}(\dot{\boldsymbol{x}} + \epsilon \delta \dot{\boldsymbol{x}}, \boldsymbol{\Omega} + \epsilon \delta \boldsymbol{\Omega}) = m \dot{\boldsymbol{x}}^T \delta \dot{\boldsymbol{x}} + \frac{1}{2} \operatorname{tr}[S(\delta \boldsymbol{\Omega}) J_d S(\boldsymbol{\Omega})^T + S(\boldsymbol{\Omega}) J_d S(\delta \boldsymbol{\Omega})^T].$$

Substitute (4.8) into the equation above and utilize properties of the trace and (4.2) to arrive at

467
$$\delta \tilde{T} = m \dot{\boldsymbol{x}}^T \delta \dot{\boldsymbol{x}} + \frac{1}{2} \operatorname{tr}[-\dot{\eta} S(J\Omega)] + \frac{1}{2} \operatorname{tr}[\eta S(\boldsymbol{\Omega} \times J\boldsymbol{\Omega})].$$

468 For the potential term, we get

469
$$\delta \tilde{U} = \frac{d}{d\epsilon} \bigg|_{\epsilon=0} \tilde{U}(\boldsymbol{x} + \epsilon \delta \boldsymbol{x}) = mg\boldsymbol{e}_{\boldsymbol{3}}^{T} \delta \boldsymbol{x}.$$

470 Then, the variation of the action integral is

471
$$\delta\mathfrak{S}(\boldsymbol{x},R) = \int_{t_0}^{t_f} \left(m \dot{\boldsymbol{x}}^T \delta \dot{\boldsymbol{x}} + \frac{1}{2} \operatorname{tr} \left[-\dot{\eta} S(J\Omega) \right] + \frac{1}{2} \operatorname{tr} \left[\eta S(\boldsymbol{\Omega} \times J\Omega) \right] \right) - \left(m g \boldsymbol{e}_{\boldsymbol{3}}^T \delta \boldsymbol{x} \right) dt.$$

472 Integrating by parts gives

473
$$\delta\mathfrak{S}(\boldsymbol{x},R) = m\dot{\boldsymbol{x}}^T \delta \boldsymbol{x} \Big|_{t_0}^{t_f} - \frac{1}{2} \operatorname{tr} \left[\eta S(J\boldsymbol{\Omega}) \right] \Big|_{t_0}^{t_f} + \int_{t_0}^{t_f} -m\ddot{\boldsymbol{x}}^T \delta \boldsymbol{x} + \frac{1}{2} \operatorname{tr} \left[\eta \left\{ S(J\dot{\boldsymbol{\Omega}}) + S(\boldsymbol{\Omega} \times J\boldsymbol{\Omega}) \right\} \right] dt + \int_{t_0}^{t_f} (-mg\boldsymbol{e}_{\boldsymbol{3}}^T \delta \boldsymbol{x}) dt.$$

475 Since δx and η vanish at t_0 and t_f , the boundary terms also vanish. This gives

476
$$\delta\mathfrak{S}(\boldsymbol{x},R) = \int_{t_0}^{t_f} -\delta\boldsymbol{x}^T \left\{ m\ddot{\boldsymbol{x}} + mg\boldsymbol{e_3} \right\} + \frac{1}{2} \operatorname{tr} \left[\eta \left\{ S(J\dot{\boldsymbol{\Omega}} + \boldsymbol{\Omega} \times J\boldsymbol{\Omega}) \right\} \right] dt.$$

477 By Hamilton's principle, $\delta \mathfrak{S}(\boldsymbol{x}, R) = 0$ for all variations $\delta \boldsymbol{x}$ and η , and by the fundamental 478 theorem of the calculus of variations, this is true if and only if the expressions in each curly 479 bracket are zero. Since η is an arbitrary skew-symmetric matrix, by Proposition 2.1, the 480 second expression must be a symmetric matrix. As a result,

481
$$0 = \operatorname{Asym}(S(J\dot{\Omega} + \Omega \times J\Omega)) = 2S(J\dot{\Omega} + \Omega \times J\Omega),$$

but $S(J\dot{\Omega} + \Omega \times J\Omega)$ vanishes if and only if $J\dot{\Omega} + \Omega \times J\Omega = 0$. Thus, the continuous equations of motion in Lagrangian form for the bouncing ellipsoid are given by

484 (4.9)
$$\begin{cases} \dot{\boldsymbol{v}} = -g\boldsymbol{e_3}, \\ J\dot{\boldsymbol{\Omega}} = J\boldsymbol{\Omega} \times \boldsymbol{\Omega}, \\ \dot{\boldsymbol{x}} = \boldsymbol{v}, \\ \dot{\boldsymbol{k}} = RS(\boldsymbol{\Omega}), \end{cases}$$

where $\boldsymbol{v} \in \mathbb{R}^3$ is the translational velocity defined as $\boldsymbol{v} = \dot{\boldsymbol{x}}$. In particular, this describes the motion of the ellipsoid in the admissible set C.

487 **4.1.2. Equations of Motion: Euler–Lagrange Equations.** Here, we consider the Euler– 488 Lagrange equations, which are a consequence of Theorem 3.1,

489
$$\left[\frac{\partial L}{\partial q} - \frac{d}{dt}\left(\frac{\partial L}{\partial \dot{q}}\right)\right] \cdot \delta q = 0,$$

490 where $q = (\mathbf{x}, R) \in C$. The Lagrangian in (4.3) is used to compute the partial derivatives,

491 (4.10a)
$$\frac{\partial L}{\partial q} = \left(\frac{\partial L}{\partial \boldsymbol{x}}, \frac{\partial L}{\partial R}\right) = (-mg\boldsymbol{e_3}, 0),$$

492 (4.10b)
$$\frac{\partial L}{\partial \dot{q}} = \left(\frac{\partial L}{\partial \dot{x}}, \frac{\partial L}{\partial \dot{R}}\right) = (m\dot{x}, \dot{R}J_d),$$

493 where matrix derivatives are involved (see Appendix C). Since $\delta q = (\delta \boldsymbol{x}, \delta R) = (\delta \boldsymbol{x}, R\eta)$, we 494 have $\left[(-mg\boldsymbol{e_3}, 0) - \frac{d}{dt}(m\dot{\boldsymbol{x}}, \dot{R}J_d) \right] \cdot (\delta \boldsymbol{x}, R\eta) = 0$, which gives us

495
$$-\delta \boldsymbol{x}^{T} \left\{ m \ddot{\boldsymbol{x}} + m g \boldsymbol{e_3} \right\} - \operatorname{tr}[\eta^{T} \left\{ R^{T} \ddot{R} J_{d} \right\}] = 0,$$

496 for all variations δx and η . Therefore, the expression in the first curly bracket vanishes 497 which is one of the equations of motion derived previously. For the second term to vanish, 498 by Proposition 2.1, the expression in the second curly bracket must be a symmetric matrix. 499 Hence, $\operatorname{Asym}(R^T \ddot{R} J_d) = 0$. Note that

500
$$R^T \ddot{R} J_d = R^T (\dot{R} S(\mathbf{\Omega}) + RS(\dot{\mathbf{\Omega}})) J_d = S(\mathbf{\Omega})^2 J_d + S(\dot{\mathbf{\Omega}}) J_d$$

501 where $\dot{R} = RS(\Omega)$, which gives

502
$$0 = \operatorname{Asym}(S(\mathbf{\Omega})^2 J_d + S(\dot{\mathbf{\Omega}}) J_d)$$

503
$$= \left[S(\mathbf{\Omega})^2 J_d - J_d S(\mathbf{\Omega})^2 \right] + \left[S(\dot{\mathbf{\Omega}}) J_d + J_d S(\dot{\mathbf{\Omega}}) \right]$$

504
$$= S(\mathbf{\Omega} \times J\mathbf{\Omega}) + S(J\dot{\mathbf{\Omega}}) = S(\mathbf{\Omega} \times J\mathbf{\Omega} + J\dot{\mathbf{\Omega}}),$$

where (2.1c) and (4.2) are used. Since the skew map is a Lie algebra isomorphism, $0 = J\dot{\Omega} + \Omega \times J\Omega$, so we obtain the full set of continuous equations of motion as in (4.9).

4.1.3. Hamilton's Equations. We consider the Legendre transformation for our modified 507 Lagrangian $\mathbb{F}L: SE(3) \times \mathfrak{se}(3) \to SE(3) \times \mathfrak{se}^*(3)$, where $\mathfrak{se}^*(3)$ is identified with $\mathfrak{se}(3)$ by the 508 Riesz representation. Using (3.11) and (4.4a), we compute 509

510

$$\begin{aligned} & \mathbb{F}\tilde{L}(\boldsymbol{x}, R, \dot{\boldsymbol{x}}, S(\boldsymbol{\Omega})) \cdot_{S} (\delta \dot{\boldsymbol{x}}, \eta) = \frac{d}{d\epsilon} \bigg|_{\epsilon=0} \tilde{L}(\boldsymbol{x}, R, \dot{\boldsymbol{x}} + \epsilon \delta \dot{\boldsymbol{x}}, S(\boldsymbol{\Omega}) + \epsilon \eta) \\ & = \delta \dot{\boldsymbol{x}}^{T}(m \dot{\boldsymbol{x}}) + \frac{1}{2} \operatorname{tr}[\eta^{T}(S(\boldsymbol{\Omega})J_{d} + J_{d}S(\boldsymbol{\Omega}))] \end{aligned}$$

512
$$= (m\dot{\boldsymbol{x}}, S(J\boldsymbol{\Omega})) \cdot_S (\delta \dot{\boldsymbol{x}}, \eta).$$

The last line is obtained by using the identity (4.2). From the Legendre transform $\mathbb{F}\tilde{L}$: 513 $(\boldsymbol{x}, R, \dot{\boldsymbol{x}}, S(\boldsymbol{\Omega})) \mapsto (\boldsymbol{x}, R, m\dot{\boldsymbol{x}}, S(J\boldsymbol{\Omega})),$ the linear and angular momentum are written as $\boldsymbol{\gamma} = \boldsymbol{\gamma}$ 514 $m\dot{x}$ and $\Pi = J\Omega$, respectively. Hence, we arrive at the continuous equations of motion in 515516Hamiltonian form,

517 (4.11)
$$\begin{cases} \dot{\boldsymbol{\gamma}} = -mg\boldsymbol{e_3} \\ \dot{\boldsymbol{\Pi}} = \boldsymbol{\Pi} \times \boldsymbol{\Omega}, \\ \dot{\boldsymbol{x}} = \frac{1}{m}\boldsymbol{\gamma}, \\ \dot{\boldsymbol{R}} = RS(\boldsymbol{\Omega}). \end{cases}$$

4.2. Jump Conditions. We derive the jump conditions for our system using (3.10). For 518519convenience, let $q(t_i) = (\mathbf{x}_i, R_i) \in \partial C$. The first condition (3.10a) gives

520
$$\left(m(\dot{\boldsymbol{x}}_{i}^{+}-\dot{\boldsymbol{x}}_{i}^{-}),(\dot{R}_{i}^{+}-\dot{R}_{i}^{-})J_{d}\right)\cdot\delta q=0,$$

for all $\delta q \in T_{q(t_i)} \partial C$, where $\lim_{t \to t_i^{\pm}} (\dot{\boldsymbol{x}}, \dot{R}) = (\dot{\boldsymbol{x}}_i^{\pm}, \dot{R}_i^{\pm})$. We also have $\dot{R}_i^{\pm} = R_i S(\boldsymbol{\Omega}_i^{\pm})$ since 521 $\lim_{t\to t_i^{\pm}}(\boldsymbol{x},R) = (\boldsymbol{x}_i,R_i)$, and so 522

523 (4.12)
$$\left(m(\dot{\boldsymbol{x}}_i^+ - \dot{\boldsymbol{x}}_i^-), R_i S(\boldsymbol{\Omega}_i^+ - \boldsymbol{\Omega}_i^-) J_d\right) \cdot \delta q = 0.$$

One immediate solution to this condition is letting $(\dot{x}_i^+, \Omega_i^+) = (\dot{x}_i^-, \Omega_i^-)$. However, this 524would cause the system to leave the admissible set C, so we will look for other solutions by 525considering the possible variations on the tangent space of the boundary point $q(t_i)$. In order 526 to accomplish this, we will consider a local representation of the boundary $\partial C = \Phi^{-1}(0)$, 527where 0 is a regular value of the collision detection function Φ . From the Submersion Level 528529 Set Theorem, we obtain

530 (4.13)
$$T_{q(t_i)}\partial C = \left\{ (\delta \boldsymbol{x}, R_i \eta) \mid \left(\frac{\partial \Phi_i}{\partial \boldsymbol{x}}, \frac{\partial \Phi_i}{\partial R} \right) \cdot (\delta \boldsymbol{x}, R_i \eta) = 0 \right\},$$

where 531

532 (4.14)
$$\left(\frac{\partial \Phi_i}{\partial \boldsymbol{x}}, \frac{\partial \Phi_i}{\partial R}\right) = \left(\frac{\partial \Phi}{\partial \boldsymbol{x}}, \frac{\partial \Phi}{\partial R}\right)\Big|_{q(t_i)}$$

We compute $\left(\frac{\partial \Phi_i}{\partial \boldsymbol{x}}, \frac{\partial \Phi_i}{\partial R}\right) \cdot (\delta \boldsymbol{x}, R_i \eta) = \delta \boldsymbol{x}^T \frac{\partial \Phi_i}{\partial \boldsymbol{x}} + \operatorname{tr}\left[\eta^T R_i^T \frac{\partial \Phi_i}{\partial R}\right]$. Applying the argument used 533 to obtain (2.6), we have that 534

$$T_{q(t_i)}\partial C = \left\{ \left(\delta \boldsymbol{x}, R_i \eta \right) \mid \delta \boldsymbol{x}^T \frac{\partial \Phi_i}{\partial \boldsymbol{x}} + \frac{1}{2} \operatorname{tr} \left[\eta^T \operatorname{Asym} \left(R_i^T \frac{\partial \Phi_i}{\partial R} \right) \right] = 0 \right\}.$$

The tangent space can be further identified as a hyperplane in \mathbb{R}^6 . This involves finding 536 $\boldsymbol{\chi}_i \in \mathbb{R}^3$ such that $S(\boldsymbol{\chi}_i) = \operatorname{Asym}\left(R_i^T \frac{\partial \Phi_i}{\partial R}\right)$. Suppose that $R_i^T \frac{\partial \Phi_i}{\partial R}$ is given, then $\boldsymbol{\chi}_i$ can be 537 computed using the inverse of the skew map defined in (2.2). It can also be defined using the rows of R_i and $\frac{\partial \Phi_i}{\partial R}$. More explicitly, start by partitioning R_i and $\frac{\partial \Phi_i}{\partial R}$ into row vectors. Let $\mathbf{r}_{i_1}, \mathbf{r}_{i_2}, \mathbf{r}_{i_3} \in S^2$ and $\phi_{i_1}, \phi_{i_2}, \phi_{i_3} \in \mathbb{R}^3$ be the successive columns of the rotation matrix R_i^T and $\frac{\partial \Phi_i}{\partial R}$, respectively. Then, 538 539540541

542
$$S(\boldsymbol{\chi}_i) = R_i^T \frac{\partial \Phi_i}{\partial R} - \frac{\partial \Phi_i}{\partial R}^T R_i$$

543
$$= \begin{bmatrix} \boldsymbol{r}_{i_1} & \boldsymbol{r}_{i_2} & \boldsymbol{r}_{i_3} \end{bmatrix} \begin{bmatrix} \boldsymbol{\phi}_{i_1}^T \\ \boldsymbol{\phi}_{i_2}^T \\ \boldsymbol{\phi}_{i_3}^T \end{bmatrix} - \begin{bmatrix} \boldsymbol{\phi}_{i_1} & \boldsymbol{\phi}_{i_2} & \boldsymbol{\phi}_{i_3} \end{bmatrix} \begin{bmatrix} \boldsymbol{r}_{i_1}^T \\ \boldsymbol{r}_{i_2}^T \\ \boldsymbol{r}_{i_3}^T \end{bmatrix}$$

544
$$= (\mathbf{r}_{i_1}\boldsymbol{\phi}_{i_1}^T - \boldsymbol{\phi}_{i_1}\mathbf{r}_{i_1}^T) + (\mathbf{r}_{i_2}\boldsymbol{\phi}_{i_2}^T - \boldsymbol{\phi}_{i_2}\mathbf{r}_{i_2}^T) + (\mathbf{r}_{i_3}\boldsymbol{\phi}_{i_3}^T - \boldsymbol{\phi}_{i_3}\mathbf{r}_{i_3}^T)$$

545
$$= S(\boldsymbol{\phi}_{i_1} \times \mathbf{r}_{i_1} + \boldsymbol{\phi}_{i_2} \times \mathbf{r}_{i_2} + \boldsymbol{\phi}_{i_3} \times \mathbf{r}_{i_3}),$$

545
$$= S(\boldsymbol{\phi}_{i_1} \times \boldsymbol{r}_{i_1} + \boldsymbol{\phi}_{i_2} \times \boldsymbol{r}_{i_2} + \boldsymbol{\phi}_{i_3} \times \boldsymbol{r}_{i_3})$$

where (2.1c) is used. Since S is invertible, 546

547 (4.15)
$$\chi_i = \phi_{i_1} \times r_{i_1} + \phi_{i_2} \times r_{i_2} + \phi_{i_3} \times r_{i_3}$$

548

535

Lemma 4.1. Let $q(t_i) = (\boldsymbol{x}_i, R_i) \in \partial C$, then 549

550 (4.16)
$$T_{q(t_i)}\partial C = \left\{ \left(\delta \boldsymbol{x}, R_i S(\boldsymbol{\zeta}) \right) \mid \delta \boldsymbol{x}^T \frac{\partial \Phi_i}{\partial \boldsymbol{x}} + \boldsymbol{\zeta}^T \boldsymbol{\chi}_i = 0 \right\},$$

where $\boldsymbol{\chi}_i$ is defined by (4.15). 551

Proof. Identify the elements of tangent space as $(\delta \boldsymbol{x}, R_i \eta) = (\delta \boldsymbol{x}, R_i S(\boldsymbol{\zeta}))$ where $\boldsymbol{\zeta} \in \mathbb{R}^3$. 552Since Asym $\left(R_i^T \frac{\partial \Phi_i}{\partial R}\right) = S(\boldsymbol{\chi}_i)$, we write $\frac{1}{2} \operatorname{tr} \left[\eta^T \operatorname{Asym} \left(R_i^T \frac{\partial \Phi_i}{\partial R}\right)\right] = \frac{1}{2} \operatorname{tr} [S(\boldsymbol{\zeta})^T S(\boldsymbol{\chi}_i)] = \boldsymbol{\zeta}^T \boldsymbol{\chi}_i$, 553where we used the induced inner product of \mathbb{R}^3 given in (2.7). 554

Remark 4.2. $T_{q(t_i)} \partial C$ can be identified with a hyperplane in \mathbb{R}^6 defined by 555

556 (4.17)
$$\mathcal{P}_{\left(\frac{\partial \Phi_i}{\partial \boldsymbol{x}} \; \boldsymbol{\chi}_i\right)} = \left\{ z \in \mathbb{R}^6 \; \middle| \; \left(\frac{\partial \Phi_i}{\partial \boldsymbol{x}}^T \; \boldsymbol{\chi}_i^T\right) z = 0 \right\}.$$

Theorem 4.3. Suppose $q(t_i) \in \partial C$. Then $\left(m(\dot{x}_i^+ - \dot{x}_i^-), R_i S(\Omega_i^+ - \Omega_i^-) J_d\right) \cdot \delta q = 0$, for all 557 $\delta q \in T_{q(t_i)} \partial C$, if and only if the first jump conditions, 558

559 (4.18a)
$$m(\dot{\boldsymbol{x}}_i^+ - \dot{\boldsymbol{x}}_i^-) = \lambda \frac{\partial \Phi_i}{\partial \boldsymbol{x}},$$

560 (4.18b)
$$J(\mathbf{\Omega}_i^+ - \mathbf{\Omega}_i^-) = \lambda \boldsymbol{\chi}_i,$$

561 are satisfied for some $\lambda \in \mathbb{R} \setminus \{0\}$.

562 *Proof.* Let
$$\delta q = (\delta \boldsymbol{x}, R_i S(\boldsymbol{\zeta}))$$
, and we compute the expression

$$\begin{cases} 663 \qquad \left(m(\dot{\boldsymbol{x}}_{i}^{+}-\dot{\boldsymbol{x}}_{i}^{-}), R_{i}S(\boldsymbol{\Omega}_{i}^{+}-\boldsymbol{\Omega}_{i}^{-})J_{d}\right) \cdot \delta q = \delta \boldsymbol{x}^{T} \left\{m(\dot{\boldsymbol{x}}_{i}^{+}-\dot{\boldsymbol{x}}_{i}^{-})\right\} + \operatorname{tr}[S(\boldsymbol{\zeta})^{T}S(\boldsymbol{\Omega}_{i}^{+}-\boldsymbol{\Omega}_{i}^{-})J_{d}] \\ = \delta \boldsymbol{x}^{T} \left\{m(\dot{\boldsymbol{x}}_{i}^{+}-\dot{\boldsymbol{x}}_{i}^{-})\right\} \end{cases}$$

+ $\frac{1}{2}$ tr[$S(\boldsymbol{\zeta})^T$ Asym($S(\boldsymbol{\Omega}_i^+ - \boldsymbol{\Omega}_i^-)J_d$)].

566 Using (4.2), Asym $(S(\mathbf{\Omega}_i^+ - \mathbf{\Omega}_i^-)J_d)) = S(J(\mathbf{\Omega}_i^+ - \mathbf{\Omega}_i^-))$, so the right-hand side becomes 567 $\delta \mathbf{x}^T \{m(\dot{\mathbf{x}}_i^+ - \dot{\mathbf{x}}_i^-)\} + \boldsymbol{\zeta}^T \{J(\mathbf{\Omega}_i^+ - \mathbf{\Omega}_i^-)\}$. If the jump conditions hold, the expression be-568 comes $\lambda \left(\delta \mathbf{x}^T \frac{\partial \Phi_i}{\partial \mathbf{x}} + \boldsymbol{\zeta}^T \boldsymbol{\chi}_i\right)$, which vanishes by Lemma 4.1. If we assume that the expression 569 vanishes, the curly brackets must be a nonzero scaling of the normal vector $\left(\frac{\partial \Phi_i}{\partial \mathbf{x}} \, \boldsymbol{\chi}_i\right) \in \mathbb{R}^6$ by 570 Remark 4.2, which yield the jump conditions.

We consider the second jump condition (3.10b), which is a statement of the conservation of energy. Recall that $E = \frac{\partial L}{\partial \dot{q}} \cdot \dot{q} - L$ and $\frac{\partial L}{\partial \dot{q}} = (m\dot{x}, \dot{R}J_d)$, so $\frac{\partial L}{\partial \dot{q}} \cdot \dot{q} = (m\dot{x}, \dot{R}J_d) \cdot (\dot{x}, \dot{R}) =$ $m \|\dot{x}\|^2 + tr[\dot{R}J_d\dot{R}^T]$. Therefore, the energy may be written as

574 (4.19a)
$$E = \frac{1}{2}m\|\dot{\boldsymbol{x}}\|^2 + \frac{1}{2}\operatorname{tr}[\dot{R}J_d\dot{R}^T] + mg\boldsymbol{e}_{\boldsymbol{3}}^T\boldsymbol{x},$$

575 (4.19b)
$$= \frac{1}{2}m\|\dot{\boldsymbol{x}}\|^2 + \frac{1}{2}\operatorname{tr}[S(\boldsymbol{\Omega})J_dS(\boldsymbol{\Omega})^T] + mg\boldsymbol{e}_{\boldsymbol{3}}^T\boldsymbol{x}$$

576 (4.19c)
$$= \frac{1}{2}m\|\dot{\boldsymbol{x}}\|^2 + \frac{1}{2}\boldsymbol{\Omega}^T J\boldsymbol{\Omega} + mg\boldsymbol{e}_{\boldsymbol{3}}^T \boldsymbol{x}.$$

The second jump condition for our system is given by $0 = E(q(t_i^+), \dot{q}(t_i^+)) - E(q(t_i^-), \dot{q}(t_i^-))$. Using (4.19c), the full set of jump conditions become

$$F_{\text{Jump}}: (\boldsymbol{x}_{i}, \boldsymbol{R}_{i}, \dot{\boldsymbol{x}}_{i}^{-}, \boldsymbol{\Omega}_{i}^{-}) \rightarrow (\lambda, \dot{\boldsymbol{x}}_{i}^{+}, \boldsymbol{\Omega}_{i}^{+})$$

$$(4.20a) \qquad m \dot{\boldsymbol{x}}_{i}^{+} = m \dot{\boldsymbol{x}}_{i}^{-} + \lambda \frac{\partial \Phi_{i}}{\partial \boldsymbol{x}},$$

$$(4.20b) \qquad J \boldsymbol{\Omega}_{i}^{+} = J \boldsymbol{\Omega}_{i}^{-} + \lambda \boldsymbol{\chi}_{i},$$

$$(4.20c) \qquad 0 = \frac{1}{2} m(\| \dot{\boldsymbol{x}}_{i}^{+} \|^{2} - \| \dot{\boldsymbol{x}}_{i}^{-} \|^{2}) + \frac{1}{2} \left(\boldsymbol{\Omega}_{i}^{+T} J \boldsymbol{\Omega}_{i}^{+} - \boldsymbol{\Omega}_{i}^{-T} J \boldsymbol{\Omega}_{i}^{-} \right).$$

579 Denote the solution to $(\dot{x}_i^+, \Omega_i^+)$ with $\lambda \neq 0$ as a discrete map F_{Jump} with the necessary 580 arguments above. In particular, λ is obtained by substituting $(\dot{x}_i^+, \Omega_i^+)$ from (4.20a) and 581 (4.20b) into (4.20c), which gives a quadratic equation for the variable λ . One root will always 582 be $\lambda = 0$, which is omitted. Then, there is a unique nonzero root λ , which gives $(\dot{x}_i^+, \Omega_i^+)$.

4.2.1. Jump Conditions: Hamiltonian Form. It is actually natural to express the jump conditions on the Hamiltonian side since both conservation of energy and conservation of momentum can easily be described on the cotangent bundle. We still write $(\boldsymbol{x}_i, R_i) \in \partial C$ as the configuration at impact. Denote the instantaneous linear and angular momentum before and after impact as $\gamma^{\pm} = m\dot{\boldsymbol{x}}^{\pm}$ and $\Pi^{\pm} = J\Omega^{\pm}$, respectively. Suppose that J is invertible, then we obtain the following result.

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589 Corollary 4.4. Given $(\boldsymbol{x}_i, R_i) \in \partial C$ and the linear and angular momentum before impact 590 $(\boldsymbol{\gamma}^-, \boldsymbol{\Pi}^-)$, there is a unique $\lambda \in \mathbb{R} \setminus \{0\}$ and $(\boldsymbol{\gamma}^+, \boldsymbol{\Pi}^+)$ satisfying

$$\tilde{F}_{\text{Jump}} : (\boldsymbol{x}_{i}, R_{i}, \boldsymbol{\gamma}^{-}, \boldsymbol{\Pi}^{-}) \to (\lambda, \boldsymbol{\gamma}^{+}, \boldsymbol{\Pi}^{+})
(4.21a) \qquad \boldsymbol{\gamma}^{+} = \boldsymbol{\gamma}^{-} + \lambda \frac{\partial \Phi_{i}}{\partial \boldsymbol{x}},
(4.21b) \qquad \boldsymbol{\Pi}^{+} = \boldsymbol{\Pi}^{-} + \lambda \boldsymbol{\chi}_{i},
(4.21c) \qquad \boldsymbol{0} = \frac{1}{2m} (\|\boldsymbol{\gamma}^{+}\|^{2} - \|\boldsymbol{\gamma}^{-}\|^{2}) + \frac{1}{2} (\boldsymbol{\Pi}^{+T} J^{-1} \boldsymbol{\Pi}^{+} - \boldsymbol{\Pi}^{-T} J^{-1} \boldsymbol{\Pi}^{-}),$$

591 where Φ is the collision detection function and χ_i is defined by (4.15).

592 As before, (γ^+, Π^+) is obtained by first solving a quadratic equation for the nonzero root λ .

593 **5.** Lie Group Variational Collision Integrators for the Bouncing Ellipsoid. For the dis-594 crete setting, we also follow the approach described in [22]. This involves the construction of 595 the discrete Lagrangian by approximating a segment of the action integral via the trapezoidal 596 rule. However, we first approximate the linear and angular velocity for a segment of the action 597 integral. We introduce the auxiliary variable $F_k \in SO(3)$ so that

599 Note that F_k represents the relative attitude between R_k and R_{k+1} , and it is guaranteed that

the attitude volves on SO(3) since $F_k \in SO(3)$. Now, using $\dot{R} = RS(\Omega)$, approximate the k-th angular velocity as

602 (5.1)
$$S(\mathbf{\Omega}_k) = R_k^T \dot{R}_k \approx R_k^T \frac{R_{k+1} - R_k}{h} = \frac{1}{h} (F_k - I_3).$$

603 The linear velocity $\dot{\boldsymbol{x}}_k$ is approximated by $(\boldsymbol{x}_{k+1} - \boldsymbol{x}_k)/h$. Substitute the approximations 604 above into the modified Lagrangian (4.4b), so the approximation of the kinetic term becomes

605
$$T(\dot{\boldsymbol{x}}, \boldsymbol{\Omega}) \approx T((\boldsymbol{x}_{k+1} - \boldsymbol{x}_k)/h, (F_k - I_3)/h)$$

$$egin{aligned} & & = 1 \left((m{x}_{k+1} - m{x}_k) / h, (F_k - I_3) / h
ight) \ & = rac{1}{2h^2} \|m{x}_{k+1} - m{x}_k\|^2 + rac{1}{h^2} \operatorname{tr}\left[(I_3 - F_k) J_d
ight]. \end{aligned}$$

607 We write the discrete Lagrangian, where $F_k = R_k^T R_{k+1}$,

$$L_d(\boldsymbol{x}_k, R_k, \boldsymbol{x}_{k+1}, F_k)$$

$$= \frac{h}{2} \left[L\left(\boldsymbol{x}_k, \frac{\boldsymbol{x}_{k+1} - \boldsymbol{x}_k}{h}, R_k, \frac{F_k - I_3}{h}\right) + L\left(\boldsymbol{x}_{k+1}, \frac{\boldsymbol{x}_{k+1} - \boldsymbol{x}_k}{h}, R_{k+1}, \frac{F_k - I_3}{h}\right) \right]$$

$$= \frac{1}{2h} m \|\boldsymbol{x}_{k+1} - \boldsymbol{x}_k\|^2 + \frac{1}{h} \operatorname{tr}[(I_3 - F_k)J_d] - \frac{1}{2} mghe_{\mathbf{3}}^T(\boldsymbol{x}_{k+1} + \boldsymbol{x}_k).$$

5.1. Discrete Equations of Motion. Unlike the Lagrangian, we write the discrete Lagrangian in only one way. Hence, we will show the discrete equations of motion away from the point of impact directly from the first result (3.17) of Theorem 3.2. More specifically, the result is obtained by taking the variations of the discrete variables on the discrete action sum and applying the discrete Hamilton's principle. 614 **5.1.1. Lagrangian Form.** Let $q_k = (\boldsymbol{x}_k, R_k)$ for $k \in \{0, \dots, i-2, i+1, \dots, N\}$. Consider 615 the following variations of the discrete variables. Namely, the variation $\delta \boldsymbol{x}_k \in \mathbb{R}^3$ of \boldsymbol{x}_k which 616 vanishes at k = 0 and k = N. The variation of R_k is given by

617 (5.3)
$$\delta R_k = R_k \eta_k$$

618 where $\eta_k \in \mathfrak{so}(3)$ and which also vanishes at k = 0 and k = N.

619 Recall that $F_k = R_k^T R_{k+1}$ and write $\delta q_k = (\delta \boldsymbol{x}_k, R_k \eta_k)$. The following identities are 620 derived by using (2.6) and matrix derivatives (see Appendix C),

621 (5.4)
$$D_2L_d(q_{k-1}, q_k, h) \cdot \delta q_k = \left(\frac{1}{h}m(\boldsymbol{x}_k - \boldsymbol{x}_{k-1}) - \frac{1}{2}mgh\boldsymbol{e}_3, -\frac{1}{h}\operatorname{Asym}(F_{k-1}^T J_d)\right) \cdot S(\delta \boldsymbol{x}_k, \eta_k),$$

622 (5.5)
$$D_1 L_d(q_k, q_{k+1}, h) \cdot \delta q_k = \left(\frac{1}{h}m(\boldsymbol{x}_k - \boldsymbol{x}_{k+1}) - \frac{1}{2}mgh\boldsymbol{e}_3, -\frac{1}{h}\operatorname{Asym}(F_k J_d)\right) \cdot S(\delta \boldsymbol{x}_k, \eta_k).$$

623 Note that $-\operatorname{Asym}(F_{k-1}^T J_d) = \operatorname{Asym}(J_d F_{k-1})$, and by the discrete Euler-Lagrange equation 624 (3.17), their sum vanishes,

625
$$0 = [D_2 L_d(q_{k-1}, q_k, h) + D_1 L_d(q_k, q_{k+1}, h)] \cdot \delta q_k$$

626
$$= -\delta \boldsymbol{x}^T \left\{ \frac{1}{h} m(\boldsymbol{x}_{k+1} - 2\boldsymbol{x}_k + \boldsymbol{x}_{k-1}) + mgh\boldsymbol{e_3} \right\} + \frac{1}{2} \operatorname{tr} \left[\eta_k^T \left\{ \frac{1}{h} \operatorname{Asym}(J_d F_{k-1} - F_k J_d) \right\} \right].$$

627 Therefore, the equation above holds, for all $k = 1, \ldots, i - 2, i + 1, \ldots, N$, if and only if the

628 expressions in the curly brackets vanish. We arrive at the discrete equations of motion in 629 Lagrangian form,

$$F_{L_d}[h,h] : (\boldsymbol{x}_{k-1}, R_{k-1}, \boldsymbol{x}_k, R_k) \mapsto (\boldsymbol{x}_k, R_k, \boldsymbol{x}_{k+1}, R_{k+1})$$
(5.6a)

$$0 = \frac{1}{h}m(\boldsymbol{x}_{k+1} - 2\boldsymbol{x}_k + \boldsymbol{x}_{k-1}) + mgh\boldsymbol{e_3},$$
(5.6b)

$$0 = \frac{1}{h}\operatorname{Asym}(J_dF_{k-1} - F_kJ_d),$$
(5.6c)

$$R_{k+1} = R_kF_k.$$

630 This gives the discrete Lagrangian map FL_d with the following parameters [h, h]: The 631 first h indicates the timestep for the interval $[t_{k-1}, t_k]$ with the corresponding configurations 632 $(\boldsymbol{x}_{k-1}, R_{k-1})$ at time t_{k-1} and (\boldsymbol{x}_k, R_k) at time t_k ; the second h indicates the timestep for 633 $[t_k, t_{k+1}]$ with its corresponding configurations (\boldsymbol{x}_k, R_k) at t_k and $(\boldsymbol{x}_{k+1}, R_{k+1})$ at t_{k+1} .

To compute the map, solve for the x_{k+1} from (5.6a). F_k is obtained next using the second, implicit equation (5.6b) where $F_{k-1} = R_{k-1}^T R_k$ (see Appendix D). We emphasize that such an implicit equation shows up in all of the discrete maps that follow, and methods for solving such equations are discussed in Appendix D. Finally, R_{k+1} is updated using (5.6c).

638 **5.1.2. Hamiltonian Form.** Using the discrete Legendre transforms, we will arrive at an-639 other set of discrete equations of motion based on discrete positions and momenta, which we 640 will collectively call *states*. Denote the linear and angular momentum by $p_k = (\gamma_k, S(\mathbf{\Pi}_k))$ 641 in analogy to the continuous case. Using Theorem 3.2 and (3.22) for the discrete Legendre 642 transforms, we see that

643 (5.7)
$$(\boldsymbol{\gamma}_k, S(\boldsymbol{\Pi}_k)) \cdot_S (\delta \boldsymbol{x}_k, \eta_k) = -D_1 L_d(q_k, q_{k+1}, h) \cdot \delta q_k$$
$$= (\frac{1}{h} m(\boldsymbol{x}_{k+1} - \boldsymbol{x}_k) + \frac{h}{2} m g \boldsymbol{e}_3, \frac{1}{h} \operatorname{Asym}(F_k J_d)) \cdot_S (\delta \boldsymbol{x}_k, \eta_k),$$

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644 (5.8)
$$(\boldsymbol{\gamma}_{k+1}, S(\boldsymbol{\Pi}_{k+1})) \cdot_S (\delta \boldsymbol{x}_{k+1}, \eta_{k+1}) = D_2 L_d(q_k, q_{k+1}, h) \cdot \delta q_{k+1}$$
$$= (\frac{1}{h} m(\boldsymbol{x}_{k+1} - \boldsymbol{x}_k) - \frac{h}{2} m g \boldsymbol{e_3}, \frac{1}{h} \operatorname{Asym}(J_d F_k)) \cdot_S (\delta \boldsymbol{x}_{k+1}, \eta_{k+1}).$$

Hence, \boldsymbol{x}_{k+1} and $\boldsymbol{\gamma}_{k+1}$ can be expressed in terms of \boldsymbol{x}_k and $\boldsymbol{\gamma}_k$ from the equations arising from the first components of (5.7) and (5.8), respectively. We also have $S(\boldsymbol{\Pi}_k) = \frac{1}{h} \operatorname{Asym}(F_k J_d)$,

647 and so

$$S(\mathbf{\Pi}_{k+1}) = \frac{1}{h}\operatorname{Asym}(J_d F_k) = F_k^T \frac{1}{h}\operatorname{Asym}(F_k J_d)F_k = F_k^T S(\mathbf{\Pi}_k)F_k = S(F_k^T \mathbf{\Pi}_k),$$

using property (2.1d) of the skew map. As a result, the discrete equations of motion in
 Hamiltonian form are given by

$$\begin{split} \tilde{F}_{L_d}[h] : (\boldsymbol{x}_k, R_k, \boldsymbol{\gamma}_k, \boldsymbol{\Pi}_k) \mapsto (\boldsymbol{x}_{k+1}, R_{k+1}, \boldsymbol{\gamma}_{k+1}, \boldsymbol{\Pi}_{k+1}) \\ (5.9a) \qquad \qquad \boldsymbol{x}_{k+1} = \boldsymbol{x}_k + \frac{h}{m} \boldsymbol{\gamma}_k - \frac{1}{2} g h^2 \boldsymbol{e_3}, \\ (5.9b) \qquad \qquad \boldsymbol{\gamma}_{k+1} = \boldsymbol{\gamma}_k - mgh \boldsymbol{e_3}, \end{split}$$

(5.9c)
$$S(\mathbf{\Pi}_k) = \frac{1}{h} \operatorname{Asym}(F_k J_d),$$

(5.9d)
$$\mathbf{\Pi}_{k+1} = F_k^{\,\mathcal{I}} \,\mathbf{\Pi}_k$$

$$(5.9e) R_{k+1} = R_k F_k$$

This gives a discrete Hamiltonian map \tilde{F}_{L_d} with the following parameter [h]: This hindicates the timestep for the interval $[t_k, t_{k+1}]$ with the corresponding states $(\boldsymbol{x}_k, R_k, \boldsymbol{\gamma}_k, \boldsymbol{\Pi}_k)$ at time t_k and $(\boldsymbol{x}_{k+1}, R_{k+1}, \boldsymbol{\gamma}_{k+1}, \boldsymbol{\Pi}_{k+1})$ at time t_{k+1} .

654 Similar to the Lagrangian form, x_{k+1} and γ_{k+1} can be computed using (5.9a) and (5.9b), 655 respectively. Compute F_k from the implicit equation (5.9c), which is used to update Π_{k+1} 656 and R_{k+1} using (5.9d) and (5.9e).

657 **5.2.** Impact Point and Time. Recall the definition of the collision detection function Φ 658 from (2.16), which allows us to detect collisions in the system. For each integration step 659 discussed in Subsection 5.1, $(\boldsymbol{x}_{k+1}, R_{k+1})$ are computed. Hence, one may check for interpen-660 etration after each integration by evaluating $\Phi(\boldsymbol{x}_{k+1}, R_{k+1})$.

If the signed distance is positive, then we proceed to the next integration step. If the signed distance is zero, then the current configuration and time is the impact point and time, and we will have to apply the discrete jump conditions. If the evaluation is negative, then interpenetration has occurred, the current integration step is discarded, and so we consider (3.18) of Theorem 3.2, and we attempt to resolve the impact point and time.

666 **5.2.1. Lagrangian Form.** Note that the impact point would occur at time $\tilde{t} = t_{i-1} + \alpha h$ 667 for some $\alpha \in (0, 1)$. Similarly, we rewrite (3.18),

668
$$0 = [D_2 L_d(q_{i-2}, q_{i-1}, h) + D_1 L_d(q_{i-1}, \tilde{q}, \alpha h)] \cdot \delta q_{i-1},$$

669 as follows,

$$F_{L_d}^{\text{Impact}}[h, \alpha h] : (\boldsymbol{x}_{i-2}, R_{i-2}, \boldsymbol{x}_{i-1}, R_{i-1}) \mapsto (\boldsymbol{x}_{i-1}, R_{i-1}, \tilde{\boldsymbol{x}}, \tilde{R})$$
(5.10a) $0 = \frac{1}{h}m(\boldsymbol{x}_{i-1} - \boldsymbol{x}_{i-2}) - \frac{1}{\alpha h}m(\tilde{\boldsymbol{x}} - \boldsymbol{x}_{i-1}) - \frac{1}{2}mg(1+\alpha)h\boldsymbol{e_3},$
(5.10b) $0 = \text{Asym}\left(\frac{1}{h}J_dF_{i-2} - \frac{1}{\alpha h}F_{i-1}J_d\right),$
(5.10c) $\tilde{R} = R_{i-1}F_{i-1},$
(5.10d) $0 = \Phi(\tilde{\boldsymbol{x}}, \tilde{R}).$

670

We compute the solution using the bisection method to solve for $\alpha \in (0,1)$. There is a unique α in the open interval such that $\Phi(\tilde{x}, \tilde{R}) = 0$. This is the case because Φ is defined to 671 be positive when the ellipsoid is above the plane and negative when the ellipsoid is interpen-672 etrating or below the plane. For the implementation, α is taken as the center of the initial 673 interval, and $(\tilde{\boldsymbol{x}}, \tilde{R})$ are solved in the same way as F_{L_d} . Then, check the sign of $\Phi(\tilde{\boldsymbol{x}}, \tilde{R})$ and as 674per the bisection algorithm, chose the left-half interval if $\Phi(\tilde{x}, \tilde{R}) < 0$, or the right-half interval 675 if $\Phi(\tilde{x}, R) > 0$. The process repeats with α given by the center of the chosen half-interval, 676 and it terminates when $\Phi(\tilde{\boldsymbol{x}}, R)$ is sufficiently small. 677

678 **5.2.2. Hamiltonian Form.** On the Hamiltonian side, we compute the following discrete Legendre transforms, 679

680
$$(\boldsymbol{\gamma}_{i-1}, S(\boldsymbol{\Pi}_{i-1})) \cdot_S (\delta \boldsymbol{x}_{i-1}, \eta_{i-1}) = -D_1 L_d(q_{i-1}, \tilde{q}, \alpha h) \cdot \delta q_{i-1},$$

681
$$(\tilde{\boldsymbol{\gamma}}, S(\tilde{\boldsymbol{\Pi}})) \cdot_S (\delta \tilde{\boldsymbol{x}}, \tilde{\boldsymbol{\eta}}) = D_2 L_d(q_{i-1}, \tilde{\boldsymbol{q}}, \alpha h) \cdot \delta \tilde{\boldsymbol{q}}.$$

We obtain the equations for the impact point in Hamiltonian form, 682

$\tilde{F}_{L_d}^{\mathrm{Impa}}$	^{ct} [αh]: $(\boldsymbol{x}_{i-1}, R_{i-1}, \boldsymbol{\gamma}_{i-1}, \boldsymbol{\Pi}_{i-1}) \mapsto (\tilde{\boldsymbol{x}}, \tilde{\boldsymbol{R}}, \tilde{\boldsymbol{\gamma}}, \tilde{\boldsymbol{\Pi}})$
(5.11a)	$ ilde{oldsymbol{x}} = oldsymbol{x}_{i-1} + rac{lpha h}{m}oldsymbol{\gamma}_{i-1} - rac{1}{2}glpha^2h^2oldsymbol{e_3},$
(5.11b)	$ ilde{oldsymbol{\gamma}} = oldsymbol{\gamma}_{i-1} - mglpha h oldsymbol{e_3},$
(5.11c)	$S(\mathbf{\Pi}_{i-1}) = \frac{1}{\alpha h} \operatorname{Asym}(F_{i-1}J_d),$
(5.11d)	$\tilde{\mathbf{\Pi}} = F_{i-1}^T \mathbf{\Pi}_{i-1},$
(5.11e)	$\tilde{R} = R_{i-1}F_{i-1},$
(5.11f)	$0 = \Phi(\tilde{\boldsymbol{x}}, \tilde{R}).$

The $\alpha \in (0,1)$ and the impact states $(\tilde{x}, R, \tilde{\gamma}, \Pi)$ are solved for using the bisection method, 683 as before. 684

5.3. Single Impact. We consider the next integration step, and this would give us the 685discrete configuration after the configuration at impact. In this subsection, assume that there 686 is one collision in the time interval (t_{i-1}, t_i) occurring at time $\tilde{t} = t_{i-1} + \alpha h$. Therefore, our 687 next discrete configuration occurs at time $t_i = \tilde{t} + (1 - \alpha)h$. 688

5.3.1. Lagrangian Form. The equation (3.19) of Theorem 3.2 is used for our next integration step. Recall the discrete energy from (3.20), which gives

691 (5.12)
$$E_d(q_k, q_{k+1}, h) = \frac{1}{2h^2} m \|\boldsymbol{x}_{k+1} - \boldsymbol{x}_k\|^2 + \frac{1}{h^2} \operatorname{tr}[(I_3 - F_k)J_d] + \frac{1}{2} m g \boldsymbol{e}_3^T(\boldsymbol{x}_{k+1} + \boldsymbol{x}_k),$$

and so (3.19a) can be easily written. In addition, the conservation of discrete momentum described with the cotangent lift in (3.19b) is written in terms of a local representation of $\partial C = \Phi^{-1}(0)$. This is the same constraint formulation as in the continuous case, so we write our next set of equations after computing

696
$$0 = i^* \left(D_2 L_d(q_{i-1}, \tilde{q}, \alpha h) + D_1 L_d(\tilde{q}, q_i, (1-\alpha)h) \right) \cdot \delta \tilde{q}.$$

697 On the Lagrangian side, we obtain

$$F_{L_d}^{i}[\alpha h, (1-\alpha)h][\lambda] : (\boldsymbol{x}_{i-1}, R_{i-1}, \tilde{\boldsymbol{x}}, \tilde{R}) \mapsto (\tilde{\boldsymbol{x}}, \tilde{R}, \boldsymbol{x}_i, R_i)$$
(5.13a)
$$0 = E_d(\tilde{q}, q_i, (1-\alpha)h) - E_d(q_{i-1}, \tilde{q}, \alpha h),$$
(5.13b)
$$0 = \frac{1}{\alpha h} m(\tilde{\boldsymbol{x}} - \boldsymbol{x}_{i-1}) - \frac{1}{(1-\alpha)h} m(\boldsymbol{x}_i - \tilde{\boldsymbol{x}}) - \frac{1}{2} mgh \boldsymbol{e_3} + \lambda \frac{\partial \tilde{\Phi}}{\partial \boldsymbol{x}},$$
(5.13c)
$$0 = \operatorname{Asym}\left(\frac{1}{\alpha h} J_d F_{i-1} - \frac{1}{(1-\alpha)h} \tilde{F} J_d\right) + \lambda \operatorname{Asym}\left(\tilde{R}^T \frac{\partial \tilde{\Phi}}{\partial R}\right),$$
(5.13d)
$$R_i = \tilde{R} \tilde{F}.$$

698

Observe that $\lambda \neq 0$ and

699 (5.14)
$$\left(\frac{\partial \tilde{\Phi}}{\partial \boldsymbol{x}}, \frac{\partial \tilde{\Phi}}{\partial R} \right) = \left(\frac{\partial \Phi}{\partial \boldsymbol{x}}, \frac{\partial \Phi}{\partial R} \right) \Big|_{(\tilde{\boldsymbol{x}}, \tilde{R})} = \left(\boldsymbol{n}, -\frac{\boldsymbol{n} \boldsymbol{n}^T \tilde{R} I_{\epsilon}^2}{\|I_{\epsilon} \tilde{R}^T \boldsymbol{n}\|} \right).$$

Furthermore, $[\lambda]$ indicates the requirement to compute it first. In fact, this is the same λ as in Theorem 4.3 in the continuous case. However, note that determining λ on the Lagrangian side can be difficult since we do not have information on the instantaneous linear and angular velocity $(\dot{\boldsymbol{x}}^-, \boldsymbol{\Omega}^-)$ before the impact. Therefore, the solution of λ is discussed on the Hamiltonian side, which follows easily from Corollary 4.4. Once λ is determined, (\boldsymbol{x}_i, R_i) can be solved similarly to the previous discrete Lagrangian maps, and one can verify that (5.13a) is satisfied.

Finally, we write the last set of equations from (3.21) by first computing

708
$$0 = [D_2 L_d(\tilde{q}, q_i, (1 - \alpha)h) + D_1 L_d(q_i, q_{i+1}, h)] \cdot \delta q_i.$$

709 In Lagrangian form, we get

$$F_{L_d}^{i+1}[(1-\alpha)h,h]:(\tilde{\boldsymbol{x}},\tilde{R},\boldsymbol{x}_i,R_i)\mapsto(\boldsymbol{x}_i,R_i,\boldsymbol{x}_{i+1},R_{i+1})$$

(5.15a)
$$0 = \frac{1}{(1-\alpha)h}m(\boldsymbol{x}_i - \tilde{\boldsymbol{x}}) - \frac{1}{h}m(\boldsymbol{x}_{i+1} - \boldsymbol{x}_i) - \frac{1}{2}mg(2-\alpha)h\boldsymbol{e_3},$$

(5.15b)
$$0 = \operatorname{Asym}\left(\frac{1}{(1-\alpha)h}J_d\tilde{F} - \frac{1}{h}F_iJ_d\right),$$

(5.15c)
$$R_{i+1} = R_iF_i.$$

5.3.2. Hamiltonian Form. Again, we compute the discrete Legendre transforms,

711
$$(\tilde{\boldsymbol{\gamma}}, S(\boldsymbol{\Pi})) \cdot_S (\delta \tilde{\boldsymbol{x}}, \tilde{\eta}) = -D_1 L_d(\tilde{q}, q_i, (1-\alpha)h) \cdot \delta \tilde{q},$$

712
$$(\boldsymbol{\gamma}_i, S(\boldsymbol{\Pi}_i)) \cdot_S (\delta \boldsymbol{x}_i, \eta_i) = D_2 L_d(\tilde{q}, q_i, (1-\alpha)h)) \cdot \delta q_i.$$

713 We obtain the equations in Hamiltonian form,

$\tilde{F}^i_{L_d}[\alpha h][\lambda]$	$]:(ilde{oldsymbol{x}}, ilde{R}, ilde{oldsymbol{\gamma}}, ilde{oldsymbol{\Pi}})\mapsto(oldsymbol{x}_i,R_i,oldsymbol{\gamma}_i,oldsymbol{\Pi}_i)$
(5.16a)	$0 = E_d(\tilde{q}, q_i, (1 - \alpha)h) - E_d(q_{i-1}, \tilde{q}, \alpha h),$
(5.16b)	$oldsymbol{x}_i = oldsymbol{ ilde{x}} + rac{(1-lpha)h}{m}oldsymbol{ ilde{\gamma}} - rac{1}{2}g(1-lpha)^2h^2oldsymbol{e_3} + \lambdarac{(1-lpha)h}{m}rac{\partial ilde{\Phi}}{\partialoldsymbol{x}},$
(5.16c)	$\boldsymbol{\gamma}_i = \tilde{\boldsymbol{\gamma}} - mg(1-lpha)h\boldsymbol{e_3} + \lambda \frac{\partial \tilde{\Phi}}{\partial \boldsymbol{x}},$
(5.16d)	$S(\tilde{\mathbf{\Pi}}) = \operatorname{Asym}\left(\frac{1}{(1-\alpha)h}\tilde{F}J_d\right) - \lambda\operatorname{Asym}\left(\tilde{R}^T\frac{\partial\tilde{\Phi}}{\partial R}\right),$
(5.16e)	$\tilde{F}\boldsymbol{\Pi}_i = \tilde{\boldsymbol{\Pi}} + \lambda \tilde{\boldsymbol{\chi}},$
(5.16f)	$R_i = \tilde{R}\tilde{F}.$

714 Note that

715

$$S(\tilde{\boldsymbol{\chi}}) = \operatorname{Asym}\left(\tilde{R}^T \frac{\partial \tilde{\Phi}}{\partial R}\right)$$

,

where $\tilde{\boldsymbol{\chi}}$ can be computed by either using S^{-1} or (4.15). Similarly, $[\lambda]$ indicates that it needs to be solved first; to solve for λ , invoke Corollary 4.4 by setting $(\tilde{\boldsymbol{\chi}}, \tilde{R}) \in \partial C$ as the configuration at impact and letting $(\tilde{\boldsymbol{\gamma}}, \tilde{\boldsymbol{\Pi}}) = (\boldsymbol{\gamma}^-, \boldsymbol{\Pi}^-)$. In fact, this will not only solve for $\lambda \neq 0$ but also $(\boldsymbol{\gamma}^+, \boldsymbol{\Pi}^+)$, and this fact will be used to optimize our algorithm in the end of this section.

Lastly, we solve for the next set of states at time t_{i+1} by computing the next set of discrete Legendre transforms,

722
$$(\boldsymbol{\gamma}_i, S(\boldsymbol{\Pi}_i) \cdot_S (\delta \tilde{\boldsymbol{x}}, \tilde{\eta}) = -D_1 L_d(q_i, q_{i+1}, h) \cdot \delta q_i,$$

723
$$(\boldsymbol{\gamma}_{i+1}, S(\boldsymbol{\Pi}_{i+1})) \cdot_S (\delta \boldsymbol{x}_i, \eta_i) = D_2 L_d(q_i, q_{i+1}, h)) \cdot \delta q_{i+1}$$

However, this yields the same discrete Hamiltonian map $\tilde{F}_{L_d}[h]$, which is unsurprising because the discrete flow from $(\boldsymbol{x}_i, R_i, \boldsymbol{\gamma}_i, \boldsymbol{\Pi}_i) \mapsto (\boldsymbol{x}_{i+1}, R_{i+1}, \boldsymbol{\gamma}_{i+1}, \boldsymbol{\Pi}_{i+1})$ on the time interval $[t_i, t_{i+1}]$ is given by $\tilde{F}_{L_d}[h]$. 727 **5.4.** Multiple Impacts. Suppose that multiple impacts occur in the interval (t_{i-1}, t_i) . For 728 concreteness, we assume that there are l impacts. From Subsection 5.2, we determined that 729 the first impact occurs at $\tilde{t} = t_{i-1} + \alpha h$ where $\alpha \in (0, 1)$. Let $\alpha_1 = \alpha$ and introduce

730
$$\alpha_k \in (0, 1 - \alpha_{\Sigma_k}), \qquad \alpha_{\Sigma_k} = \sum_{j=1}^k \alpha_j,$$

where k = 1, 2, ..., l. Then, denote the configurations of impact by $\tilde{q}_k = (\tilde{x}_k, \tilde{R}_k)$ which occurs at the time $\tilde{t}_k = t_{i-1} + \alpha_{\Sigma_k} h$ for each assumed collision; we also write $\tilde{R}_{k+1} = \tilde{R}_k \tilde{F}_k$. In addition, $\Phi(\tilde{x}_k, \tilde{R}_k) = 0$.

5.4.1. Lagrangian Form. For our next integration step, we combine the conservation of discrete energies and

736
$$0 = [D_2 L_d(q_{i-1}, \tilde{q}_1, \alpha_1 h) + D_1 L_d(\tilde{q}_1, \tilde{q}_2, \alpha_2 h)] \cdot \delta \tilde{q}_1,$$

737 so we arrive at

$$F_{L_{d}}^{\text{Impact+}}[\lambda][\alpha_{1}h, \alpha_{2}h] : (\boldsymbol{x}_{i-1}, R_{i-1}, \tilde{\boldsymbol{x}}_{1}, \tilde{R}_{1}) \mapsto (\tilde{\boldsymbol{x}}_{1}, \tilde{R}_{1}, \tilde{\boldsymbol{x}}_{2}, \tilde{R}_{2})$$
(5.17a) $0 = E_{d}(\tilde{q}_{1}, \tilde{q}_{2}, \alpha_{2}h) - E_{d}(q_{i-1}, \tilde{q}_{1}, \alpha_{1}h),$
(5.17b) $0 = \frac{1}{\alpha_{1}h}m(\tilde{\boldsymbol{x}}_{1} - \boldsymbol{x}_{i-1}) - \frac{1}{\alpha_{2}h}m(\tilde{\boldsymbol{x}}_{2} - \tilde{\boldsymbol{x}}_{1}) - \frac{1}{2}mg(\alpha_{1} + \alpha_{2})h\boldsymbol{e}_{3} + \lambda\frac{\partial\tilde{\Phi}_{1}}{\partial\boldsymbol{x}},$
(5.17c) $0 = \text{Asym}\left(\frac{1}{\alpha_{1}h}J_{d}F_{i-1} - \frac{1}{\alpha_{2}h}\tilde{F}_{1}J_{d}\right) + \lambda \text{Asym}\left(\tilde{R}_{1}^{T}\frac{\partial\tilde{\Phi}_{1}}{\partial R}\right),$
(5.17d) $\tilde{R}_{2} = \tilde{R}_{1}\tilde{F}_{1},$
(5.17e) $0 = \boldsymbol{n}^{T}\boldsymbol{x}_{2} - \|I_{\epsilon}\tilde{R}_{2}^{T}\boldsymbol{n}\|.$

738

Observe that $\lambda \neq 0$ and

739 (5.18)
$$\left(\frac{\partial \tilde{\Phi}_k}{\partial \boldsymbol{x}}, \frac{\partial \tilde{\Phi}_k}{\partial \boldsymbol{R}}\right) = \left(\frac{\partial \Phi}{\partial \boldsymbol{x}}, \frac{\partial \Phi}{\partial \boldsymbol{R}}\right)\Big|_{(\tilde{\boldsymbol{x}}_k, \tilde{R}_k)} = \left(\boldsymbol{n}, -\frac{\boldsymbol{n}\boldsymbol{n}^T \tilde{R}_k I_{\epsilon}^2}{\|I_{\epsilon} \tilde{R}_k^T \boldsymbol{n}\|}\right),$$

for k = 1, 2, ..., l - 1. We solve for the next discrete configuration $(\tilde{x}_2, \tilde{R}_2)$ by solving for λ first in the same way as $F_{L_d}^i[\alpha h, (1 - \alpha)h]$. Next, $\alpha_2 \in (0, 1 - \alpha_{\Sigma_1})$ is determined using the bisection method and using $\Phi(\tilde{x}_2, \tilde{R}_2)$ for the stopping criteria. Given both λ and α_2 , proceed to solve $(\tilde{x}_2, \tilde{R}_2)$ using (5.17b)-(5.17d).

In general, we use the same map $F_{L_d}^{\text{Impact}+}[\alpha_k h, \alpha_{k+1}h][\lambda]$ to find the subsequent impact configurations for k = 2, ..., l-1. Once we have determined the last collision, we can find the configuration (\boldsymbol{x}_i, R_i) using $F_{L_d}^i[\alpha_l h, (1 - \alpha_{\Sigma_l})h][\lambda]$ from Subsection 5.3. Finally, we use $F_{L_d}^{i+1}[(1 - \alpha_{\Sigma_l})h, h]$ to determine the configuration $(\boldsymbol{x}_{i+1}, R_{i+1})$.

748 **5.4.2.** Hamiltonian Form. Denote the linear and angular momentum for the configu-749 rations at impact by $(\tilde{\gamma}_k, \tilde{\Pi}_k)$, respectively, where $1 \leq k \leq l$. We compute the Legendre 750 transforms,

$$(\tilde{\boldsymbol{\gamma}}_1, S(\tilde{\boldsymbol{\Pi}}_1)) \cdot_S (\delta \tilde{\boldsymbol{x}}_1, \tilde{\eta}_1) = -D_1 L_d(\tilde{q}_1, \tilde{q}_2, \alpha_2 h) \cdot \delta \tilde{q}_1,$$

752
$$(\tilde{\boldsymbol{\gamma}}_2, S(\tilde{\boldsymbol{\Pi}}_2)) \cdot_S (\delta \tilde{\boldsymbol{x}}_2, \tilde{\eta}_2) = D_2 L_d(\tilde{q}_1, \tilde{q}_2, \alpha_2 h) \cdot \delta \tilde{q}_2.$$

753 Therefore, we have

$$\begin{split} \tilde{F}_{L_d}^{\text{Impact}+}[\lambda][\alpha_2 h] : (\tilde{\boldsymbol{x}}_1, \tilde{R}_1, \tilde{\gamma}_1, \tilde{\boldsymbol{\Pi}}_1) \mapsto (\tilde{\boldsymbol{x}}_2, \tilde{R}_2, \tilde{\gamma}_2, \tilde{\boldsymbol{\Pi}}_2) \\ (5.19a) & 0 = E_d(\tilde{q}_1, \tilde{q}_2, \alpha_2 h) - E_d(q_{i-1}, \tilde{q}_1, \alpha_1 h), \\ (5.19b) & \tilde{\boldsymbol{x}}_2 = \tilde{\boldsymbol{x}}_1 + \frac{\alpha_2 h}{m} \tilde{\gamma}_1 - \frac{1}{2} g \alpha_2^2 h^2 \boldsymbol{e_3} + \lambda \frac{\alpha_2 h}{m} \frac{\partial \tilde{\Phi}_1}{\partial \boldsymbol{x}}, \\ (5.19c) & \tilde{\gamma}_2 = \tilde{\gamma}_1 - mg \alpha_2 h \boldsymbol{e_3} + \lambda \frac{\partial \tilde{\Phi}_1}{\partial \boldsymbol{x}}, \\ (5.19d) & S(\tilde{\boldsymbol{\Pi}}_1) = \text{Asym} \left(\frac{1}{\alpha_2 h} \tilde{F}_1 J_d\right) - \lambda \text{Asym} \left(\tilde{R}_1^T \frac{\partial \tilde{\Phi}_1}{\partial R}\right), \\ (5.19e) & \tilde{F}_1 \tilde{\boldsymbol{\Pi}}_2 = \tilde{\boldsymbol{\Pi}}_1 + \lambda \tilde{\boldsymbol{\chi}}_1, \\ (5.19f) & \tilde{R}_2 = \tilde{R}_1 \tilde{F}_1. \\ (5.19g) & 0 = \boldsymbol{n}^T \boldsymbol{x}_2 - \|I_{\epsilon} \tilde{R}_2^T \boldsymbol{n}\|. \end{split}$$

754 Note that

755

$$S(\tilde{\boldsymbol{\chi}}_1) = \operatorname{Asym}\left(\tilde{R}_1^T \frac{\partial \tilde{\Phi}_1}{\partial R}\right).$$

Again, λ is solved for first by using \tilde{F}_{Jump} from Corollary 4.4, and $\alpha_2 \in (0, 1 - \alpha_{\Sigma_1})$ is subsequently determined using the bisection method. As a result, $(\tilde{\boldsymbol{x}}_2, \tilde{R}_2, \tilde{\boldsymbol{\gamma}}_2, \tilde{\boldsymbol{\Pi}}_2)$ can be calculated using (5.19b)-(5.19f).

Now, we use $\tilde{F}_{L_d}^{\text{Impact}+}[\lambda][\alpha_k h]$ to determine the next set of impact states at time \tilde{t}_k for $k = 3, \dots, l$. After determining the previous collision states, we can find the next states $(\boldsymbol{x}_i, R_i, \boldsymbol{\gamma}_i, \boldsymbol{\Pi}_i)$ using $\tilde{F}_{L_d}^i[(1 - \alpha_{\Sigma_l})h][\lambda]$. Lastly, the discrete Hamiltonian map $\tilde{F}_{L_d}[h]$ is used to determined $(\boldsymbol{x}_{i+1}, R_{i+1}, \boldsymbol{\gamma}_{i+1}, \boldsymbol{\Pi}_{i+1})$ away from the last point of impact.

5.5. The Algorithm and the Zeno Phenomenon: A Summary. The collection of inte-763 grators discuss previously shall be called *Lie group variational collision integrators* (LGVCI). 764which have been written in both Lagrangian and Hamiltonian forms. In this section, we pro-765vide two algorithms that summarize the LGVCI in the Hamiltonian form: The first algorithm 766recalls all of the necessary integrators in (5.9), (5.11), (5.16), and (5.19). The second algorithm 767 is streamlined by utilizing only the discrete Hamiltonian map F_{L_d} , the bisection method for 768 α_j , and the jump map F_{Jump} . We only provide the summary on the Hamiltonian side because 769 it is common to describe hybrid systems in this way. However, it is also straightforward to 770 write the algorithms on the Lagrangian side by imitating the ones presented here. 771

772 **5.5.1. Algorithm 1.** Let us consider Algorithm 5.1: For the inputs, we have the initial 773 states, $(\boldsymbol{x}_{\text{init}}, R_{\text{init}}, \boldsymbol{\gamma}_{\text{init}}, \Pi_{\text{init}})$, and the number of discrete timesteps to be taken, M. The 774 algorithm returns $(t_i^j, \boldsymbol{x}_i^j, R_i^j, \boldsymbol{\gamma}_i^j, \Pi_i^j)$ for $0 \le i \le M$. This is the set of all the discrete states

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($x_i^j, R_i^j, \gamma_i^j, \Pi_i^j$) occurring at time t_i^j . In particular, $t_i^0 = ih$, where h is the fixed timestep. In the case where impacts occur within the interval $[t_i^0, t_{i+1}^0]$, t_i^j represents the times of the collisions, where $1 \le j \le c_i$ and c_i is number of collisions in the interval. The following parameters are required for the algorithm:

- m is the mass of the rigid-body.
- $980 extbf{e} g$ is the gravitational acceleration.
- J is the standard inertia matrix.

785

- I_{ϵ} is the diagonal matrix defined in (2.13) for the ellipsoid.
- \boldsymbol{n} is the normal vector of the plane \mathcal{P} .
- ϵ_{tol} is the tolerance used in Newton's (Appendix D) and bisection method.

The algorithm has the following brief structure: Variables are initialized, and the initial 786distance between the ellipsoid and plane is stored in dist, which is our indicator for collision 787 and/or interpenetration. Next, a for-loop for $1 \leq i \leq M$ has statements falling into one 788 of the three conditions: dist > 0, dist < 0, and dist = 0. The first condition uses $F_{L_d}[h]$ 789 to determine states away from the point of impact. The second condition discards the states 790 with interpenetration and determine the states of the first impact using $\tilde{F}_{L_d}^{\text{Impact}}[\alpha_j h]$. The last 791 condition takes care of the cases where there could be multiple impacts using $\tilde{F}_{L_d}^{\text{Impact}+}[\lambda][\alpha_j h]$ 792 or single impact using $\tilde{F}_{L_d}^i[(1 - \alpha_{\text{tot}})h][\lambda]$. 793

Lastly, note that it is possible to encounter the Zeno phenomenon in our hybrid systems 794 795 due to the form of the ellipsoid and other factors; these include the sectional curvature at the point of impact and the relative distribution of the rotational and translational energies. 796 As a result, a footnote to indicate where to include exception handling for Zeno phenomenon 797 is made in the algorithm. The Zeno phenomenon would manifest itself in the algorithm by 798 arbitrarily large values of j in the while-loop, and α_i that are all approximately machine 799 800 precision. To remedy this, we recommend either breaking out of all the loops or returning the 801 outputs when j exceeds a user-defined threshold.

5.5.2. Algorithm 2. We present a more streamlined algorithm, which only uses \tilde{F}_{L_d} , 802 the bisection method, and F_{Jump} . This is based on the observation that the other discrete 803 Hamiltonian maps are equivalent to some combination of these three subroutines. Namely, 804 $\tilde{F}_{L_d}^{\text{Impact}}[\alpha_j h] = \tilde{F}_{L_d}[\alpha_j h]$ after $\alpha_j \in (0, 1 - \alpha_{\text{tot}})$ is determined using bisection method. There 805 is also $\tilde{F}_{L_d}^i[(1-\alpha_{\rm tot})h][\lambda] = \tilde{F}_{L_d}[(1-\alpha_{\rm tot})h]$ after λ is determined and the instantaneous 806 states before impact are updated with the momenta after the impact from \tilde{F}_{Jump} . Finally, 807 $\tilde{F}_{L_d}^{\text{Impact +}}[\lambda][\alpha_j h] = \tilde{F}_{L_d}[\alpha_j h]$ following both the updates from \tilde{F}_{Jump} and bisection method for α_j . Consequently, we have the more streamlined Algorithm 5.2 with the same parameters, 808 809 inputs, and return. 810

6. Extensions: Tilted Planes and Other Rigid-Bodies. We extend our problem of the bouncing ellipsoid by considering tilted planes and/or other rigid-bodies. This is a natural next step because we want to describe the collision dynamics of different hybrid systems. Note that the theories discussed in both the continuous and discrete cases remain the same with the exception of the collision detection function Φ . Furthermore, it is sufficient that the partial derivatives exists for all "probable" configurations of impact, which will be discussed further Algorithm 5.1 Compute $(t_i^j, \boldsymbol{x}_i^j, R_i^j, \boldsymbol{\gamma}_i^j, \boldsymbol{\Pi}_i^j)$ **Require:** $\overline{h, m, g, J, I_{\epsilon}, \boldsymbol{n}, \epsilon_{\text{tol}}}$ Input: $(\boldsymbol{x}_{\text{init}}, R_{\text{init}}, \boldsymbol{\gamma}_{\text{init}}, \boldsymbol{\Pi}_{\text{init}}), M$ $\alpha_i = 0;$ j = 0; $\alpha_{\rm tot} = 0;$ i = 0; $(t_i^j, \boldsymbol{x}_i^j, R_i^j, \boldsymbol{\gamma}_i^j, \boldsymbol{\Pi}_i^j) \leftarrow (0, \boldsymbol{x}_{\text{init}}, R_{\text{init}}, \boldsymbol{\gamma}_{\text{init}}, \boldsymbol{\Pi}_{\text{init}});$ dist $\leftarrow \Phi(\boldsymbol{x}_i^j, R_i^j);$ for i = 1 : M do if dist > 0 then j = 0; $(\mathbf{x}_{i}^{j}, R_{i}^{j}, \mathbf{\gamma}_{i}^{j}, \mathbf{\Pi}_{i}^{j}) \leftarrow \tilde{F}_{L_{d}}[h](\mathbf{x}_{i-1}^{j}, R_{i-1}^{j}, \mathbf{\gamma}_{i-1}^{j}, \mathbf{\Pi}_{i-1}^{j}); \quad t_{i}^{j} = t_{i-1}^{j} + h;$ dist $\leftarrow \Phi(\boldsymbol{x}_i^j, R_i^j);$ else if dist < 0 then $\begin{array}{ll} i=i-1; & j=j+1; \\ (\boldsymbol{x}_{i}^{j},R_{i}^{j},\boldsymbol{\gamma}_{i}^{j},\boldsymbol{\Pi}_{i}^{j}) \leftarrow \tilde{F}_{L_{d}}^{\mathrm{Impact}}[\alpha_{j}h](\boldsymbol{x}_{i}^{j-1},R_{i}^{j-1},\boldsymbol{\gamma}_{i}^{j-1},\boldsymbol{\Pi}_{i}^{j-1}); & t_{i}^{j}=t_{i}^{j-1}+\alpha_{j}; \end{array}$ $\alpha_{\text{tot}} \leftarrow \alpha_{\text{tot}} + \alpha_j;$ dist $\leftarrow \Phi(\boldsymbol{x}_i^j, R_i^j);$ else i = i - 1;while dist ≤ 0 do if dist = 0 then $(\boldsymbol{x}_{\text{temp}}, R_{\text{temp}}, \boldsymbol{\gamma}_{\text{temp}}, \boldsymbol{\Pi}_{\text{temp}}) \leftarrow \tilde{F}^{i}_{L_{d}}[(1 - \alpha_{\text{tot}})h][\lambda](\boldsymbol{x}^{j}_{i}, R^{j}_{i}, \boldsymbol{\gamma}^{j}_{i}, \boldsymbol{\Pi}^{j}_{i});$ dist $\leftarrow \Phi(\boldsymbol{x}_{\text{temp}}, R_{\text{temp}});$ if dist > 0 then i = i + 1;j = 0; $\alpha_j = 0;$ $\alpha_{\text{tot}} = 0;$ $(\boldsymbol{x}_{i}^{j}, R_{i}^{j}, \boldsymbol{\gamma}_{i}^{j}, \boldsymbol{\Pi}_{i}^{j}) \leftarrow (\boldsymbol{x}_{\text{temp}}, R_{\text{temp}}, \boldsymbol{\gamma}_{\text{temp}}, \boldsymbol{\Pi}_{\text{temp}}); \quad t_{i}^{j} = t_{i-1}^{j} + h;$ end if else $j = j + 1;^{\dagger}$ $(\boldsymbol{x}_{i}^{j}, R_{i}^{j}, \boldsymbol{\gamma}_{i}^{j}, \boldsymbol{\Pi}_{i}^{j}) \leftarrow \tilde{F}_{L_{d}}^{\mathrm{Impact}+}[\lambda][\alpha_{j}h](\boldsymbol{x}_{i}^{j-1}, R_{i}^{j-1}, \boldsymbol{\gamma}_{i}^{j-1}, \boldsymbol{\Pi}_{i}^{j-1}); \qquad t_{i}^{j} = t_{i}^{j-1} + \alpha_{j};$ $\alpha_{\rm tot} = \alpha_{\rm tot} + \alpha_j;$ dist $\leftarrow \Phi(\boldsymbol{x}_{i}^{j}, R_{i}^{j});$ end if end while end if end for return $(t_i^j, \boldsymbol{x}_i^j, R_i^j, \boldsymbol{\gamma}_i^j, \boldsymbol{\Pi}_i^j)$

[†] Zeno phenomenon: j is arbitrarily large, so break/return is recommended.

Algorithm 5.2 Compute $(t_i^j, \boldsymbol{x}_i^j, R_i^j, \boldsymbol{\gamma}_i^j, \boldsymbol{\Pi}_i^j)$ **Require:** $h, m, g, J, I_{\epsilon}, \boldsymbol{n}, \epsilon_{tol}, M$ Input: $(\boldsymbol{x}_{\text{init}}, R_{\text{init}}, \boldsymbol{\gamma}_{\text{init}}, \boldsymbol{\Pi}_{\text{init}})$ i = 0; j = 0; $\alpha_i = 0;$ $\alpha_{\rm tot} = 0;$ $(t_i^j, \boldsymbol{x}_i^j, R_i^j, \boldsymbol{\gamma}_i^j, \boldsymbol{\Pi}_i^j) \leftarrow (0, \boldsymbol{x}_{\text{init}}, R_{\text{init}}, \boldsymbol{\gamma}_{\text{init}}, \boldsymbol{\Pi}_{\text{init}});$ dist $\leftarrow \Phi(\boldsymbol{x}_{i}^{j}, R_{i}^{j});$ for i = 1 : M do if dist ≥ 0 then if dist > 0 then j = 0; $(\mathbf{x}_{i}^{j}, R_{i}^{j}, \mathbf{\gamma}_{i}^{j}, \mathbf{\Pi}_{i}^{j}) \leftarrow \tilde{F}_{L_{d}}[h](\mathbf{x}_{i-1}^{j}, R_{i-1}^{j}, \mathbf{\gamma}_{i-1}^{j}, \mathbf{\Pi}_{i-1}^{j}); \quad t_{i}^{j} = t_{i-1}^{j} + h;$ dist $\leftarrow \Phi(\boldsymbol{x}_i^j, R_i^j);$ else i = i - 1; $(\lambda, \boldsymbol{\gamma}_i^j, \boldsymbol{\Pi}_i^j) \leftarrow \tilde{F}_{\text{Jump}}(\boldsymbol{x}_i^j, R_i^j, \boldsymbol{\gamma}_i^j, \boldsymbol{\Pi}_i^j);$ $(\boldsymbol{x}_{\text{temp}}, R_{\text{temp}}, \boldsymbol{\gamma}_{\text{temp}}, \boldsymbol{\Pi}_{\text{temp}}) \leftarrow \tilde{F}_{L_d}[(1 - \alpha_{\text{tot}})h](\boldsymbol{x}_i^j, R_i^j, \boldsymbol{\gamma}_i^j, \boldsymbol{\Pi}_i^j);$ dist $\leftarrow \Phi(\boldsymbol{x}_{\text{temp}}, R_{\text{temp}});$ if dist > 0 then i = i + 1; j = 0; $\alpha_i = 0;$ $\alpha_{tot} = 0;$ $(\boldsymbol{x}_{i}^{j}, R_{i}^{j}, \boldsymbol{\gamma}_{i}^{j}, \boldsymbol{\Pi}_{i}^{j}) \leftarrow (\boldsymbol{x}_{\text{temp}}, R_{\text{temp}}, \boldsymbol{\gamma}_{\text{temp}}, \boldsymbol{\Pi}_{\text{temp}}); \quad t_{i}^{j} = t_{i-1}^{j} + h;$ end if end if else i = i - 1; $j = j + 1;^{\dagger}$ $\alpha_i \leftarrow \text{Bisection}(0, 1 - \alpha_{\text{tot}});$ $(\boldsymbol{x}_{i}^{j}, R_{i}^{j}, \boldsymbol{\gamma}_{i}^{j}, \boldsymbol{\Pi}_{i}^{j}) \leftarrow \tilde{F}_{L_{d}}[\alpha_{j}h](\boldsymbol{x}_{i}^{j-1}, R_{i}^{j-1}, \boldsymbol{\gamma}_{i}^{j-1}, \boldsymbol{\Pi}_{i}^{j-1}); \qquad t_{i}^{j} = t_{i}^{j-1} + \alpha_{i};$ $\alpha_{\text{tot}} \leftarrow \alpha_{\text{tot}} + \alpha_j;$ dist $\leftarrow \Phi(\boldsymbol{x}_i^j, R_i^j);$ end if end for return $(t_i^j, \boldsymbol{x}_i^j, R_i^j, \boldsymbol{\gamma}_i^j, \boldsymbol{\Pi}_i^j)$

 † Zeno phenomenon: j is arbitrarily large, so break/return is recommended.

in this section. Hence, the different collision detection functions and their partial derivativesare discussed here for different hybrid systems.

In this section, the general planes $\mathcal{P}(\tilde{n}, D)$ defined in (2.12) will be referenced. Furthermore, we define the signed distance function for an arbitrary half-plane

821 (6.1)
$$\mathcal{HP}(\tilde{\boldsymbol{n}}, D) = \{ \boldsymbol{z} \in \mathbb{R} \mid \tilde{\boldsymbol{n}}^T \boldsymbol{z} + D \le 0 \},\$$

which is the region opposite from the direction of the normal vector \tilde{n} of the plane:

Proposition 6.1. The signed distance function $\psi_{\mathcal{P}} : \mathbb{R}^3 \to \mathbb{R}$ for the half plane $\mathcal{HP}(\tilde{n}, D)$ is defined by

825 (6.2)
$$\psi_{\mathcal{P}}(\boldsymbol{z}) = \tilde{\boldsymbol{n}}^T \boldsymbol{z} + D = \begin{cases} -d_2(\boldsymbol{z}, \mathcal{P}(\tilde{\boldsymbol{n}}, D)), & \boldsymbol{z} \in \mathcal{HP}(\tilde{\boldsymbol{n}}, D), \\ d_2(\boldsymbol{z}, \mathcal{P}(\tilde{\boldsymbol{n}}, D)), & \boldsymbol{z} \in \mathcal{HP}(\tilde{\boldsymbol{n}}, D)^c. \end{cases}$$

6.1. Tilted Planes. Recall that our system describing the bouncing ellipsoid involves the horizontal plane denoted as $\mathcal{P} = \{ \boldsymbol{z} \in \mathbb{R}^3 \mid \boldsymbol{n}^T \boldsymbol{z} = 0 \}$ where $\boldsymbol{n} = (0, 0, 1)^T$. For tilted planes, it suffices to consider planes with a normal vector

829
$$\tilde{n} \in S^2_+ = \{ \boldsymbol{z} \in \mathbb{R}^3 \mid \|\boldsymbol{z}\| = 1 \text{ and } z_3 > 0 \},$$

that pass through the origin because shifted planes with the same normal vector \tilde{n} are equivariant with respect to translations in the gravitational direction. Furthermore, we require $\tilde{n}_3 > 0$ so that the orientation is preserved; specifically, this gives the minus sign in our formula below when the minimum is taken (see Theorem A.11 in Appendix A.2). The case $\tilde{n} = (0, 0, 1)^T \in S^2_+$ gives the horizontal plane in our original discussion.

835 Consider the tilted plane

836 (6.3)
$$\mathcal{P}_{\tilde{\boldsymbol{n}}} = \{ \boldsymbol{z} \in \mathbb{R}^3 \mid \tilde{\boldsymbol{n}}^T \boldsymbol{z} = 0 \}$$

Now, suppose $(\boldsymbol{x}, R) \in SE(3)$ so that the arbitrary ellipsoid, $\mathcal{E}' = T_{(\boldsymbol{x},R)}(\mathcal{E})$, is above the tilted plane; the collision detection function is written as

839 (6.4)
$$d_2(\mathcal{E}', \mathcal{P}_{\tilde{\boldsymbol{n}}}) = \Phi(\boldsymbol{x}, R) = \tilde{\boldsymbol{n}}^T \boldsymbol{x} - \|I_{\mathcal{E}} R^T \tilde{\boldsymbol{n}}\|.$$

840 Then, $\mathcal{E}' \cap \mathcal{P}_{\tilde{n}} = \emptyset$ if and only if $\tilde{n}^T x > \|I_{\mathcal{E}} R^T \tilde{n}\|$. The partial derivatives are computed in 841 the same way as before:

842 (6.5)
$$\left(\frac{\partial \Phi}{\partial \boldsymbol{x}}, \frac{\partial \Phi}{\partial R}\right) = \left(\tilde{\boldsymbol{n}}, -\frac{\tilde{\boldsymbol{n}}\tilde{\boldsymbol{n}}^T R I_{\epsilon}^2}{\|I_{\epsilon} R^T \tilde{\boldsymbol{n}}\|}\right).$$

6.2. Convex Rigid-Bodies and Interface Implicit Representations. We propose a way to construct the collision detection function Φ for some convex domains.

Suppose the domain of the rigid-body $\mathcal{B} \subset \mathbb{R}^3$ is convex and compact, and $\rho : \mathcal{B} \to \mathbb{R}$ is the mass density function such that the center of mass of \mathcal{B} coincides with the origin. The function also provides the respective standard and nonstandard inertia matrix J and J_d for our numerical implementation. Now, let the *interface* $\partial \mathcal{B}$ be C^k where $k \geq 2$. This interface

divides \mathbb{R}^3 into two separate regions where the interior \mathcal{B}^o is the *inside* and the complement 849 \mathcal{B}^{c} is the *outside* of the domain. In addition, the assumption on smoothness guarantees that 850 there are no edges – a curve where two surfaces meet – and no vertices, points where two or 851 more edges meet; also, it guarantees that the partial derivatives of Φ will be continuous. 852

853 Now, the interface can be represented in two ways: In an *explicit* representation, one explicitly writes all the points that belong to the interface. On the other hand, *implicit* 854 representation of the interface is given by the zero level set of some implicit function ϕ after 855 a constant shift if necessary [37]. In fact, the signed distance functions (SDFs) $\psi : \mathbb{R}^3 \to \mathbb{R}$ 856 are a subset of such implicit functions defined by 857

858 (6.6)
$$\psi(\boldsymbol{z}) = \begin{cases} d_2(\boldsymbol{z}, \partial \mathcal{B}), & \boldsymbol{z} \in \mathcal{B}^c \\ -d_2(\boldsymbol{z}, \partial \mathcal{B}), & \boldsymbol{z} \in \mathcal{B}, \end{cases}$$

but it also gives $\|\nabla \psi\| = 1$ everywhere except on the zero level set. We reserve ψ for SDFs and 859 ϕ as the general class of functions whose zero level set is $\partial \mathcal{B}$. In particular, these functions are 860 positive for z outside of the interface, negative for z inside of the interface, and zero otherwise. 861

Of course, $|\psi(z)|$ is the distance from z to the interface. Lastly, note that 862

863
$$\nabla \psi(\boldsymbol{z}) = N(\boldsymbol{z}),$$

where N is the outward normal vector field. Essentially, the collision detection function and 864 its partial derivatives between the body and plane will be constructed using the SDF or some 865 866 other implicit representation of the interface.

6.2.1. Theory: The Collision Detection. Suppose the hybrid system involves a convex 867 rigid-body \mathcal{B} with C^k boundary $(k \geq 2)$, and let the SDF ψ or some implicit representation ϕ 868 of the rigid-body be given. The collision detection problem is described first in the body-fixed 869 frame: Suppose $\mathcal{P}_{\tilde{n}} = \mathcal{P}(\tilde{n}, 0)$ is a plane containing the origin and $\mathcal{B}' = T_{(\boldsymbol{x},R)}(\mathcal{B})$ be the 870 translated, rotated body above the plane for some $(\boldsymbol{x}, R) \in SE(3)$. Suppose $\mathcal{B}' \cap \mathcal{P}_{\tilde{\boldsymbol{n}}} = \emptyset$, and 871 so the distance between them can be written equivalently as 872

$$d_2(\mathcal{B}', \mathcal{P}_{\tilde{\boldsymbol{n}}}) = d_2(\mathcal{B}, \mathcal{P}'_{\tilde{\boldsymbol{n}}}),$$

where 874

875 (6.7)
$$\mathcal{P}'_{\tilde{\boldsymbol{n}}} = \mathcal{P}(R^T \tilde{\boldsymbol{n}}, \tilde{\boldsymbol{n}}^T \boldsymbol{x}) = \{ \boldsymbol{z} \in \mathbb{R}^3 \mid \tilde{\boldsymbol{n}}^T R \boldsymbol{z} + \tilde{\boldsymbol{n}}^T \boldsymbol{x} = 0 \},$$

since $T_{(\boldsymbol{x},R)}$ is an isometry (see Appendix A.2). This equivalent view of the distance is 876 constructed from the body-fixed frame, so one can imagine a shifted and rotated plane 877 $\mathcal{P}'_{\tilde{\boldsymbol{n}}} = T_{(\boldsymbol{x},R)}^{-1}(\mathcal{P}_{\tilde{\boldsymbol{n}}})$ in this frame. Furthermore, the construction is made in this frame because 878 the body elements in \mathcal{B} , especially its boundary, are fixed; this allows us to easily construct 879 Φ using a constrained optimization problem, where the constraint is $\rho \in \partial \mathcal{B}$. 880

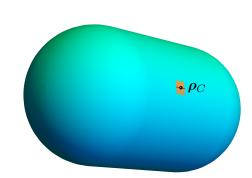
Let us denote $\rho_C \in \partial \mathcal{B}$ as a closest point to the plane $\mathcal{P}'_{\tilde{n}}$ in the body-fixed frame, meaning 881

882
$$d_2(\boldsymbol{\rho}_C, \mathcal{P}'_{\tilde{\boldsymbol{n}}}) = d_2(\mathcal{B}, \mathcal{P}'_{\tilde{\boldsymbol{n}}}).$$

For most configurations $(\boldsymbol{x}, R) \in SE(3)$, there is a unique $\boldsymbol{\rho}_C$ because \mathcal{B} is convex and has C^k 883 (k > 2) boundary. The exception occurs for configurations that have zero Gaussian curvature 884



(a) The neighborhood of $\rho_C \in \partial \mathcal{B}$ is locally flat where all the points are also closest to the plane.



(b) The neighborhood of $\rho_C \in \partial \mathcal{B}$ with a straight red curve where all the points are also closest to the plane.

Figure 3: Examples of neighborhood with non-unique closest points on convex bodies with C^{∞} boundary.

at the point $\rho_C \in \partial \mathcal{B}$, which can be categorized into two cases (see Figure 3): In the first 885 case, both principal curvatures at ρ_C are zeros, and the neighborhood of ρ_C is a plane which 886 is parallel to the $\mathcal{P}'_{\tilde{n}}$. This implies that all the points in this locally planar neighborhood 887 is also closest to the plane (e.g., see Figure 3a). In the second case, one of the principal 888 curvatures at ρ_C is zero; in particular, a neighborhood of ρ_C in $\partial \mathcal{B}$ will have a curve with 889 points which are all closest to the plane, and this curve is parallel to the plane as well, e.g., 890 see Figure 3b. However, these cases are *improbable* in numerical computations because the 891 892 discrete configuration (x, R) must be exact for the two cases to arise. Therefore, it is also improbable for these cases to arise at the point of impact, so we only consider the *probable* 893 configurations where ρ_C is unique. This notion of probable configurations is analogous to the 894 notion of general position or genericity that arises in computational geometry and algebraic 895 geometry. 896

The improbable configurations are highlighted here because we do not want to dismiss them entirely, since they are still possible configurations of hybrid systems. However, the LGVCI algorithm would not capture these configurations, which is actually a good representation of the hybrid systems in real life. In particular, the set of improbable configurations is a set of measure zero in all the configurations away from impacts. Furthermore, the improbable configurations will not occur during impact in our implementation as well, so $\mathcal{B} \cap \mathcal{P}'_{\tilde{n}} = \{\rho_C\}$ at the configurations of impact.

Given this unique point ρ_C , the collision detection and its partial derivatives can now be discussed. Let (\boldsymbol{x}, R) be a probable configuration so that $\mathcal{P}'_{\tilde{\boldsymbol{n}}}$ is given in the body-fixed frame. In theory, ρ_C is dependent on (\boldsymbol{x}, R) and can be determined since $\partial \mathcal{B}$ is compact. Namely, $\rho_C(\boldsymbol{x}, R)$ is determined by the arg min of all possible distances from the boundary points to

34

908 the plane:

909 (6.8)
$$\boldsymbol{\rho}_{C}(\boldsymbol{x}, R) = \underset{\boldsymbol{\rho} \in \partial \mathcal{B}}{\operatorname{arg\,min}} \psi_{\mathcal{P}_{\tilde{\boldsymbol{n}}}'}(\boldsymbol{\rho})$$
$$= \underset{\{\boldsymbol{\rho} \in \mathbb{R}^{3} | \phi(\boldsymbol{\rho}) = 0\}}{\operatorname{arg\,min}} \tilde{\boldsymbol{n}}^{T} R \boldsymbol{\rho} + \tilde{\boldsymbol{n}}^{T} \boldsymbol{x},$$

using the signed distance function for the plane defined in Proposition 6.1. Also, recall that $\phi^{-1}(0) = \partial \mathcal{B}$ for an implicit representation ϕ , and this remains true if the representation is an SDF. As a result, the collision detection function is defined simply as

913 (6.9)
$$\Phi(\boldsymbol{x}, R) = \min_{\{\boldsymbol{\rho} \in \mathbb{R}^3 | \phi(\boldsymbol{\rho}) = 0\}} \tilde{\boldsymbol{n}}^T R \boldsymbol{\rho} + \tilde{\boldsymbol{n}}^T \boldsymbol{x}$$
$$= \tilde{\boldsymbol{n}}^T R \boldsymbol{\rho}_C(\boldsymbol{x}, R) + \tilde{\boldsymbol{n}}^T \boldsymbol{x}.$$

J

Indeed, this is desirable because this function also encodes interpenetration. Namely, if the plane intersects the body with interpenetration for a given configuration $(\boldsymbol{x}, R) \in SE(3)$, then $\tilde{\boldsymbol{n}}^T R \boldsymbol{\rho} + \tilde{\boldsymbol{n}}^T \boldsymbol{x} < 0$ for some $\boldsymbol{\rho} \in \partial \mathcal{B}$, and so $\Phi(\boldsymbol{x}, R) < 0$. In addition, $\Phi(\boldsymbol{x}, R) < 0$ for configurations where the body completely passes through the plane because all the boundary points would be in the interior of the half-plane $\mathcal{HP}'_{\tilde{\boldsymbol{n}}}$ defined in (6.1).

Finally, one may solve for $\rho_C(x, R) \in \partial \mathcal{B}$ using the constrained optimization problem in (6.9). Recall that possible solutions to the problem are the critical points for the Lagrangian function

$$\mathcal{L}(\boldsymbol{x}, R; \boldsymbol{
ho}) = \tilde{\boldsymbol{n}}^T R \boldsymbol{
ho} + \tilde{\boldsymbol{n}}^T \boldsymbol{x} + \lambda \phi(\boldsymbol{
ho}),$$

923 where λ is the Lagrange multiplier. Observe that $\frac{\partial \mathcal{L}}{\partial \rho} = \mathbf{0}$ implies that $\lambda \nabla \phi(\rho) = -R^T \tilde{n}$. 924 Hence we have the following remark:

925 Remark 6.2. Given a probable configuration $(\boldsymbol{x}, R) \in SE(3), \ \boldsymbol{\rho}_C \in \partial \mathcal{B}$ is unique and 926 $\boldsymbol{z} = \boldsymbol{\rho}_C$ is a solution to $\nabla \psi(\boldsymbol{z}) = \pm R^T \tilde{\boldsymbol{n}}$ or $\lambda \nabla \phi(\boldsymbol{z}) = -R^T \tilde{\boldsymbol{n}}$ for some $\lambda \in \mathbb{R}$.

Intuitively, this means that the tangent plane at $\rho_C \in \partial \mathcal{B}$ is parallel to $\mathcal{P}'_{\tilde{n}}$ since they share the same normal vector up to a scalar. Furthermore, as a consequence of Remark 6.2, one might conclude that the dependencies should be changed, $\rho_C(\boldsymbol{x}, R) \longrightarrow \rho_C(R)$.

930 **6.2.2. Theory: The Partial Derivatives of** Φ . We continue our discussion with the partial 931 derivatives of the collision detection function. By the smoothness of the interface, its partial 932 derivatives exist and are continuous. Specifically, they should exist for probable configurations 933 at impact since the partial derivatives are computed during collision. Given the setup for the 934 constrained optimization problem, we can simply compute the partial derivatives in general 935 as

936 (6.10)
$$\left. \left(\frac{\partial \Phi}{\partial \boldsymbol{x}}, \frac{\partial \Phi}{\partial R} \right) \right|_{(\boldsymbol{x},R)} = \left(\tilde{\boldsymbol{n}}, \tilde{\boldsymbol{n}} \boldsymbol{\rho}_C^T + \tilde{\boldsymbol{n}}^T R \frac{\partial \boldsymbol{\rho}_C}{\partial R} \right).$$

937 Intuitively, $\frac{\partial \Phi}{\partial x}$ makes sense because given a fixed R, the gradient of Φ with respect to x, 938 the position of the center of mass of the body, should point towards the direction of greatest 939 increase for Φ ; this is, indeed, the normal vector of the plane \tilde{n} . Hence, given any admissible 940 configurations and configurations at impact, $\frac{\partial \Phi}{\partial x} = \tilde{n}$.

On the other hand, $\frac{\partial \Phi}{\partial R}$ depends on the uniqueness of ρ_C which is satisfied by our discussion 941 of probable configurations. In fact, suppose an improbable configuration $(x, R) \in SE(3)$ 942 is given so that ρ_C is not unique, and we decide to choose one of the closest points as a 943 representation. Then, the partial derivative $\frac{\partial \Phi}{\partial B}$ will not be well-defined, and it will be different 944 945 for each representation. This is unsurprising, since the directional derivative along any fixed rotation will be drastically different for the different elements of the closest point set. Lastly, 946 we may also check this against the example of the ellipsoid. In Appendix A.1, we showed 947 that the closest, unique point on the boundary of the ellipsoid to an arbitrary plane is (A.5). 948 Hence, given the fixed setup where $\mathcal{P}'_{\tilde{n}}$ has normal vector $R^T \tilde{n}$ pointing towards the body, the 949 closest point is 950

$$\boldsymbol{
ho}_C(R) = -rac{I_\epsilon^2 R^T \, ilde{oldsymbol{n}}}{\|I_\epsilon R^T ilde{oldsymbol{n}}\|}.$$

952 One can show that $\tilde{\boldsymbol{n}}^T R \frac{\partial \boldsymbol{\rho}_C}{\partial R} = 0$ using (M.f) of Proposition C.1 in Appendix C and then get 953 $\frac{\partial \Phi}{\partial R}(\boldsymbol{x}, R) = \tilde{\boldsymbol{n}} \boldsymbol{\rho}_C(\boldsymbol{x}, R)^T = -\frac{\tilde{\boldsymbol{n}} \tilde{\boldsymbol{n}}^T R I_{\epsilon}^2}{\|I_{\epsilon} R^T \tilde{\boldsymbol{n}}\|}$ for all admissible configurations and configurations at 954 impact. Indeed, this was already derived in (6.5).

6.3. Convex Polyhedra. In this section, we extend our theory for hybrid systems by considering rigid-bodies that are convex polyhedra. In particular, a convex polyhedron \mathcal{B} is defined by the convex hull of a collection of vertices $\{v_j\}_{j=1}^l$ where $l \geq 4$. Its centroid and moment of inertia may be computed using the Mirtich's algorithm in [33]. Otherwise, one may utilize built-in functions RegionCentroid and MomentOfInertia in Mathematica 12 to compute the centroid and inertia matrix, respectively. Of course, the centroid of the body can be made to coincide with the origin by shifting the vertices.

Note that \mathcal{B} has a C^0 interface, which consists of faces, edges, and vertices. In subsection 962 6.2, we discussed the improbable configurations where the closest point to the plane is not 963 unique, and this case would arise when any one of the faces or edges of the polyhedron is 964 closest and parallel to the plane. However again, these configurations remain improbable due 965 966 to finite numerical precision and numerical roundoff. In fact, suppose one of the edges is closest and parallel to the plane; a small perturbation in the configuration of the body will 967 leave one of the two vertices of the edge as the closest point to the plane. Similarly, if one of 968 the faces is closest and parallel to the plane, a small perturbation will leave one of the vertices 969 of the face as the closest point. As a result, the closest point is almost always a vertex, and 970 971 it is unique; this also ensures that a vertex is always the singleton at the point of impact.

Despite this simplification, we face a challenge applying our jump conditions at the points 972 of impact, which are sharp corners of the polyhedron. One possible approach is to replace our 973 continuous equations of motion with *differential inclusions*. This will yield a set of possible 974momenta that lie in the normal cone (in the sense of convex analysis) at the configuration of 975 impact (x_i, R_i) . To compute a realization, we would have to choose a momenta within the 976 normal cone to determine the state after the collision, which adds an element of randomness 977 to the simulation which is undesirable. We avoid these issues altogether by considering a 978 979 regularization of the rigid-body. Using this method, the modified convex polyhedron has a C^{∞} boundary. Hence, its impact point on the plane will be a singleton, and the partial 980 derivatives will exist. 981

951



Figure 4: Examples of exaggerated ϵ -rounding in blue and green of the cuboid interface in orange.

Let us begin by considering the SDF ψ for the convex polyhedron \mathcal{B} , then the interface is given by the zero level set, i.e., $\psi^{-1}(0) = \partial \mathcal{B}$. We modify the SDF, so that the zero level set has smoothness near the vertices, and this is done using ϵ -rounding where ϵ is a sufficiently small parameter: Define the new SDF $\psi_{\epsilon} : \mathbb{R}^3 \to \mathbb{R}$ by

986 (6.11)
$$\psi_{\epsilon}(\boldsymbol{x}) = \psi(\boldsymbol{x}) - \epsilon.$$

Then, the zero level set of ψ_{ϵ} is similar to \mathcal{B} , but its surfaces and edges are ϵ -distance further away from each respective surfaces and edges. Furthermore, the corners are now rounded with a radius of curvature bounded from below by ϵ , for example in Figure 4. Hence, define $\partial \mathcal{B}^{\epsilon} = \psi_{\epsilon}^{-1}(0)$ as the new interface of interest, and let \mathcal{B}^{ϵ} be the new rigid-body of interest defined by the convex hull of $\partial \mathcal{B}^{\epsilon}$.

For a probable configuration $(\boldsymbol{x}, R) \in SE(3)$, we are given the plane $\mathcal{P}'_{\tilde{\boldsymbol{n}}}$ in the body-fixed frame. Now, the unique, closest point on the boundary $\partial \mathcal{B}^{\epsilon}$ will always be on the rounded portion of a particular corner. This can be shown by first determining the closest point $\rho_C \in \partial \mathcal{B}$, which is one of the vertices:

996
$$\boldsymbol{\rho}_C(\boldsymbol{x}, R) = \operatorname*{arg\,min}_{\boldsymbol{\rho} \in \{\boldsymbol{v}_j\}_{j=1}^l} \tilde{\boldsymbol{n}}^T R \boldsymbol{\rho} + \tilde{\boldsymbol{n}}^T \boldsymbol{x}.$$

⁹⁹⁷ Then the closest point on $\partial \mathcal{B}^{\epsilon}$ is ϵ -distance along the normal vector towards the plane. In ⁹⁹⁸ other words,

999 (6.12)
$$\boldsymbol{\rho}_C^{\epsilon}(\boldsymbol{x},R) = \boldsymbol{\rho}_C(\boldsymbol{x},R) - \epsilon R^T \tilde{\boldsymbol{n}}.$$

1000 In particular, this point $\rho_C^{\epsilon} \in \partial \mathcal{B}^{\epsilon}$, represented as a vector, always forms obtuse angles with 1001 all the surfaces and edges at the vertex ρ_C (see Figure 5). This means that ρ_C^{ϵ} is unique and 1002 always on the rounded corner.

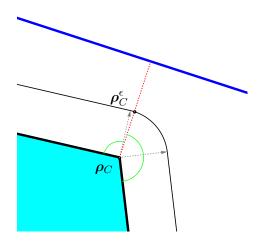


Figure 5: In 2D, the point $\rho_C^{\epsilon} \in \partial \mathcal{B}^{\epsilon}$ is always on the rounded corner forming obtuse angles in green.

Finally, following the definition in (6.9), define the collision detection function for the rounded convex polyhedron \mathcal{B}^{ϵ} as

1005 (6.13)
$$\Phi(\boldsymbol{x},R) = \tilde{\boldsymbol{n}}^T R \boldsymbol{\rho}_C(\boldsymbol{x},R) + \tilde{\boldsymbol{n}}^T \boldsymbol{x} - \epsilon.$$

1006 Also, recall that $\rho_C(x, R)$ is the vertex of the convex polyhedron \mathcal{B} that is closest to the plane. 1007 Then the partial derivatives are computed to be

1008 (6.14)
$$\left. \left(\frac{\partial \Phi}{\partial \boldsymbol{x}}, \frac{\partial \Phi}{\partial R} \right) \right|_{(\boldsymbol{x},R)} = \left(\tilde{\boldsymbol{n}}, \tilde{\boldsymbol{n}} \boldsymbol{\rho}_C^T + \tilde{\boldsymbol{n}}^T R \frac{\partial \boldsymbol{\rho}_C}{\partial R} - \epsilon \tilde{\boldsymbol{n}} \tilde{\boldsymbol{n}}^T R \right)$$

1009 **6.4.** Union and Intersection of Convex Rigid-Bodies. Recall that the signed distance 1010 function ψ is a subset of the functions that can represent the interface implicitly. In this representation, we have many accessible geometric tools including boolean operations for 1011 advanced constructive solid geometry (CSG). These boolean operations include the union, 1012 intersection, and complement, and the resulting rigid-bodies will be composed of the convex 1013 1014 rigid-bodies in the previous discussion. However, they will no longer be convex in general after the boolean operations. Therefore, $\rho_C \in \partial \mathcal{B}$, the closest point to the plane, is not necessarily 1015unique for certain configurations including configurations of impact. Nevertheless, we include 1016 this discussion because the LGVCI will still generally capture the dynamics where the collision 1017set $\mathcal{B} \cap \mathcal{P}'_{\tilde{n}}$ is a singleton and the CSG is minimal. 1018

In this section, we only consider two rigid-bodies for the union and intersection and discuss the collision detection and its partial derivatives for the resulting, constructed body. Notably, the new rigid-body must have its center of mass coinciding with the origin. Additional union/intersection of bodies can be considered but it should be minimal. This should be done with care due to the center of mass and its geometry which could significantly increase the chance of non-singleton intersection at the point of impact. 1025 Suppose ψ_1 and ψ_2 are two SDFs whose corresponding convex bodies, \mathcal{B}_1 and \mathcal{B}_2 , do not 1026 necessarily have the centers of mass at the origin. Let $\mathcal{B}_1 \cap \mathcal{B}_2 \neq \emptyset$, and suppose $\mathcal{B}_{\cup} = \mathcal{B}_1 \cup \mathcal{B}_2$ 1027 is their union with a center of mass coinciding with the origin. Then the corresponding SDF 1028 $\psi_{\cup} : \mathbb{R}^3 \to \mathbb{R}$ is defined by

1029 (6.15)
$$\psi_{\cup}(\boldsymbol{x}) = \min\{\psi_1(\boldsymbol{x}), \psi_2(\boldsymbol{x})\}.$$

1030 Furthermore, its gradient is given by

1031 (6.16)
$$\nabla \psi_{\cup}(\boldsymbol{x}) = \begin{cases} \nabla \psi_1(\boldsymbol{x}) & \text{if } \psi_1(\boldsymbol{x}) < \psi_2(\boldsymbol{x}), \\ \nabla \psi_2(\boldsymbol{x}) & \text{if } \psi_2(\boldsymbol{x}) < \psi_1(\boldsymbol{x}), \\ \nabla \psi_1(\boldsymbol{x}) & \text{if } \psi_1(\boldsymbol{x}) = \psi_2(\boldsymbol{x}) \text{ and } \nabla \psi_1(\boldsymbol{x}) = \nabla \psi_2(\boldsymbol{x}), \\ \text{Undefined} & \text{if } \psi_1(\boldsymbol{x}) = \psi_2(\boldsymbol{x}) \text{ and } \nabla \psi_1(\boldsymbol{x}) \neq \nabla \psi_2(\boldsymbol{x}). \end{cases}$$

Recall that the gradient is needed to resolve the impact, so we would like to avoid the last case, which does not arise in most CSG. The case arise when $\boldsymbol{x} \in \partial \mathcal{B}_1 \cap \partial \mathcal{B}_2$ is the singleton for the intersection at impact and $\nabla \psi_1(\boldsymbol{x}) \neq \nabla \psi_2(\boldsymbol{x})$. However, this could arise in unusual examples of union such as a small ellipsoid completely sitting inside a larger ellipsoid where their boundaries share a point.

For the intersection of the bodies, $\mathcal{B}_{\cap} = \mathcal{B}_1 \cap \mathcal{B}_2$, we also assume that its center of mass is at the origin. The corresponding SDF $\psi_{\cap} : \mathbb{R}^3 \to \mathbb{R}$ is defined as

1039 (6.17)
$$\psi_{\cap}(\boldsymbol{x}) = \max\{\psi_1(\boldsymbol{x}), \psi_2(\boldsymbol{x})\}.$$

1040 The gradient is given by

1041 (6.18)
$$\nabla \psi_{\cap}(\boldsymbol{x}) = \begin{cases} \nabla \psi_1(\boldsymbol{x}) & \text{if } \psi_1(\boldsymbol{x}) > \psi_2(\boldsymbol{x}), \\ \nabla \psi_2(\boldsymbol{x}) & \text{if } \psi_2(\boldsymbol{x}) > \psi_1(\boldsymbol{x}), \\ \nabla \psi_1(\boldsymbol{x}) & \text{if } \psi_1(\boldsymbol{x}) = \psi_2(\boldsymbol{x}) \text{ and } \nabla \psi_1(\boldsymbol{x}) = \nabla \psi_2(\boldsymbol{x}), \\ \text{Undefined } \text{if } \psi_1(\boldsymbol{x}) = \psi_2(\boldsymbol{x}) \text{ and } \nabla \psi_1(\boldsymbol{x}) \neq \nabla \psi_2(\boldsymbol{x}). \end{cases}$$

1042 For this gradient, the last case is as similarly unlikely as the improbable configurations dis-1043 cussed for the general convex rigid-body. Put another way, the gradient is generically well-1044 defined.

Lastly, the complement is also introduced since it is useful for CSG. Given a convex rigidbody \mathcal{B} and the SDF ψ , the SDF for the complement \mathcal{B}^{C} is given by

1047 (6.19)
$$\psi_{\rm C}(\boldsymbol{x}) = -\psi(\boldsymbol{x}).$$

7. Numerical Experiments. Numerical experiments are performed using Algorithm 5.2 for the following four hybrid systems consisting of different rigid-bodies and planes:

1050 Case I: Triaxial ellipsoid over the horizontal plane.

1051 Case II: Triaxial ellipsoid over tilted plane.

1052Case III: Union of ellipsoids over the horizontal plane.

1053Case IV: A cube over the horizontal plane.

	Rigid-Body Properties	Center(s)	Inertia Matrix (J)			
Case I	$I_{\epsilon} = \operatorname{diag}(2, 3, 4)$	(0, 0, 0)	$\begin{bmatrix} 5 & 0 & 0 \\ 0 & 4 & 0 \\ 0 & 0 & 2.6 \end{bmatrix}$			
Case II	$I_{\epsilon} = \text{diag}(2,3,4)$	(0, 0, 0)	$\begin{bmatrix} 5 & 0 & 0 \\ 0 & 4 & 0 \\ 0 & 0 & 2.6 \end{bmatrix}$			
Case III	$I_{\epsilon_1} = \text{diag}(3, 4, 5)$ $I_{\epsilon_2} = \text{diag}(6, 1, 1)$	(1.5, 0, 0) + c (-4.5, 0, 0) + c	$\begin{bmatrix} 7.5932718 & 6e-7 & -4e-7 \\ -6e-7 & 9.9326434 & 0 \\ 4e-7 & 0 & 8.2731252 \end{bmatrix}$			
Case IV	$s = 2\sqrt{3}, \epsilon = 10^{-13}$	(0, 0, 0)	$\begin{bmatrix} 2 & 0 & 0 \\ 0 & 2 & 0 \\ 0 & 0 & 2 \end{bmatrix}$			

Table 1: Parameters for the hybrid systems where c = (-0.9937128, 0, 0).

Recall that the horizontal plane has the normal vector $\mathbf{n}^T = (0, 0, 1)$ and passes through the origin. In Case II, the tilted plane is a two-degree counterclockwise rotation of the horizontal plane about the *y*-axis, so the normal vector is

1057
$$\tilde{\boldsymbol{n}} = \begin{bmatrix} \cos\theta & 0 & \sin\theta \\ 0 & 1 & 0 \\ -\sin\theta & 0 & \cos\theta \end{bmatrix} \boldsymbol{n},$$

where $\theta = -\frac{\pi}{90}$. Furthermore, the rigid-bodies are described by the parameters given in Table 1059 1. In Case III, the body is a union of two ellipsoids whose centers are shifted by c so that 1060 the centroid coincide with the origin. Lastly, in Case IV, the rigid-body is a cube whose 1061 side-length is denoted by s, and its ϵ -rounding parameter is $\epsilon = 10^{-13}$. Table 1 also includes 1062 the standard inertia matrices J, and these matrices in Case I, II, and IV are computed using 1063 standard formulas assuming constant uniform density: Namely,

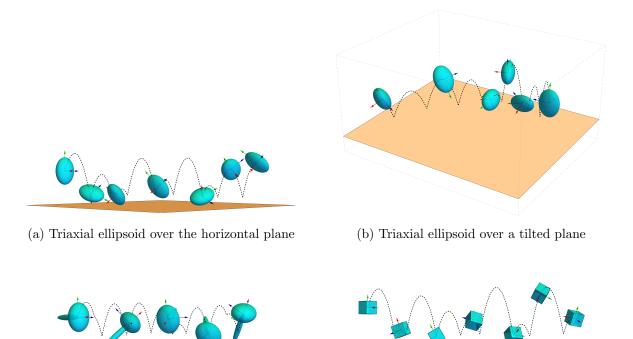
1064
$$J_{\text{Ellipsoid}} = \frac{1}{5}m \operatorname{diag}(b^2 + c^2, a^2 + c^2, a^2 + b^2),$$

1065
$$J_{\text{Cube}} = \frac{1}{6}ms^2 I_3,$$

where $I_{\epsilon} = \text{diag}(a, b, c)$ give the parameters for $J_{\text{Ellipsoid}}$, and m is the total mass of the rigidbody. In all of the cases, m = 1 including Case III where the standard matrix is computed and scaled reasonably using the built-in function MomentOfInertia in Mathematica 12. We fixed the following parameters for all cases including g = 9.80665 for the gravitational acceleration, h = 0.01 for the timestep, and $\epsilon_{\text{tol}} = 10^{-15}$ for tolerance. Recall that ϵ_{tol} is used as the tolerance in the bisection method to determine α_j , and in Newton's method to solve for the implicit equations, which are further discussed in Appendix D.

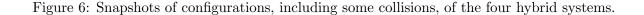
1073 For all four cases, the initial values are

1074
$$\boldsymbol{x}_0 = (0, 0, 10)^T$$
,



(c) Union of ellipsoids over the horizontal plane

(d) A cube over the horizontal plane



1075
$$\gamma_0 = (2, 2, 10)^T,$$

1076
$$\Pi_0 = (4, -4, 4)^T,$$

1077
$$R_0 = I_3.$$

1078 Lastly, note that the Zeno phenomenon does not arise in our four simulations because all 1079 the rigid-bodies are C^{∞} smooth, including the cube which was regularized using ϵ -rounding. 1080 However, the Zeno phenomenon could be observed if we considered a non-smooth rigid-body 1081 whose boundary is C^k for some finite $k \geq 2$.

1082 **7.1.** Snapshots. Fortunately, our numerical experiments for the hybrid systems can be visualized as a sequence of snapshots, as shown in Figure 6. We can make a number of 1083 collective observations for Case I, II, and IV, which share the horizontal plane in common. 1084In particular, the top-down view of the path of the center of mass for each rigid-body is 1085a straight line. This is immediate because the updates on the linear momenta away from 1086collisions are only affected by the gravitational direction, the z-component, in (5.9b). In 1087 addition, the instantaneous update on the linear momenta after the collision in (5.16c) and (5.19c) is dependent on $\frac{\partial \Phi}{\partial x}^T = \mathbf{n}^T = (0, 0, 1)$, affecting only the z-component again. Hence, 1088 1089only Case III with a tilted plane exhibits a curved path for the center of mass when viewed 1090 1091 top-down.

7.2. Transfer of Energy. Moreover, each hybrid system is closed, and their collisions are 1092 elastic. Therefore, the total energy within each system is conserved, which can be observed 1093 in Figure 7, where the data is taken over a short time interval. By conservation of angular 1094 momentum, the rotational kinetic energy (R.E.) is constant in the time intervals that are 10951096 away from the collisions, and so the translational kinetic + potential energy (T.P.E.) is also constant. However, we observe a transfer of energies between T.P.E and R.E. after each 1097 collision because the normal force, perpendicular to the plane, is no longer passing through 1098 the center of mass; as a result, this not only imparts an instantaneous impulse on the center of 1099mass, changing its linear momentum but also an instantaneous angular impulse on the body, 1100changing the angular momentum after the collision. By the conservation of energy in the 1101 system, the sum of T.P.E. and R.E. still gives the total energy, but these changes in momenta 1102 induce the transfer of energy between T.P.E. and R.E., which can be seen as jumps in Figure 11031104 7.

1105 **7.3. Short-Term & Long-Term Behaviors.** Both short-term and long-term energy be-1106 haviors are shown for our hybrid systems in Figure 8. For Case I and Case II, short-term energy behaviors are shown to illustrate how collisions affect the conservation properties. In 1107 particular, the red-dashed lines in both figures indicate the times of the impacts, and we see 1108 that the trend of the absolute energy errors shift after each impact; however, the magnitude 1109 of the error about each collision is essentially the same, so the jump conditions F_{Jump} still 1110 produce near energy conservation in the same manner as the discrete flow away from impacts. 1111 We also explore the long-term energy behaviors in Case III and Case IV with 10^5 integra-1112 tion steps; there are roughly 806 collisions that occur in Case III and roughly 652 collisions 1113 1114in Case IV. Overall, the LGCVI exhibits good long-time near energy conservation; however, there is a drift in both absolute energy errors, which is attributed to the numerous collisions 1115in each hybrid system. This is the case because the discrete Lagrangian variational mechanics 1116 away from the collisions uses a fixed timestep h, and preserves a modified Hamiltonian up to 1117an exponentially small error for exponentially long times by virtue of backward error analy-1118 sis. However, the modified Hamiltonian is an asymptotic expansion in the timestep h, and a 1119different timestep is taken during collisions in order to resolve the collision time accurately. 1120 This results in the preservation of a slightly different modified Hamiltonian, which results in 1121 1122a small energy drift after each collision.

Interestingly, we observe that this drift in energy appears to be roughly monotonic; otherwise, the absolute energy error plots would have negative trends for long periods of time as well. This could be the consequence of our use of the bisection method to resolve the collision time and our bias in choosing a configuration that is admissible, so we choose an approximation such that $\Phi(\mathbf{x}_i, R_i) \approx 0$ is always positive.

1128 **7.4.** Sensitivity to Initial Conditions. The hybrid systems involving rigid-bodies that are 1129 not spherical are, in general, sensitive to initial conditions by nature. As a result, we expect 1130 our collision algorithm to capture this when we apply slight changes in the initial position 1131 or attitude of the rigid-body, while maintaining the same linear and angular momentum. By 1132 following the path of the center of mass height (C.M. height) over the horizontal plane, we 1133 observe this sensitivity in Figure 9. The black dashed line is the path of C.M. height for 1134 the original initial conditions (x_0, R_0). The path of C.M. height of the slightly perturbed

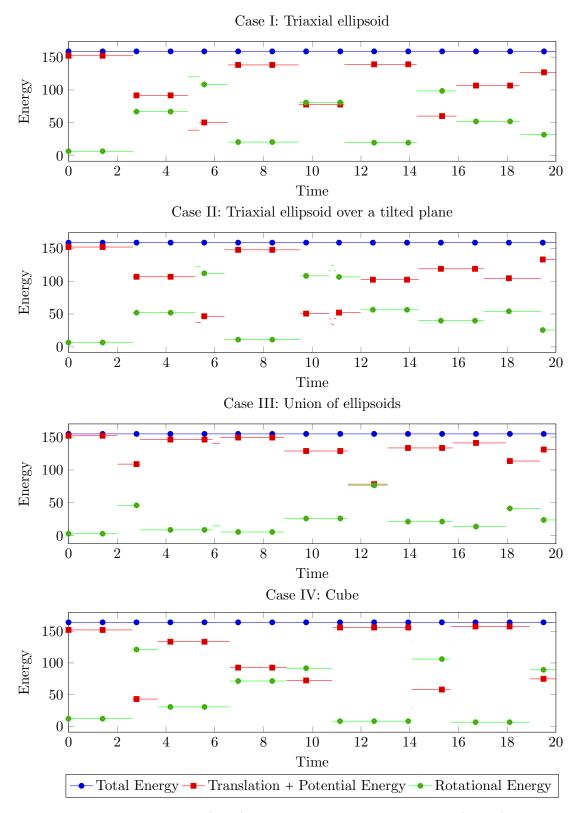


Figure 7: Plots of total energy (T.E.), translational + potential energy (T.P.E), and rotational energy (R.E) of different hybrid systems demonstrating the exchange of energies between T.P.E and R.E. after each collision.

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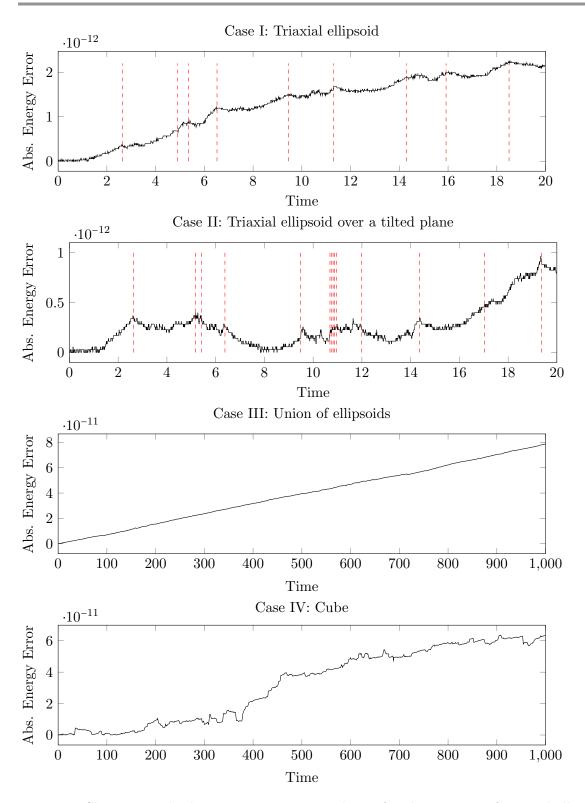


Figure 8: Short-term absolute energy errors are shown for the systems of triaxial ellipsoid and its counterpart with the tilted plane. Long-term absolute energy errors are shown for the systems of union of ellipsoids and cube, demonstrating energy drift after many collisions.

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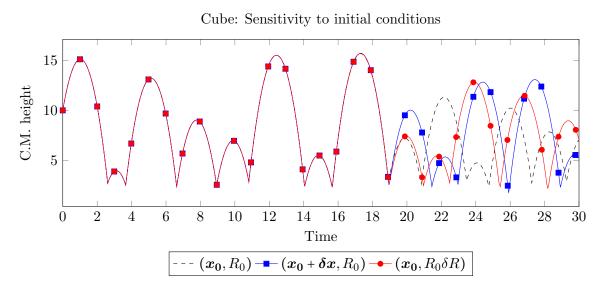


Figure 9: Plot of the height of the center of mass of the cube, for a given initial condition and two slight perturbations in position and attitude, to illustrate that the sensitivity to initial conditions is captured by our proposed collision algorithm.

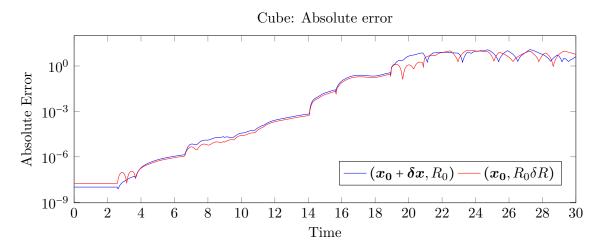


Figure 10: Absolute Error $\|\boldsymbol{x}_{\text{per}}(t) - \boldsymbol{x}(t)\|_2 + \|R_{\text{per}}(t) - R(t)\|_2$ plots for perturbed initial position and attitude in blue and red, respectively. The matrix norm is the induced 2-norm.

1135 position $(\boldsymbol{x_0} + \boldsymbol{\delta x}, R_0)$ and attitude $(\boldsymbol{x_0}, R_0 \boldsymbol{\delta R})$ are plotted in square-blue and circle-red lines, 1136 respectively; the perturbations are set as

1137
$$\boldsymbol{\delta x} = (0, 0, 10^{-8})^T$$

1138
$$\delta R = \begin{bmatrix} \cos\theta & 0 & \sin\theta \\ 0 & 1 & 0 \\ -\sin\theta & 0 & \cos\theta \end{bmatrix},$$

1139 where $\theta = 10^{-8}\pi$. By observing the C.M. heights, the discrete flows of the two perturbed 1140 systems appear to be the same as the unperturbed system for the first few collisions; however, 1141 the flows noticeably diverge after the eighth collision with the plane, and the plots in Figure 1142 9 clearly illustrate this.

1143 To ensure that the divergence of these trajectories is intrinsic to the nature of the system 1144 instead of being a consequence of the design of the algorithm, we look at the absolute errors:

1145
$$\operatorname{Err}_{\operatorname{per}}(t) = \|\boldsymbol{x}_{\operatorname{per}}(t) - \boldsymbol{x}(t)\|_{2} + \|R_{\operatorname{per}}(t) - R(t)\|_{2},$$

1146 where $(\boldsymbol{x}_{per}(t), R_{per}(t))$ represent the discrete flow for the perturbed initial conditions either in 1147 position $(\boldsymbol{x}_{pos}(t), R_{pos}(t))$ or in attitude $(\boldsymbol{x}_{att}(t), R_{att}(t))$. Of course $(\boldsymbol{x}(t), R(t))$ is the discrete 1148 flow for the system with unperturbed initial conditions (\boldsymbol{x}_0, R_0) . Note also that the matrix 1149 norm above is induced by the 2-norm. From the small perturbations, the initial absolute 1150 errors are

1151
$$\operatorname{Err}_{\operatorname{pos}}(0) = 10^{-8},$$

1152
$$\operatorname{Err}_{\operatorname{att}}(0) \approx 1.75 \cdot 10^{-8},$$

and these errors change in magnitude after each collision as seen in Figure 10, illustrating that the differences between the trajectories are amplified after each collision. Eventually, the absolute errors saturate because the perturbed systems become unrelated to the original system beside conserving approximately the same total energy.

8. Conclusions and Future Directions. We develop an algorithm to simulation a class of 1157 hybrid systems that is the extension of the classical bouncing ball hybrid system. Now, this 11581159class of hybrid systems in 3-dimensions comprise of a general convex rigid-body bouncing elas-1160 tically on a horizontal or tilted plane. The resulting algorithm is called a Lie group variational 1161 collision integrator (LGVCI) which is based on a combination of the work done in nonsmooth Lagrangian mechanics for collision variational integrators and Lie group variational integra-1162tors. Consequently, the LGVCI provide a combination of discrete flow maps for the hybrid 1163 1164systems away from the points of collision and jump conditions to update the instantaneous state after each collision. Our integrators heavily depend on the collision detection function 1165 Φ between the rigid-body and plane. 1166

Initially, we develop the algorithm for a model problem involving an ellipsoid and a hor-1167izontal plane. However, our theory easily extends to more general systems involving tilted 11681169 planes, unions and/or intersections of convex rigid-bodies by modifying and constructing the collision detection function. Furthermore, we introduce a convenient and straightforward reg-1170 ultrization using ϵ -rounding for the collision responses of convex rigid-bodies with corners. In 11711172general, these bodies are convex polyhedra. This development avoids the need for complicated nonlinear convex analysis of the corner impacts involving different inclusions and the computa-1173 1174tion of normal/tangent cones at the configuration of impact. Consequently, the regularization 1175 provides a unique deterministic response after the collision using the jump conditions we have 1176 derived, while still exhibiting the full range of potential outcomes that the formulation in-1177 volving differential inclusions and normal/tangent cones would exhibit by varying the initial 1178 conditions slightly.

1179 We performed extensive numerical experiments for four different hybrid systems: Case I: 1180 Triaxial ellipsoid over the horizontal plane; Case II: Triaxial ellipsoid over tilted plane; Case III: Union of ellipsoids over the horizontal plane; Case IV: Cube over the horizontal plane. 1181 1182Various observations and analyses of the algorithm are made from the experiments: First, the hybrid systems that impact a horizontal plane have a straight line center of mass motion 1183when viewed from the top. This is apparent since gravity on the rigid-body affects only the 1184 z-component, and there are no frictional forces involved. Second, by the conservation of total 1185energy of the system, there is a transfer of energy between rotational kinetic energy (R.E.) 1186 and translational kinetic + potential energy (T.P.E) during each collision; the cause is due 1187to the asymmetry of the rigid-body, whose instantaneous impulse at impact no longer passes 1188 through the center of mass causing changes in R.E. while the system conserve total energy. 1189 Third, the sensitive dependence on initial conditions for the hybrid systems is captured in 1190 the algorithm, and this sensitivity becomes apparent once a sufficient number of collisions has 1191 1192 occurred.

1193 Analysis of the algorithm demonstrates that the LGVCI is symplectic and momentumpreserving by design and has long-term, near energy conservation along both the smooth 1194portions and jumps of the system. However, there is a one-directional drift in the energy 1195errors, and we have attributed this to the approximate solution using the bisection method 1196 to the condition $\Phi(\mathbf{x}_i, R_i) = 0$ for the impact configuration; notably, the non-interpenetration 1197 constraint of the problem leads to an approximate point of impact where $\Phi(x_i, R_i)$ is always 1198 slightly positive. The use of a non-uniform timestep at the point of impact results in a slight 11991200 energy drift as variational integrators derive their long-time, near energy conservation from 1201 their conservation to exponentially small error of a modified Hamiltonian, but the modified Hamiltonian that is conserved is dependent on the timestep. 1202

1203 For future research, we intend to extend our variational collision integrators approach to 1204 hybrid systems with external and contact forces, as briefly discussed in [10] for the continuous setting. Moreover, we can consider elastic bodies (e.g., hyperelastic materials) which can be 1205formalized in the discrete variational method using the appropriate elastic potential energies. 12061207 From here, we may consider the analysis of collisions for convex-nonconvex rigid-bodies and 1208 then construct numerical integrators for these extensions. This is naturally a topic of interest as they more closely resemble real-world, complex systems that exhibit energy dissipation 1209 during collisions. 1210

There are also interesting applications to geometric control and optimal control on the 1211 1212 Special Euclidean group SE(3) and its submanifolds with boundary, which is the setting of this paper. We propose to study these geometric control problems for two reasons: First, the 1213 configuration space SE(3) is global, and its elements represent configurations of a real-world 1214 1215object uniquely, which is advantageous for describing dynamical systems analytically and to 1216 provide numerical methods based on variational principles. This is particularly critical when considering systems that exhibit large rotational motions that cannot be effectively described 1217 1218 using local coordinate based approaches. Second, novel algorithms that are both efficient and

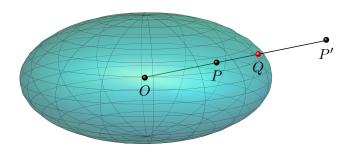


Figure 11: The pole of a point P with respect to an ellipsoid is P'.

geometrically exact can be developed for systems on SE(3) with unilateral constraints, which is demonstrated in [23] for a bilateral constraint.

Appendix A. Inversive Geometry of Ellipsoid & The Distance Formula. In this section, we recall some definitions and prove basic results in inversive geometry for an ellipsoid. We will ultimately arrive at the distance formula between a plane and an arbitrary ellipsoid, assuming that their intersection is empty. However, we start by formulating the distance between a plane and a standard ellipsoid, which will be defined in the first subsection.

1226 **A.1. Distance Between the Standard Ellipsoid and Plane..** Suppose a, b, c > 0 and 1227 $\mathbf{h} = (h_1, h_2, h_3) \in \mathbb{R}^3$, and for convenience, consider the function $f_{\mathcal{E}_h} : \mathbb{R}^3 \to \mathbb{R}$ defined by

1228 (A.1)
$$f_{\mathcal{E}_{h}}(\boldsymbol{z}) = \frac{(z_{1} - h_{1})^{2}}{a^{2}} + \frac{(z_{2} - h_{2})^{2}}{b^{2}} + \frac{(z_{3} - h_{3})^{2}}{c^{2}}.$$

1229 Then, let

1230

$$\mathcal{E}_{\boldsymbol{h}}(a,b,c) = \left\{ \boldsymbol{z} \in \mathbb{R}^3 \mid f_{\mathcal{E}_{\boldsymbol{h}}}(\boldsymbol{z}) \le 1 \right\}$$

1231 be the standard ellipsoid centered at h with its semiaxes of lengths a, b, and c lying parallel to 1232 the coordinate axes x, y, and z, respectively. We write $\mathcal{E}_{h} = \mathcal{E}_{h}(a, b, c)$ when the lengths are 1233 understood, and so $\partial \mathcal{E}_{h}$ is simply the boundary of the ellipsoid. We also denote the ellipsoid 1234 centered at the origin by \mathcal{E} , which is our main ellipsoid of reference.

1235 Definition A.1. Let O be the origin, and suppose $P \in \mathcal{E}$ and $Q \in \partial \mathcal{E}$ such that the points 1236 O, P, and Q are collinear on the ray \overrightarrow{OP} . Then the **pole** of P, denoted by P', with respect to 1237 (w.r.t.) the ellipsoid satisfies

1238 (A.2)
$$\left|\overline{OP}\right| \cdot \left|\overline{OP'}\right| = \left|\overline{OQ}\right|^2$$
,

1239 and also lies on the ray \overrightarrow{OP} , see Figure 11.

1240 Proposition A.2. Given $p \in \mathbb{R}^3$, the pole of p w.r.t. \mathcal{E} is given by

1241 (A.3)
$$p' = \frac{p}{f_{\mathcal{E}}(p)}.$$

1242 Furthermore, (p')' = p.

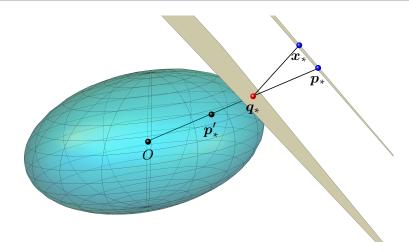


Figure 12: The pole of a plane with respect to an ellipsoid is p'_* .

1243 *Proof.* First, we determine $\boldsymbol{q} \in \partial \mathcal{E}$ lying on the ray from the origin to \boldsymbol{p} , and so $\boldsymbol{q} = t\boldsymbol{p}$ such 1244 that $f_{\mathcal{E}}(\boldsymbol{q}) = f_{\mathcal{E}}(t\boldsymbol{p}) = 1$. Note that $f_{\mathcal{E}}(t\boldsymbol{p}) = t^2 f_{\mathcal{E}}(\boldsymbol{p})$, so this gives $t = 1/\sqrt{f_{\mathcal{E}}(\boldsymbol{p})}$, implying 1245 that $\boldsymbol{q} = \boldsymbol{p}/\sqrt{f_{\mathcal{E}}(\boldsymbol{p})}$. Similarly, for $\boldsymbol{p'}$ lying on the same ray, $\boldsymbol{p'} = t'\boldsymbol{p}$. It also satisfies (A.2), 1246 so

1247
$$\|\boldsymbol{p}\|\|\boldsymbol{p'}\| = \|\boldsymbol{q}\|^2 \iff |t'|\|\boldsymbol{p}\|^2 = \frac{\|\boldsymbol{p}\|^2}{f_{\mathcal{E}}(\boldsymbol{p})} \iff t' = \frac{1}{f_{\mathcal{E}}(\boldsymbol{p})}.$$

1248 This gives (A.3). Lastly,

$$(\mathbf{p}')' = \frac{\mathbf{p}'}{f_{\mathcal{E}}(\mathbf{p}')} = \frac{\frac{\mathbf{p}}{f_{\mathcal{E}}(\mathbf{p})}}{f_{\mathcal{E}}\left(\frac{\mathbf{p}}{f_{\mathcal{E}}(\mathbf{p})}\right)} = \frac{\frac{\mathbf{p}}{f_{\mathcal{E}}(\mathbf{p})}}{\frac{1}{f_{\mathcal{E}}(\mathbf{p})^2}f_{\mathcal{E}}(\mathbf{p})} = \mathbf{p}.$$
1249

1250 Now, we introduce some notation for the planes in \mathbb{R}^3 . Let

1251
$$\boldsymbol{n} = (A, B, C) \in S^2 \text{ and } D \in \mathbb{R},$$

1252 where $S^2 = \{ \boldsymbol{z} \in \mathbb{R}^3 \mid ||\boldsymbol{z}|| = 1 \}$, and let

1253
$$\mathcal{P}(\boldsymbol{n}, D) = \{ \boldsymbol{z} \in \mathbb{R}^3 \mid Az_1 + Bz_2 + Cz_3 + D = 0 \} = \{ \boldsymbol{z} \in \mathbb{R}^3 \mid \boldsymbol{n}^T \boldsymbol{z} + D = 0 \}.$$

1254 Again, we write $\mathcal{P} = \mathcal{P}(\boldsymbol{n}, D)$ when the unit normal \boldsymbol{n} and the constant D are understood. 1255 Recall that for $X, Y \subset \mathbb{R}^3$, their distance is

1256
$$d_2(X,Y) = \inf\{\|\boldsymbol{x} - \boldsymbol{y}\| \mid \boldsymbol{x} \in X \text{ and } \boldsymbol{y} \in Y\}.$$

1257 Proposition A.3. Consider the planes $\mathcal{P} = \mathcal{P}(n, D)$ and $\mathcal{P}' = \mathcal{P}(n, D')$, then

1258 (A.4)
$$d_2(\mathcal{P}, \mathcal{P}') = |D - D'|.$$

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Lemma A.4. Suppose $\mathcal{P} = \mathcal{P}(\boldsymbol{n}, D)$ and $\mathcal{P}' = \mathcal{P}(\boldsymbol{n}, D')$, and let $X \subset \mathcal{P}'$, then

1260
$$d_2(\mathcal{P}, X) = d_2(\mathcal{P}, \mathcal{P}')$$

1261 Lemma A.5. Given an ellipsoid \mathcal{E} and plane $\mathcal{P} = \mathcal{P}(n, D)$ where $\mathcal{E} \cap \mathcal{P} = \emptyset$,

1262 (i) there exists $x_* \in \mathcal{P}$ and $q_* \in \mathcal{E}$ such that

1263
$$\| \boldsymbol{q}_* - \boldsymbol{x}_* \| = d_2(\mathcal{E}, \mathcal{P})$$

1264 Moreover, $q_* \in \partial \mathcal{E}$, and it is given by one of these expressions,

1265 (A.5)
$$q_*^{\pm} = \pm \frac{(Aa^2, Bb^2, Cc^2)}{\sqrt{A^2a^2 + B^2b^2 + C^2c^2}}.$$

1266 (ii) The point $p_* \in \mathcal{P}$, lying on the ray from the origin to q_* , and its pole p'_* are

1267 (A.6a)
$$p_* = -\frac{D}{A^2a^2 + B^2b^2 + C^2c^2}(Aa^2, Bb^2, Cc^2),$$

1268 (A.6b)
$$p'_{*} = -\frac{1}{D}(Aa^{2}, Bb^{2}, Cc^{2}).$$

1269 We call p'_{*} the pole of the plane \mathcal{P} w.r.t. \mathcal{E} (see Figure 12).

1270 *Proof.* We prove Lemma A.5 here. Note that \mathcal{E} is compact and \mathcal{P} is a closed set, so a 1271 basic result for metric spaces give us the existence of $x_* \in \mathcal{P}$ and $q_* \in \mathcal{E}$. Now, consider the 1272 set

1273
$$J = \{ D' \in \mathbb{R} \mid \mathcal{P}(n, D') \cap \mathcal{E} \neq \emptyset \}.$$

1274 One can show that this set is compact and connected, and hence a closed interval in \mathbb{R} . Let 1275 $X_{D'} = \mathcal{P}(\boldsymbol{n}, D') \cap \mathcal{E}$, where $D' \in J$. We get

1276
$$d_2(\mathcal{E}, \mathcal{P}) = \inf_{D' \in J} d_2(X_{D'}, \mathcal{P}) = \inf_{D' \in J} d_2(\mathcal{P}(\boldsymbol{n}, D'), \mathcal{P}) = \inf_{D' \in J} |D - D'|,$$

1277 where we used Lemma A.4 and Proposition A.3. Since J is compact, there exists a $D \in J$ 1278 that attains the infimum. Furthermore, the function $D' \mapsto |D - D'|$ is strictly monotonic on 1279 J, so the constant of interest is either the maximum or minimum of J.

1280 Intuitively, the planes of interest are the tangent planes at some points $q_* \in \partial \mathcal{E}$ with 1281 normal vectors with the same direction as $\pm n$. These points are computable: First, we 1282 compute $\nabla f_{\mathcal{E}}(z) = 2\left(\frac{x_1}{a^2}, \frac{x_2}{b^2}, \frac{x_3}{c^2}\right)$, which is a normal vector on $\partial \mathcal{E}$ for $z \in \partial \mathcal{E}$. Now, we find z1283 such that $\nabla f_{\mathcal{E}}(z) = \pm n$. This gives

1284
$$\boldsymbol{z} = \pm \frac{1}{2} (Aa^2, Bb^2, Cc^2),$$

1285 but we are looking for $q_* = tz \in \partial \mathcal{E}$, implying that $f_{\mathcal{E}}(q_*) = f_{\mathcal{E}}(tz) = 1$. Hence,

1286
$$t = \frac{2}{\sqrt{A^2 a^2 + B^2 b^2 + C^2 c^2}},$$

1287 which gives (A.5).

1288 Since $p_* \in \mathcal{P}$ lies on the ray from the origin to q_* , $p_* = t'z$ such that $n^T p_* + D = 0$. 1289 Finding t' gives us (A.6a). We use Proposition A.2 to compute p'_* . Note that $p'_* \in \mathcal{E}$ since 1290 p_* is in the complement of the ellipsoid, and so $f_{\mathcal{E}}(p'_*) < 1$. 1291 We state our main results next:

1292 Theorem A.6. Given an ellipsoid \mathcal{E} and plane \mathcal{P} , then $\mathcal{E} \cap \mathcal{P} = \emptyset$ if and only if $f_{\mathcal{E}}(p'_*) < 1$, 1293 where p'_* is the pole of the plane \mathcal{P} w.r.t. \mathcal{E} .

1294 Theorem A.7. Given an ellipsoid \mathcal{E} and plane \mathcal{P} where $\mathcal{E} \cap \mathcal{P} = \emptyset$, then

1295 (A.7)
$$d_2(\mathcal{E}, \mathcal{P}) = \min\{|D \pm \sqrt{A^2 a^2 + B^2 b^2 + C^2 c^2}|\} = \min\{|D \pm ||I_{\mathcal{E}} \boldsymbol{n}|||\},\$$

1296 where $I_{\mathcal{E}} = diag(a, b, c)$.

1297 *Proof.* The planes of interest are the tangent planes of \mathcal{E} with the unit normal \boldsymbol{n} , and there 1298 are two possible tangent planes located at $\boldsymbol{q}_{*}^{\pm} = \pm (Aa^2, Bb^2, Cc^2)/\sqrt{A^2a^2 + B^2b^2 + C^2c^2}$. 1299 Hence, the equations of the planes are

1300
$$\boldsymbol{n}^{T}(\boldsymbol{z}-\boldsymbol{q}_{*}^{\pm})=0 \iff Az_{1}+Bz_{2}+Cz_{3}\mp\sqrt{A^{2}a^{2}+B^{2}b^{2}+C^{2}c^{2}}=0.$$

1301 For convenience, denote $\mathcal{P}^{\pm} = \mathcal{P}(\boldsymbol{n}, D^{\pm})$, where $D^{\pm} = \mp \sqrt{A^2 a^2 + B^2 b^2 + C^2 c^2}$. Then, using 1302 Lemma A.5, we obtain

1303
$$d_2(\mathcal{E}, \mathcal{P}) = \min\{d_2(\mathcal{P}^{\pm}, \mathcal{P})\} = \min\{|D - D^{\pm}|\}.$$

1304 One can quickly check that $D^{\pm} = \mp \| I_{\mathcal{E}} \boldsymbol{n} \|$.

1305 **A.2. Distance Between an Arbitrary Ellipsoid and Plane..** We shall discuss the distance 1306 between an arbitrary ellipsoid and plane, assuming non-intersection. However, an explanation 1307 of the relationship between the standard and arbitrary ellipsoid using the configuration space 1308 SE(3) will be given first.

1309 Consider the ellipsoid \mathcal{E} and $(\boldsymbol{x}, R) \in SE(3)$. For convenience, we introduce the following 1310 maps,

1311
$$T_{\boldsymbol{x}}: \mathbb{R}^3 \longrightarrow \mathbb{R}^3, \quad \boldsymbol{z} \longmapsto \boldsymbol{z} + \boldsymbol{x},$$
1312 $T_R: \mathbb{R}^3 \longrightarrow \mathbb{R}^3, \quad \boldsymbol{z} \longmapsto R \boldsymbol{z}.$

1313 Then, let $T_{(\boldsymbol{x},R)}(\boldsymbol{z}) = (T_{\boldsymbol{x}} \circ T_R)(\boldsymbol{z}) = R\boldsymbol{z} + \boldsymbol{x}$, which is the action of (\boldsymbol{x},R) on $\mathbb{R}^3 \times \{1\}$ 1314 given by homogeneous transformations (2.8b). As a result, the *arbitrary ellipsoid* is the image 1315 $T_{(\boldsymbol{x},R)}(\mathcal{E})$ given by the configuration element (\boldsymbol{x},R) . Now, we prove our final result.

1316 Lemma A.8. Suppose $(\boldsymbol{x}, R) \in SE(3)$, then $T_{(\boldsymbol{x},R)}$ is an isometry.

1317 Lemma A.9. Suppose
$$\boldsymbol{x} \in \mathbb{R}^3$$
 and $\mathcal{P} = \mathcal{P}(\boldsymbol{n}, D)$, then $T_{\boldsymbol{x}}(\mathcal{P}) = \mathcal{P}(\boldsymbol{n}, D - \boldsymbol{n}^T \boldsymbol{x})$.

1318 *Proof.* The equation of the plane $T_x(\mathcal{P}) = \{z' = z + x \mid z \in \mathcal{P}\}$ is

1319
$$\boldsymbol{n}^T(\boldsymbol{z}'-\boldsymbol{x}) + D = 0 \iff \boldsymbol{n}^T\boldsymbol{z}' + (D - \boldsymbol{n}^T\boldsymbol{x}) = 0$$

1320 Hence, the new constant is $D - \boldsymbol{n}^T \boldsymbol{x}$.

1321 Lemma A.10. Suppose $R \in SO(3)$ and $\mathcal{P} = \mathcal{P}(\boldsymbol{n}, D)$, then $T_R(\mathcal{P}) = \mathcal{P}(R\boldsymbol{n}, D)$.

Proof. Note that $T_R(\mathcal{P}) = \{ \boldsymbol{z}' = R\boldsymbol{z} \mid \boldsymbol{z} \in \mathcal{P} \}$, and elements of the set satisfy 1322

$$(R\boldsymbol{n})^T\boldsymbol{z}' + D = 0 \iff \boldsymbol{n}^T R^T \boldsymbol{z}' + D = 0.$$

Hence, the new unit normal is Rn. 1324

Theorem A.11. Consider the ellipsoid \mathcal{E} and $(\boldsymbol{x},R) \in SE(3)$. Let the ellipsoid \mathcal{E}' = 1325 $T_{(\boldsymbol{x},R)}(\mathcal{E})$ and $\mathcal{P} = \mathcal{P}(\boldsymbol{n},D)$ such that $\mathcal{E}' \cap \mathcal{P} = \emptyset$. Then, 1326

1327 (A.8)
$$d_2(\mathcal{E}', \mathcal{P}) = \min\{|(D + \boldsymbol{n}^T \boldsymbol{x}) \pm ||I_{\mathcal{E}} R^T \boldsymbol{n}||\}.$$

Furthermore, $\mathcal{E}' \cap \mathcal{P} = \emptyset$ if and only if 1328

1329 (A.9)
$$||I_{\mathcal{E}}R^T \boldsymbol{n}|| < |D + \boldsymbol{n}^T \boldsymbol{x}|.$$

Proof. We will use Theorem A.7 in this proof, but it only applies for the standard ellipsoid. 1330 We must then consider $(T_{(\boldsymbol{x},R)})^{-1}(\mathcal{E}') = \mathcal{E}$, where the inverse is also an isometry because $T_{(\boldsymbol{x},R)}$ is an isometry by Lemma A.8. Let us denote $\mathcal{P}' = (T_{(\boldsymbol{x},R)})^{-1}(\mathcal{P})$, and so we arrive at the 1331 1332equivalent problem: 1333 133

$$d_2(\mathcal{E}',\mathcal{P}) = d_2(\mathcal{E},\mathcal{P}')$$

Note that $(T_{(x,R)})^{-1} = T_{R^T} \circ T_{-x}$, so 1335

1336
$$\mathcal{P}' = (T_{R^T} \circ T_{-\boldsymbol{x}})(\mathcal{P}(\boldsymbol{n}, D)) = T_{R^T}(\mathcal{P}(\boldsymbol{n}, D + \boldsymbol{n}^T \boldsymbol{x})) = \mathcal{P}(R^T \boldsymbol{n}, D + \boldsymbol{n}^T \boldsymbol{x}),$$

where we used Lemmas A.9 and A.10. Applying Theorem A.7 gives (A.8). 1337

Now, from equation (A.6b), the pole of the plane \mathcal{P}' w.r.t. to \mathcal{E} is $p'_* = -I_{\mathcal{E}}^2 R^T n / (D + I_{\mathcal{E}})^2 R^T n / (D + I_{\mathcal{E}})^$ 1338 $\boldsymbol{n}^T \boldsymbol{x}$). Then, Theorem A.6 tells us that $f_{\mathcal{E}}(\boldsymbol{p'_*}) < 1$, which gives 1339

1340
$$\frac{\boldsymbol{n}^T R I_{\mathcal{E}}^2 R^T \boldsymbol{n}}{(D + \boldsymbol{n}^T \boldsymbol{x})^2} < 1 \implies \|I_{\mathcal{E}} R^T \boldsymbol{n}\| < |D + \boldsymbol{n}^T \boldsymbol{x}|,$$

where $f_{\mathcal{E}}(\boldsymbol{z}) = \boldsymbol{z}^T (I_{\mathcal{E}}^{-1})^2 \boldsymbol{z}$. 1341

Appendix B. Standard and Nonstandard Inertia Matrix of a Rigid-Body. Let \mathcal{B} denote 1342 the set of body elements of a rigid-body, and let $(x, R) \in SE(3)$ describe the configuration of 1343 the rigid-body. The inertial position of a body element of \mathcal{B} is $x + R\rho$, where $\rho \in \mathbb{R}^3$ is the 1344position of the body element relative to the origin of the body-fixed frame. We define 1345

1346 (B.1)
$$J_d = \int_{\mathcal{B}} \boldsymbol{\rho} \boldsymbol{\rho}^T \, dm = \int_{\mathcal{B}} \begin{bmatrix} x^2 & xy & xz \\ xy & y^2 & yz \\ xz & yz & z^2 \end{bmatrix} \, dm$$

as the nonstandard inertia matrix, where $\rho^T = (x y z)$. On the other hand, the standard 1347 inertia matrix is defined by 1348

1349 (B.2)
$$J = \int_{\mathcal{B}} S(\boldsymbol{\rho})^T S(\boldsymbol{\rho}) \, dm = \int_{\mathcal{B}} \begin{bmatrix} y^2 + z^2 & -xy & -xz \\ -xy & x^2 + z^2 & -yz \\ -xz & -yz & x^2 + y^2 \end{bmatrix} \, dm.$$

1350 Using the property (2.1b), we have that $S(\boldsymbol{\rho})^T S(\boldsymbol{\rho}) = (\boldsymbol{\rho}^T \boldsymbol{\rho}) I_3 - \boldsymbol{\rho} \boldsymbol{\rho}^T$, giving us (4.1),

$$J = \operatorname{tr}[J_d]I_3 - J_d$$

$$J_{d} = \frac{1}{2} \operatorname{tr}[J]I_{3} - J$$

1353 Proposition B.1. If $\Omega \in \mathbb{R}^3$, then $S(J\Omega) = S(\Omega)J_d + J_dS(\Omega)$.

1354 *Proof.* Let $(\Omega_1, \Omega_2, \Omega_3) = \mathbf{\Omega} \in \mathbb{R}^3$. Let $J_{ij} = \int_{\mathcal{B}} ij \, dm$, where $i, j \in \{x, y, z\}$, so given 1355 i = x and j = y, $J_{xy} = \int_{\mathcal{B}} xy \, dm$. One can show that the left-hand side and the right-hand 1356 side are equal to $S(\boldsymbol{\omega})$, where

$$\boldsymbol{\omega} = \begin{pmatrix} (J_{yy} + J_{zz})\Omega_1 - J_{xy}\Omega_2 - J_{xz}\Omega_3\\ -J_{xy}\Omega_1 + (J_{xx} + J_{zz})\Omega_2 - J_{yz}\Omega_3\\ -J_{xz}\Omega_1 - J_{yz}\Omega_2 + (J_{xx} + J_{yy})\Omega_3 \end{pmatrix}.$$

1358 Proposition B.2. Suppose $\Omega \in \mathbb{R}^3$, then $\frac{1}{2} \operatorname{tr}[S(\Omega) J_d S(\Omega)^T] = \frac{1}{2} \Omega^T J \Omega$.

1359 *Proof.* Note that

1357

1360
$$\frac{1}{2}\operatorname{tr}[S(\mathbf{\Omega})J_dS(\mathbf{\Omega})^T] = \frac{1}{2}\int_{\mathcal{B}} S(\mathbf{\Omega})\boldsymbol{\rho}\boldsymbol{\rho}^T S(\mathbf{\Omega})\,dm = \frac{1}{2}\int_{\mathcal{B}} \|S(\mathbf{\Omega})\boldsymbol{\rho}\|^2\,dm.$$

1361 Then, recall that $S(\Omega)\rho = \Omega \times \rho = -\rho \times \Omega = -S(\rho)\Omega$, from which we obtain

1362
$$\frac{1}{2} \int_{\mathcal{B}} \|S(\mathbf{\Omega})\boldsymbol{\rho}\|^2 \, dm = \frac{1}{2} \int_{\mathcal{B}} \|S(\boldsymbol{\rho})\mathbf{\Omega}\|^2 \, dm = \frac{1}{2} \int_{\mathcal{B}} \mathbf{\Omega}^T S(\boldsymbol{\rho})^T S(\boldsymbol{\rho})\mathbf{\Omega} \, dm = \frac{1}{2} \mathbf{\Omega}^T J \mathbf{\Omega}.$$

1363 **Appendix C. Matrix Derivatives.** In this section, we derive the derivatives with respect 1364 to a matrix for the following cases:

1365 Proposition C.1. Let $\boldsymbol{a}, \boldsymbol{b}, \boldsymbol{v} \in \mathbb{R}^n$ and $A \in \mathbb{R}^{n \times n}$, then 1366 $M.a \; \frac{\partial}{\partial X} (\boldsymbol{a}^T X \boldsymbol{b}) = \boldsymbol{a} \otimes \boldsymbol{b} = \boldsymbol{a} \boldsymbol{b}^T$,

1367
$$M.b \; \frac{\partial}{\partial X} (\boldsymbol{a}^T X^T \boldsymbol{b}) = \boldsymbol{b} \otimes \boldsymbol{a} = \boldsymbol{b} \boldsymbol{a}^T,$$

1368
$$M.c \; \frac{\partial}{\partial X}(\operatorname{tr}[AXB]) = A^T B^T,$$

1369
$$M.d \frac{\partial}{\partial X}(\operatorname{tr}[AX^TB]) = BA,$$

1370 $M.e \ \boldsymbol{v}^T \frac{\partial}{\partial X}(AX\boldsymbol{b}) = \boldsymbol{v}^T(A \otimes \boldsymbol{b}) = A^T \boldsymbol{v} \otimes \boldsymbol{b} = A^T \boldsymbol{v} \boldsymbol{b}^T,$

1371
$$M.f \ \boldsymbol{v}^T \frac{\partial}{\partial X} (AX^T \boldsymbol{b}) = \boldsymbol{b} \otimes A^T \boldsymbol{v} = \boldsymbol{b} \boldsymbol{v}^T A.$$

1372 *Proof.* It suffices to show (M.a), (M.c), and (M.e) since differentiation and transposition 1373 commute.

1374 (M.a) We write the partial derivative with respect to X_{kl} for each k, l = 1, ..., n. Then,

1375
$$\frac{\partial}{\partial X_{kl}} \left(\sum_{i,j} a_i X_{ij} b_j \right) = a_i \delta_{ik} \delta_{jl} b_j = a_k b_l$$

1376 Hence, $(ab^T)_{kl} = a_k b_l$.

1377 (M.c) Similarly, we have

1378
$$\frac{\partial}{\partial X_{kl}} \left(\sum_{i} (AXB)_{ii} \right) = \frac{\partial}{\partial X_{kl}} \left(\sum_{i,j,h} A_{ij} X_{jh} B_{hi} \right) = \sum_{i} A_{ij} \delta_{jk} \delta_{hl} B_{hi} = \sum_{i} A_{ik} B_{li}.$$

1379 Recall that $(A^T B^T)_{kl} = \sum_i (A^T)_{ki} (B^T)_{il} = \sum_i A_{ik} B_{li}$. 1380 (M.e) This case is treated with care since $\frac{\partial}{\partial X} (AXb) \in \mathbb{R}^{n \times n \times n}$, a third-order tensor.

1381
$$\sum_{j} \left[\frac{\partial}{\partial X} (AXb) \right]_{jkl} v_j = \sum_{j} \frac{\partial (AXb)_j}{\partial X_{kl}} v_j$$

1382
$$= \sum_{j,m,p} \frac{\partial}{\partial X_{kl}} (A_{jm} X_{mp} b_p v_j)$$

$$=\sum_{j}A_{jm}\delta_{mk}\delta_{pl}b_{p}v_{j}$$

$$=\sum_{j}A_{jk}v_{j}b_{l}$$

1385 Note that $(A^T \boldsymbol{v} \boldsymbol{b}^T)_{kl} = \sum_j (A^T)_{kj} (\boldsymbol{v} \boldsymbol{b}^T)_{jl} = \sum_j A_{jk} v_j b_l.$

1386 Appendix D. Solving the Implicit Function. Suppose that $g \in \mathbb{R}^3$, and we want to solve 1387 for $F \in SO(3)$ so that

1388 (D.1)
$$S(\boldsymbol{g}) = \operatorname{Asym}(FJ_d) = FJ_d - J_d F^T.$$

1389 Note that F is linear in the equation above, but it is implicit since there is a constraint that 1390 $F \in SO(3)$. We introduce two solution approaches based on two different retractions and 1391 Newton's method.

1392 **D.1. Exponential Map.** When $F \in SO(3)$, it can be expressed by the exponential map 1393 on the Lie algebra $\mathfrak{so}(3)$, i.e., $F = \exp(\Omega)$ for some $\Omega \in \mathfrak{so}(3)$. Since $\mathfrak{so}(3) \simeq \mathbb{R}^3$ via the skew 1394 map, we also have $F = \exp(S(\mathbf{f}))$ for some $\mathbf{f} \in \mathbb{R}^3$, and it can be expressed explicitly using 1395 the *Rodrigues' rotation formula*:

1396 (D.2)
$$F = \exp(S(\boldsymbol{f})) = I_3 + \frac{\sin \|\boldsymbol{f}\|}{\|\boldsymbol{f}\|} S(\boldsymbol{f}) + \frac{1 - \cos \|\boldsymbol{f}\|}{\|\boldsymbol{f}\|^2} S(\boldsymbol{f})^2.$$

1397 Substituting (D.2) into (D.1), we have that

1398
$$S(\boldsymbol{g}) = \frac{\sin \|\boldsymbol{f}\|}{\|\boldsymbol{f}\|} S(J\boldsymbol{f}) + \frac{1 - \cos \|\boldsymbol{f}\|}{\|\boldsymbol{f}\|^2} S(\boldsymbol{f} \times J\boldsymbol{f})$$

1399 Since the skew map is linear, we can view the above equation in terms of vectors, $\boldsymbol{g} = G(\boldsymbol{f})$, 1400 where $G : \mathbb{R}^3 \to \mathbb{R}^3$ is defined by

1401 (D.3)
$$G(f) = \frac{\sin \|f\|}{\|f\|} Jf + \frac{1 - \cos \|f\|}{\|f\|^2} f \times Jf.$$

1402 To solve for f in the equation $\mathbf{0} = G(f) - g$, we apply Newton's method,

1403 (D.4)
$$f_{k+1} = f_k - \nabla G(f)^{-1} (G(f) - g),$$

1404 where the Jacobian of G is given by

1405
$$\nabla G(f) = \frac{\|f\| \cos \|f\| - \sin \|f\|}{\|f\|^3} Jf \otimes f + \frac{\sin \|f\|}{\|f\|} J$$
$$\|f\| \sin \|f\| - 2(1 - \cos \|f\|) + \frac{1}{2} \int df df df$$

$$+ \frac{\|\boldsymbol{f}\| \otimes \boldsymbol{M} \| \|\boldsymbol{f}\|}{\|\boldsymbol{f}\|^4} (\boldsymbol{f} \times J\boldsymbol{f}) \otimes \boldsymbol{f} \\ + \frac{1 - \cos \|\boldsymbol{f}\|}{\|\boldsymbol{f}\|^2} (S(\boldsymbol{f})J - S(J\boldsymbol{f})).$$

1407

1408 **D.2. Cayley Transformation.** Similarly, the *Cayley transformation* is a local diffeomor-1409 phism between the Lie algebra and the Lie group, so $\operatorname{Cay} S(f_c) = F \in SO(3)$ for some 1410 $f_c \in \mathbb{R}^3$. The Cayley transformation is defined by

1411 (D.5)
$$F = \operatorname{Cay} S(f_c) = (I_3 + S(f_c))(I - S(f_c))^{-1}.$$

1412 Motivated by the Neumann series, we can express the map above more concretely without 1413 the inverse. In particular, we write

1414 (D.6)
$$(I - S(\boldsymbol{f_c}))^{-1} = \frac{1}{1 + \|\boldsymbol{f_c}\|^2} \left((1 + \|\boldsymbol{f_c}\|^2) I_3 + S(\boldsymbol{f_c}) + S(\boldsymbol{f_c})^2 \right).$$

1415 As a result, the Cayley transformation is written explicitly as

1416 (D.7)
$$F = \operatorname{Cay} S(\boldsymbol{f_c}) = \frac{1}{1 + \|\boldsymbol{f_c}\|^2} \left((1 + \|\boldsymbol{f_c}\|^2) I_3 + 2S(\boldsymbol{f_c}) + 2S(\boldsymbol{f_c})^2 \right).$$

1417 Now substituting (D.7) into (D.1), we arrive at

1418
$$S(\boldsymbol{g}) = \frac{2}{1 + \|\boldsymbol{f_c}\|^2} \left(S(J\boldsymbol{f_c}) + S(\boldsymbol{f_c} \times J\boldsymbol{f_c}) \right)$$

1419 By the same argument, we can write the equation in vector form as $\boldsymbol{g} = G_c(\boldsymbol{f_c})$ where 1420 $G_c : \mathbb{R}^3 \to \mathbb{R}^3$ is defined by

1421 (D.8)
$$G_c(f_c) = \frac{2}{1 + \|f_c\|^2} \left(Jf_c + f_c \times Jf_c \right).$$

1422 The solution to $\mathbf{0} = G_c(\mathbf{f_c}) - \mathbf{g}$ is given by Newton's method (D.4), where the Jacobian is 1423 given by

1424
$$\nabla G_c(\boldsymbol{f_c}) = \frac{2}{1 + \|\boldsymbol{f_c}\|^2} \left((J + S(\boldsymbol{f_c})J - S(J\boldsymbol{f_c})) - \frac{2}{1 + \|\boldsymbol{f_c}\|^2} (J\boldsymbol{f_c} + \boldsymbol{f_c} \times J\boldsymbol{f_c}) \otimes \boldsymbol{f_c} \right).$$

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1425	R	e	fe	re	n	ces

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